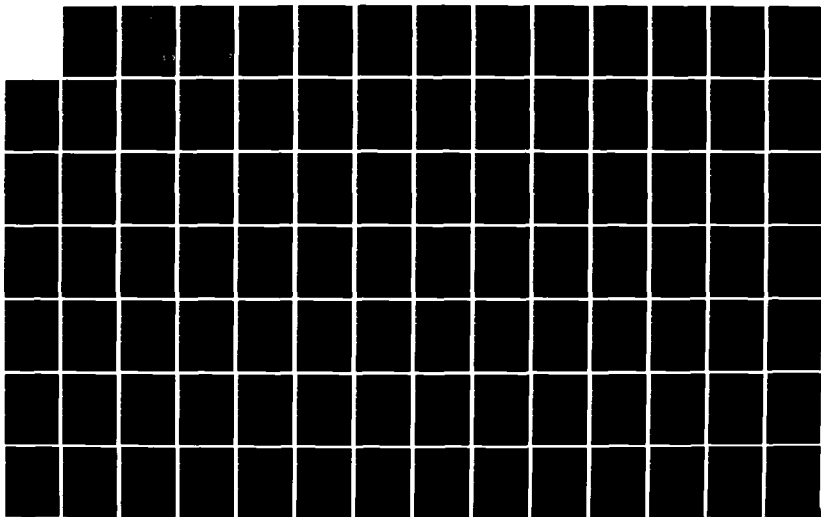
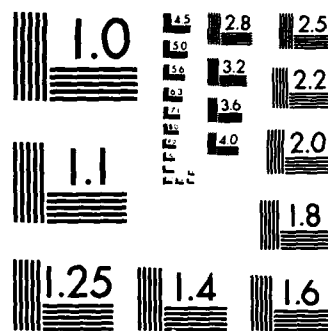


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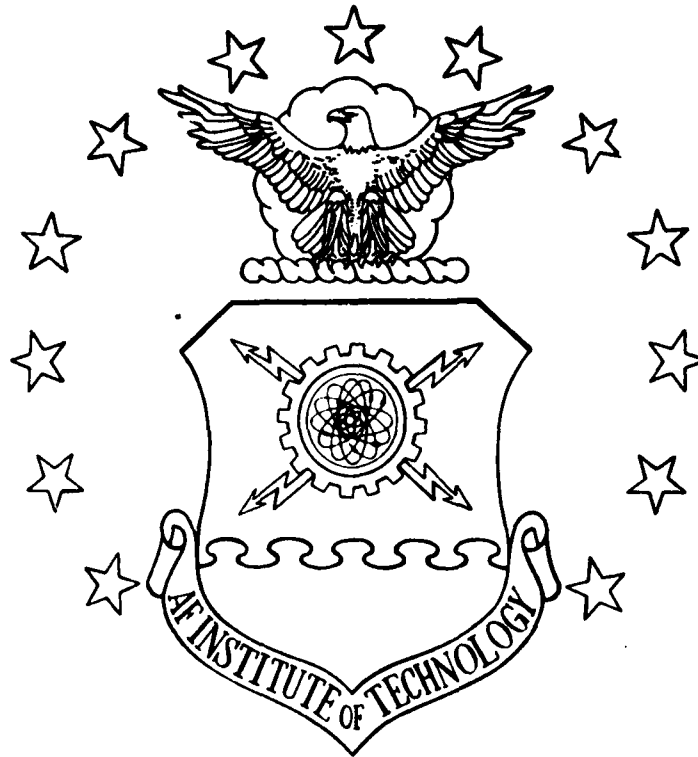
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HEATING PARAMETER ESTIMATION USING
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WIND TUNNEL TEST ARTICLES

THESIS

Neil T. Cahoon
Captain, USAF

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**HEATING PARAMETER ESTIMATION USING COAXIAL
THERMOCOUPLE GAUGES IN WIND TUNNEL
TEST ARTICLES**

THESIS

**Presented to the Faculty of the School of Engineering
of the Air Force Institute of Technology
Air University
In Partial Fulfillment of the
Requirements for the Degree of
Master of Science in Aeronautical Engineering**

**Neil T. Cahoon, B.S.E.
Captain, USAF**

December 1984

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List of Symbols

A, A', b, d	Coefficient Matrices ($i \times i$)
c	Specific Heat
E	Residual Error Vector (m)
G	Kalman Gain Vector (i)
H	Thermocouple Location Matrix ($m \times i$)
h_0	Magnitude of Heat Transfer Coefficient Ratio
h_s	Heat Transfer Coefficient Derivative
h_{bar}	Heat Transfer Coefficient Ratio
h_{ref}	Reference Heat Transfer Coefficient at Zero State
I	Identity Matrix
J_k	Conditional Information Matrix
k	Thermal Conductivity
L	Total Number of Spatial Node Points
P	Covariance Matrix ($i \times i$)
Q	Model Error Covariance Matrix ($i \times i$)
q	Heating Rate
R_m	Covariance for m th measurement
S	Score Vector (k)
$S_{i,k}$	Sensitivity Vectors (i) for the k th Parameter
T_{aw}	Adiabatic Wall Temperature
t	Time
U	Temperature Vector (i)

x	Spatial Coordinate
Y	Thermocouple Measurement Vector
α	Angle of Attack
ϵ	Emissivity
θ	Parameter Vector
μ_n	Measurement Vector at nth Time Point
ρ	Density
σ	Stefan-Boltzmann Constant
ϕ	Transition Matrix
ϕ_c	Scaling Parameter for Specific Heat
ϕ_k	Scaling Parameter for Thermal Conductivity

Superscripts

-	a priori Propagation
+	a posteriori Propagation
*	Parameter Estimate
n	Time Level
s	Iteration Level
T	Transpose

Subscripts

i	Spatial Node Point
k	Number of Model Parameter
m	Number of Thermocouples
o	Freestream Conditions

Abstract

A heat energy balance is applied to a coaxial thermocouple gage for parameter estimation in wind tunnel test articles. This method can significantly reduce wind tunnel test costs and time. Modifications to the data reduction technique HEATEST (HEATing ESTimation) are made. The program allows for transient test techniques to be used as well as assuming an isothermal wall. A non-linear convective heat transfer coefficient model may also be used. Data is generated to test the new program. Temperature profiles throughout the thermocouple gage were good and were compared with changes in time step, thermocouple length, and number of discrete node points. The estimation of the convective heat transfer coefficient and thermal conductivity were excellent.

**HEATING PARAMETER ESTIMATION USING COAXIAL
THERMOCOUPLE GAUGES IN WIND TUNNEL
TEST ARTICLES**

I. INTRODUCTION

1.1 Background

The determination of heat transfer rates on hypersonic configurations upon reentry is important for the survival of the vehicle. The problem is that the heat rate is not a quantity which may be directly scaled from model tests in wind tunnels. However, the parameters which make up the heat rate equation (thermal conductivity and specific heat, for example) can be scaled from which the heat rate may then be calculated. Wind tunnel heat transfer measurements have traditionally used a thin walled test model fabricated with thermocouples mounted on the inside skin surface. The "thermal model" then yields a heat rate based on temperature measurement from the thermocouple. Another technique uses a coaxial thermocouple gage mounted in a thick skin model. A discussion of the two methods follows.

The Traditional Thin Skin Model

The "thermal model" of a traditional wind tunnel thin

skin model assumes that all of the heat penetrates the thin skin via conduction to a standard thermocouple gage mounted on the back face. No lateral conduction is assumed and since the emittance of the steel model is low, radiation is assumed negligible. A typical heat rate measurement data point is acquired by injecting the cooled model into the wind tunnel at a known temperature and time and by measuring the temperature at later times. The model is then removed from the tunnel, cooled, and a change in configuration is made in preparation for the next injection and subsequent data point. There are three very severe limitations associated with this technique (Ref 1). The first is the inability to acquire more than one data point during any one injection. The thermal model simply does not allow for the type of change in configuration or model attitude which can be accomplished using dynamic testing techniques (to be discussed later). Associated with this limitation is the long cooling time between each test which significantly increases the cost per data point for the overall test. A second limitation is the special thin skin model which must be fabricated, further contributing to increased test cost. Finally, the assumption of no lateral conduction through the model may in fact be a poor assumption at some critical locations with large curvature. An alternate type of gage, the coaxial thermocouple, can eliminate these limitations with an overall effect of reducing time and cost.

A New Application For An Old Thermocouple

The coaxial thermocouple gage is shown in Fig. 1.1. It consists of a constantan (a metal alloy) jacket surrounding a chromel core with a thin layer of insulation separating the two metals. The coaxial gage is mounted in a steel model thick enough so that the thermal pulse is not sensed on the backface (ie. the model wall is considered a semi-infinite slab). The thermocouple surface is formed when the gage is lightly sanded to match the contour of the model. Some gages are available with backface temperature monitoring to assure that the thermal pulse does not reach the backface in any given run so that an analytical integration of the heat equation can be used to determine the heating rate history. The backface temperature information prior to this investigation is not factored into the data reduction process, however.

Operation of the coax gage is based on uniform conduction along the gage length which would necessitate the model be made of a material with similar thermal properties (Ref 3). The thermal properties of stainless steel match very closely with the gage properties, therefore, the presence of the gage is negligible. The matching of thermal properties also enhances accuracy. A coaxial gage which is matched thermally with the model allows an isothermal wall assumption, whereas other gauges such as calorimeters and thin film gages are not thermally matched, and cause a non-isothermal wall. Measured heat transfer can be in error by

$$\begin{aligned}\{\dot{U}\} &= [A]\{U\} + \{b\} + W(t) \\ \{\dot{S}_k\} &= [A]\{S_k\} + \{d_k\}\end{aligned}\quad (3-1)$$

These equations are solved using a tridiagonal algorithm in subroutine TPS3 for the temperature states, and subroutine SENS for the sensitivity of the temperature to the k th parameter. Propagation of the covariance, P , of the temperature state at each node is accomplished by the approximate difference equation,

$$\begin{aligned}P(t_n^-) &= \phi(\Delta t)P(t_{n-1}^+)\phi^T(\Delta t) \\ &+ \int_{t_{n-1}}^{t_n} \phi(t_n-\lambda)Q\phi^T(t_n-\lambda)d\lambda\end{aligned}\quad (3-2)$$

where ϕ is the transition matrix and where the $-$ and $+$ superscripts are used to denote the expected values before an update (or a priori) and updated (or a posteriori) values, respectively. This calculation is made in subroutine TPSOSP2.

A model of the temperature measurement process must be used for the Kalman filter equations. The measurement equation to identify thermocouple location is,

$$Y(t_n) = H \{U(t_n)\} + \{\mu_n\}\quad (3-3)$$

The updated temperature is calculated by,

$$U(t_n^+) = U(t_n^-) + GE(t_n)\quad (3-4)$$

time. They are found by employing a Kalman filter - a set of recursive equations that optimally combine the propagation of the model equations with measurement updates at each sample time. After the entire temperature - time (state) history has been calculated, a gradient algorithm is used to solve for best estimates of the parameters according to a maximum likelihood criterion.

The second type of estimate consists of the parameters defined in the parameter vector, $\{\theta\}$, as given in Equation 2-10. These parameters remain essentially constant throughout a transient maneuver profile such as a pitch sweep and are estimated based on data from the entire maneuver history. The process of estimating states and parameters is then iterated for convergence to some optimal estimate.

The method used to estimate the states and parameters is formulated from stochastic estimation theory and is known as adaptive estimation. A detailed development of the estimation equations is beyond the scope of this thesis and the reader is referred to References 7 and 8 for more detail.

The thermal model equations for temperature and sensitivity have already been written in the matrix stochastic estimation form of Equation 2-12 as,

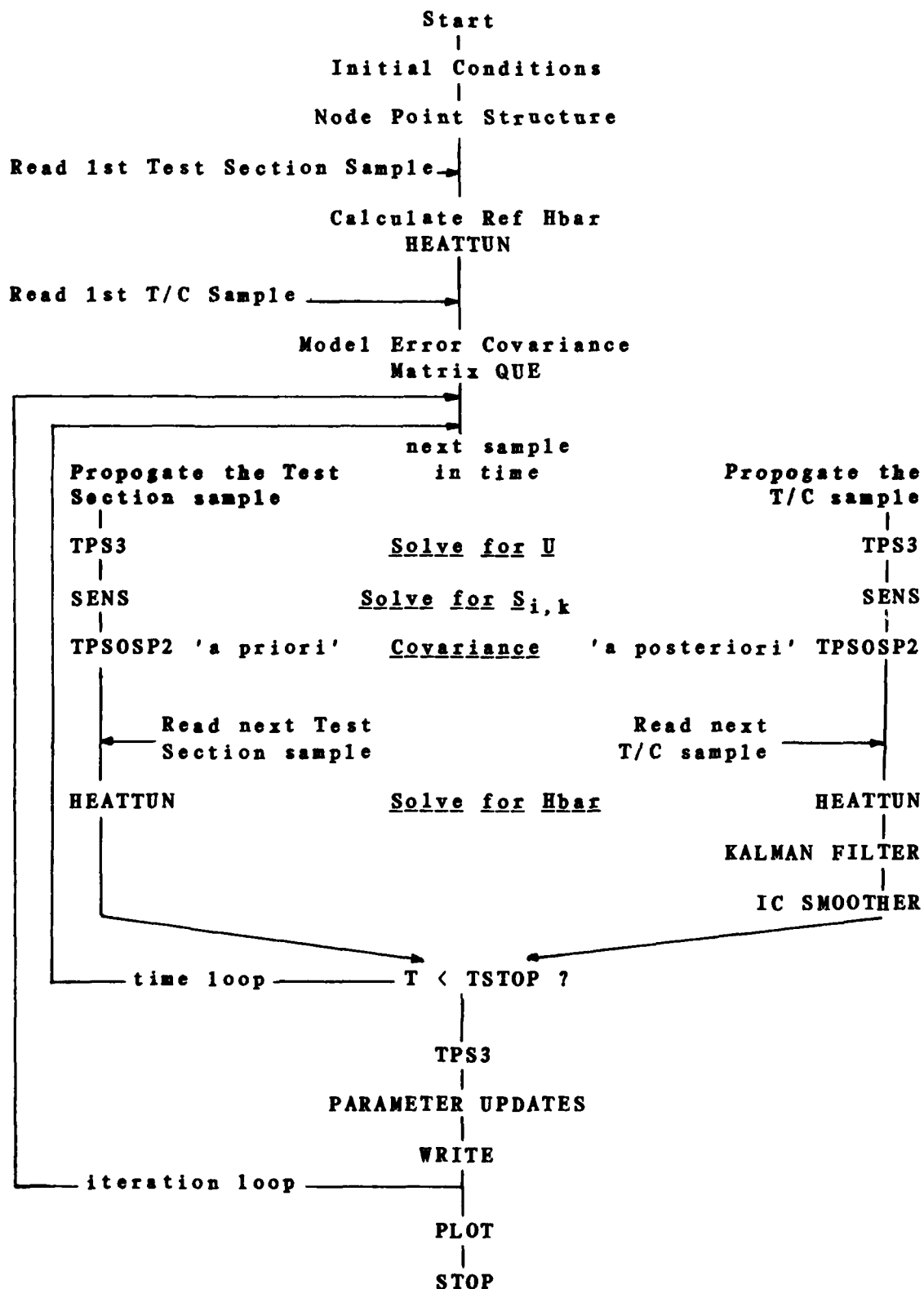


Figure 3.1 HEATEST Algorithm Summary

III. HEATEST OVERVIEW

The HEATEST program was originally developed to determine and model heat rates from the Space Shuttle Orbiter thermocouple data, hence most of its nomenclature references flight data samples and trajectory samples. The wind tunnel equivalence of the trajectory sample would be the test section conditions at the time of the sample (ie. density, velocity, pressure, etc.). The flight data are the thermocouple measurements from the coaxial gages.

An algorithm summary of the HEATEST program is given in Figure 3.1. The initial conditions for the temperature profile, $U(t)$, and the initial covariance, $P(t)$, are specified at the start of the wind tunnel test run. Heating model initial parameters, and the initial reference values for the heating model are read in as inputs to the program. Initial sensitivities of the state are specified to be zero. The node point structure throughout the depth of the thermocouple is then calculated from the input of the length of the gage and the number of node points.

Two types of estimates must be made in order to describe the thermodynamic environment in the wind tunnel. The first type are the state estimates, which are defined by each node temperature. These state estimates are not constant since the temperature varies throughout the maneuver, and hence, must be estimated at each node point in

The {b} vector is not used directly for the covariance equation, but is approximated by an error model given by,

$$Q_{\text{error}} = [h_{\text{bar}} h_{\text{ref}} (T_{\text{aw}} - U_1) \Delta x / \phi_k k]^2 \quad (2-15)$$

where S is the sensitivity. The i subscript identifies the node point and the second subscript identifies the particular parameter number. The sensitivity equations may also be written in the familiar form,

$$[A']\{S_{i,k}^n\} + \{d\} = 0 \quad (2-12)$$

The sensitivity equations are developed and summarized in Appendix B.

2.3 Covariance Equation

Propagation of the covariance of the temperature state at each node requires the equations to be of the form

$$\begin{aligned} \{\dot{U}\} &= [A]\{U\} + \{b\} \\ \text{and } \{\dot{S}_{i,k}\} &= [A]\{S_k\} + \{d_k\} \end{aligned} \quad (2-13)$$

Substituting the definitions of Equations (A-14) into Equations (2-7) and (2-8) and rearranging yields a common tridiagonal [A] matrix for the above equations which is shown presently,

$$[A] = \begin{bmatrix} \frac{-[RM_1 + RP_1 + 4\epsilon\sigma(U_1^{n,s})^3 + h_{bar}h_{ref}]}{RCX_1} & \frac{RP_1}{RCX_1} & 0 \\ & \frac{RM_i}{RCX_i} & \frac{-(RM_i + RP_i)}{RCX_i} \\ & 0 & \frac{RP_i}{RCX_i} \end{bmatrix} \quad (2-14)$$

$$[A']\{U_i^n\} + \{b\} = 0 \quad (2-9)$$

where $[A']$ is an $n \times n$ tridiagonal matrix of material properties and $\{U_i^n\}$ is the n -dimensional column vector of unknown temperature at each node point for each time.

In general, the unknown parameters in this model formulation are the heat transfer coefficient intercept, h_0 , the slopes h_{a1} and h_{a2} , and the scaling parameters for specific heat and thermal conductivity, ϕ_c and ϕ_k , respectively. These parameters may be defined as a vector, θ , of unknown parameters for use in the system identification scheme as,

$$\theta = \{h_0, h_{a1}, h_{a2}, \phi_c, \phi_k\}^T \quad (2-10)$$

The primary purpose of the heating estimation program is to obtain best estimates of these parameters during transient test maneuvers. To estimate these parameters it is necessary to calculate the model sensitivity to each unknown parameter.

2.2 Sensitivity Equations

The derivative of Equation (2-7) with respect to each parameter yields equations of the same form as Equation (2-9) from which the HEATEST program propagates the sensitivity. For example, the sensitivity of the temperature with respect to h_0 would be written as follows,

$$\frac{\partial U}{\partial \theta_1} = \frac{\partial U}{\partial h_0} = S_{h0} = S_{i,1} \quad (2-11)$$

level defined by the superscript n ,

$$\begin{aligned}(U_1^{n,s+1})^4 &= (U_1^{n,s})^4 + 4(U_1^{n,s})^3[U_1^{n,s+1} - U_1^{n,s}] \\ &= -3(U_1^{n,s})^4 + 4(U_1^{n,s})^3 U_1^{n,s+1}\end{aligned}\quad (2-6)$$

Substituting Equation (2-6) into (2-5) yields,

$$\begin{aligned}\frac{\rho \phi_c \Delta x}{2} \frac{U_1^n - U_1^{n-1}}{\Delta t} &= \frac{-\phi_k^{k_1+1/2}}{\Delta x} U_1^n + \frac{\phi_k^{k_1+1/2}}{\Delta x} U_2^n \\ &\quad - \epsilon \sigma [-3(U_1^{n,s})^4 + 4(U_1^{n,s})^3 U_1^{n,s+1} - (U_1^n)^4] \\ &\quad + [h_0 + h_{a1}(\alpha - \alpha_1) + h_{a2}(\alpha - \alpha_2)] h_{ref}(T_{aw} - U_1^{n,s+1})\end{aligned}\quad (2-7)$$

The model equation for the interior node points, $(i=2,imax)$, yields,

$$\begin{aligned}\rho \phi_l c \Delta x \frac{U_i^n - U_i^{n-1}}{\Delta t} &= \frac{\phi_k^{k_i-1/2}}{\Delta x} U_{i-1} \\ &\quad - \frac{\phi_k^{(k_i-1/2+k_i+1/2)}}{\Delta x} U_i \\ &\quad + \frac{\phi_k^{k_i+1/2}}{\Delta x} U_{i+1}\end{aligned}\quad (2-8)$$

Equations (2-7) and (2-8) can be rearranged into the familiar matrix form,

parameters other than those included in the reference heat transfer coefficient are summarized by the static transfer relation or heat transfer coefficient ratio, $h_{bar} = h/h_{ref}$. Here, the ratio is assumed to be piecewise linear with respect to angle of attack as derived from Lagrange Interpolation Theory (Ref 6 and Ref therein).

$$h_{bar} = [h_0 + h_{a1}(\alpha - \alpha_1) + h_{a2}(\alpha - \alpha_2)] \quad (2-4)$$

where h_0 is the magnitude of the heat transfer coefficient, h at the reference conditions, α_1 , specified by the one subscript. The heating parameters h_0 , h_{a1} , and h_{a2} are considered to be unknown and constant over a prescribed time period, and will be estimated by the HEATEST program. The parameters correspond to a derivative with respect to deflection angle of the model. Thus, for constant step size, the model equation at the surface node, ($i=1$), becomes,

$$\begin{aligned} \frac{\rho \phi_c c \Delta x}{2} \frac{U_1^n - U_1^{n-1}}{\Delta t} &= \frac{-\phi_k^{k1+1/2}}{\Delta x} U_1^n \\ &+ \frac{\phi_k^{k1+1/2}}{\Delta x} U_2^n - \epsilon \sigma [(U_1^n)^4 - (U_{\infty}^n)^4] \\ &+ [h_0 + h_{a1}(\alpha - \alpha_1) + h_{a2}(\alpha - \alpha_2)] h_{ref} (T_{aw} - U_1^n) \end{aligned} \quad (2-5)$$

The non-linear radiation term is quasi-linearized on an iteration level defined by the superscript s and by the time

where

- ϵ radiative emissivity
- σ StefanBoltzmann constant
- c material Specific Heat
- ρ material density
- k Thermal Conductivity
- ϕ_c Specific Heat scaling parameter
- ϕ_k Thermal Conductivity scaling parameter

The material specific heat and thermal conductivity are both scaled by the two factors ϕ_c and ϕ_k , respectively, hence the value for c and k will remain unchanged. The parameters ϕ_c and ϕ_k will be estimated by the HEATEST program. Coefficients with subscripts which are less than one or greater than n are zero. The radiation and heat rate terms are also zero except at the surface node. Equation 2-1 includes terms due to conduction from adjacent node points $k_{i-1/2}/\Delta x_{i-1}$, surface radiation $\epsilon\sigma U_i^4$, and the convective transfer of energy as obtained from the heating model. The resulting system of implicit difference equations must be solved simultaneously.

The heating model for the convective transfer of energy is based on Newton's Law of Cooling,

$$q = h(T_{aw} - T) \quad (2-2)$$

Non-dimensionalizing by a reference heat transfer coefficient, h_{ref} yields,

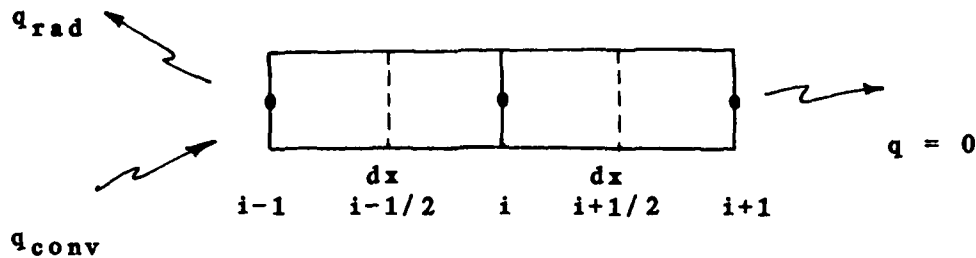
$$q = h_{bar} h_{ref} (T_{aw} - U_i) \quad (2-3)$$

The dependance of the heat transfer coefficient on

II. THE THERMAL MODEL

2.1 Temperature Equations

A cross section of the one-dimensional model is given in Figure 1.1 and below as a typical coaxial thermocouple gage.



An energy balance is performed on each element. The thermal conductivity, k , is taken as an average between each node. Fourier's Law of Heat Conduction throughout the gage, the Stefan-Boltzmann Law for radiation and Newton's Law of Cooling for convection on the surface face yield a system of n nonlinear differential equations of the form:

$$\begin{aligned}
 & [(\rho_i \phi_c c_i \Delta x_i + \rho_{i-1} \phi_c c_{i-1} \Delta x_{i-1}) / 2] [(U_i^n - U_i^{n-1}) \Delta t] \\
 & = \phi_k [k_{i-1/2} / \Delta x_{i-1}] U_{i-1}^n \\
 & \quad - \phi_k [k_{i-1/2} / \Delta x_{i-1} + k_{i+1/2} / \Delta x_{i+1}] U_i^n \\
 & \quad + \phi_k [k_{i+1/2} / \Delta x_{i+1}] U_{i+1}^n \\
 & \quad - \epsilon \sigma (U_i^4 - U_{i-1}^4) + h_{bar} h_{ref} (T_{aw} - U_i) \quad (2-1)
 \end{aligned}$$

gage using a Kalman filter(Ref 2). The purpose of this investigation is to incorporate the appropriate thermal model equations into the HEATEST program for application to coaxial thermocouple gages as used in the wind tunnel. Two reasons for doing this are, 1) replace the analytical with the semi-infinite assumption, ie. extend the run time or shorten the gage, and, 2) estimate thermal properties. Testing and verification of the modified program is necessary for verification of the validity of the results. Simulated data are generated by an analytical solution, and are processed for testing purposes for which the results are known.

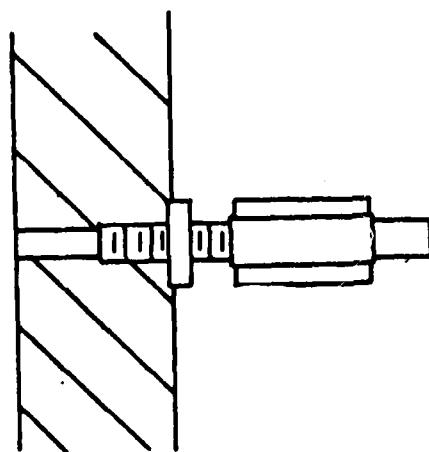
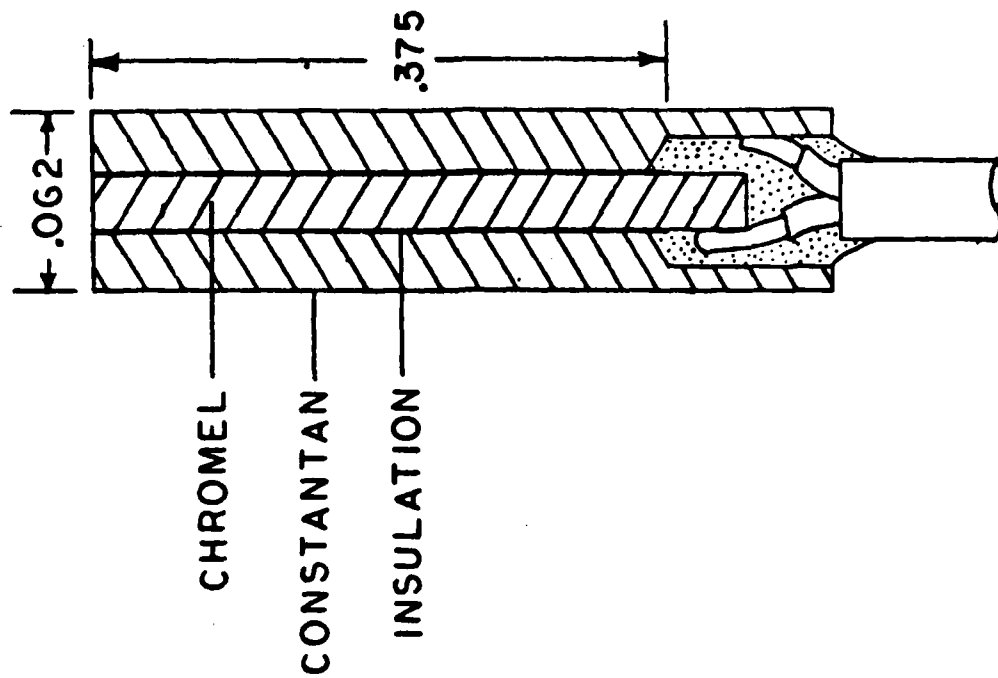
1.3 Overview

A development of the pertinent temperature, sensitivity, and covariance equations will be developed for introduction into the HEATEST program in Chapter II. Details in format for programming may be found in Appendix A & B. Chapter III is an overview of the HEATEST algorithm and shows how the equations developed in Chapter II are utilized. Chapter IV outlines the method for testing the program and offers a discussion of the test cases made and results. Finally, conclusions about the validity of the modifications, and suggestions for further improvement are included in Chapter V.

up to 40% because of the non-isothermal (Ref 2). The same rugged model built for pressure measurements can be used for temperature measurements as well, which would further reduce the wind tunnel costs. The data reduction technique, which uses the temperature time history, eliminates the requirement to fix the model configuration or attitude during any one run, hence dynamic testing techniques may be used similar to the flight test technique used for the Space Shuttle Orbiter (Ref 4). The model may be swept in angle of attack, for example, to determine heat rates as a function of angle of attack. All of these attributes along with a short response time and no required calibration (ref Knox) combine to yield the wind tunnel engineer a tool of marked improvement over previous methods.

1.2 OBJECTIVES

A method of analysis to identify the aerothermodynamic flight environment and update the thermal model of the engineering simulation of the Space Shuttle Orbiter was designed by the Air Force Flight Test Center. This method is in the form of a digital computer program called HEATEST (HEATing ESTimation). The program provides a correlation of the heating as well as a heat rate time history. The program integrates numerically instead of relying on some analytical assumption. It satisfies a maximum likelihood criteria for each parameter and obtains best estimates for the temperature at discrete nodes throughout the length of the



MODEL TCS
COAXIAL PROBE MOUNTED
IN A METAL WALL

FIGURE 1.1 COAXIAL THERMOCOUPLE GAGE

where

$$G = P(t_n^-)H^T[HP(t_n^-)H^T + R_m]^{-1}$$

$$E = Y(t_n) - HU(t_n^-)$$

The updated sensitivities are calculated by,

$$S_k(t_n^+) = [I-GH]P(t_n^-)[I-GH]^T + GR_mG^T \quad (3-5)$$

The updated covariance is calculated by,

$$P(t_n^+) = [I-GH]P(t_n^-)[I-GH]^T + GR_mG^T \quad (3-6)$$

To alleviate the problem of imprecise initial conditions, a fixed point smoothing algorithm has been added to the HEATEST program. Details of the smoother and its effects in the adaptive estimation scheme may be found in Reference 6.

Finally, the best estimates of the parameters are then estimated at the end of a specified time segment by the gradient algorithm,

$$\theta^* = \theta - [\partial^2 F / \partial \theta^2]^{-1} \partial F / \partial \theta = \theta + J^{-1}S \quad (3-7)$$

where,

$$J_{i,j} = \sum_{n=1}^N S_{i,k}(t_n^-)H^T[HP(t_n^-)H^T + R_m]HS_{i,k}(t_n^-)$$

$$S_k = \sum_{n=1}^N S_{i,k}(t_n^-)H^T[R_m]^{-1}[Y(t_n) - HU(t_n^-)]$$

The matrix J is an approximation for the Jacobian or conditional information matrix and is given in component

form by $J_{i,k}$. The score vector, S_k , is used to approximate the gradient of the likelihood function for a large number of time samples.

Using these equations, best estimates for the temperature time history (states) at each node can be found, as well as the deviation in temperature as provided by the covariance matrix. Also, an estimate of the parameter uncertainty is provided by the Cramer-Rao bound. The Cramer-Rao bound relates the conditional information matrix to the covariance of the parameter estimate.

IV. RESULTS

4.1 Test Procedure

To test the validity of the program modifications, a set of contrived data was generated. It's development assumes that the heat rate due to convection at the surface node is constant and equal to the heat rate due to conduction at the surface. The heat rate due to convection is given by,

$$q = h(T_{aw} - T_w)$$

or

$$q = h_{bar} h_{ref} (T_{aw} - T_w) \quad (4-1)$$

where h_{bar} is defined as in Equation A-12. The equation for the heat rate due to conduction assuming a one-dimensional, homogenous, semi-infinite solid is as follows (Ref 9),

$$q = \frac{(\rho c k)^{1/2}}{\pi} \int_0^t \frac{dT_w(\tau)}{d\tau} \frac{d\tau}{(t-\tau)^{1/2}} \quad (4-2)$$

where t = time from start of heating
 $T(t)$ = surface temperature rise
 τ = dummy variable of integration

Equating Equations 4-1 and 4-2 yields,

$$q = h_{\text{bar}} h_{\text{ref}} (T_{\text{aw}} - T_{\text{w}}) = \frac{(\rho c k)^{1/2}}{\mu} \int \frac{dT_{\text{w}}(\tau) d\tau}{(t-\tau)^{1/2}} \quad (4-3)$$

Two different expressions for the derivative of the wall temperature with respect to time were used. The first implied a linear change in temperature with respect to time yielding a constant for dT_{w}/dt and the second expression assumes that temperature was a quadratic function of time as shown,

<u>Linear</u>	<u>Quadratic</u>
$T_{\text{w}} = bt + c$	$T_{\text{w}} = at^2 + bt + c$
$\frac{dT_{\text{w}}}{dt} = b$	$\frac{dT_{\text{w}}}{dt} = 2at + b$

Solving Equation 4-3 for T_{aw} so that h and q are constant and after making the indicated substitutions and integrating yields,

Linear assumption

$$T_{\text{aw}} = T_{\text{w}} + \frac{2b}{h_{\text{bar}} h_{\text{ref}}} \sqrt{\frac{\rho c k t}{\pi}} \quad (4-4)$$

Quadratic assumption

$$T_{\text{aw}} = T_{\text{w}} + \frac{2}{h_{\text{bar}} h_{\text{ref}}} \sqrt{\frac{\rho c k t}{\pi}} \left(b + \frac{4at}{3} \right) \quad (4-5)$$

A short computer program was written to produce a temperature-time history in the data tape format for the HEATEST program. For the above equations, h_{ref} and h_{bar} (the estimated parameter) were set equal to 1 and the coefficients a , b , and c , were selected to yield reasonable values for T_w .

4.2 Test Cases

The reference test case was taken to be a 10 sec. simulated wind tunnel test run using the linear data provided from the previous section. Thermocouple samples and test section samples were provided at the rate of one sample per second. The objective was to examine the rate of convergence of the temperature states and to estimate the first parameter, h_0 . Recall from Section 4.1 that the data was generated to yield a value of one for h_0 . Also of interest, was the validity of the model to the semi-infinite solid assumption (ie. no change in the temperature at the back face node throughout any specified time segment).

The input data is shown in Figure 4.1 and is generated digitally depending upon a desired time step (Δt). The initial temperature throughout each gage is assigned a value of 60°F. Figure 4.2 shows the input temperature values for a $\Delta t = 1$ sec. as used in the reference test case.

Figure 4.3 identifies the temperature state at the 2 sec.(lowest curve), 6 sec.(middle curve), and 10 sec.(top curve) times following the first iteration through the

updated HEATEST program. It clearly shows that at the back face node, the semi-infinite solid assumption used to derive the data is violated. This is indicated by the change in backface (node 6) temperature with time. Note also, however, that the temperature gradient at the back nodes (between nodes 5 and 6) is zero due to the adiabatic wall assumption. It should also be pointed out that surface node temperature response to the given input was immediate with no time lag.

Figure 4.4 is similar to Figure 4.3 except the temperature states and parameter estimates have been iterated to convergence, in this case, three times. Overlaying the two figures shows no perceptible difference between them, and the data shows no variations in values until after the second decimal point. In spite of the response of the back face node which would ordinarily invalidate the test, the estimated value for h_0 was .99598, within .4% of the desired value of 1! Several test cases will be compared to this reference by examining changes in time step, thermocouple length, and the number of node points. Also, an examination of the ability of the program to estimate the other parameters, follows.

4.2.1 Changes in Time Step

Figures 4.4, 4.5, and 4.6 show temperature states at 2, 6, and 10 sec. for Δt equal to 1, .5, and .25 sec., respectively. All three curves required three iterations

for convergence. The curves are all similar in shape with almost no perceptible differences. However, if the 10 sec. curves from $\Delta t = 1$ sec. and $\Delta t = .25$ sec. are overlayed as in Figure 4.7, a small difference may be noted at the backface nodes indicating that, indeed, a decrease in time step will yield a profile which will more closely approximate the model.

4.2.2 Changes in Thermocouple Length

Changes in thermocouple length offered the most dramatic changes in temperature state as can be seen in Figures 4.8, 4.9, and 4.10 where the lengths range from .1 ft., .05ft., and .025 ft., respectively. The two longer lengths converged within three iterations while the short thermocouple length took four iterations to converge. The extra iteration is most likely due to the large deviation from the model and the large differences in temperature state from one time step to the next. Figure 4.11 compares the temperature state of each thermocouple length after the 10 sec. run. The lowest curve is associated with the longest thermocouple, and the upper curve is associated with the short gage.

4.2.3 Changes in Number of Node Points

For this comparison, a 10 sec. run with 6 node points and a 3 sec. run with 6 node points (Figures 4.12 and 4.13) will be compared with a 10 sec. run with 12 node points and

a 3 sec. run with 12 node points (Figures 4.14 and 4.15). Each of the test runs were converged by the third iteration. At both of the different run times, increasing the number of nodes yielded a solution which more closely approximates the model (ie. the semi-infinite solid at the back face node) and gave correspondingly better estimates for h_0 . The comparison of temperature states is better represented by Figures 4.16 and 4.17 which directly compares 6 nodes and 12 nodes interspersed evenly throughout the .05 ft. long thermocouple for 10 sec. and 3 sec. run times, respectively.

4.2.4 Parameter Estimation

The ability of the algorithm to estimate h_0 has already been discussed. To summarize, even when the output temperature states clearly violate the semi-infinite solid assumption used in generating the data, the estimated values of h_0 remain within 15%. The 15% is a worst case number derived from the short thermocouple using a course grid for a long run time. An average deviation which considers all of the test cases evaluated is closer to 3%.

To estimate ϕ_k , an erroneous data value was given for K_{data} , wherein the program iterated to a value for ϕ_k which, when multiplied by the erroneous K_{data} , would yield the correct value, ie. $\phi_k K_{data} = K_{correct}$. The erroneous K_{data} which was input into the program was 14% in error of the $K_{correct}$ value. The value provided by the program for ϕ_k when multiplied by K_{data} yielded a value within .9% of

the K_{correct} value!

The estimate of ϕ_c was not as successful, however. After 6 iterations, the solution was diverging from the expected value. A suspected sign error in the ϕ_c sensitivity calculation is the most likely cause.

The quadratic input data was used as a comparison to the linear data to challenge the algorithm, ie. the more complicated the input the more difficult the estimation process. No direct comparison may be made of temperature, however, since the input data is different. The parameter estimation of h_g using the quadratic input data was still excellent yielding .1%, while the estimate using linear data was somewhat better at .05%.

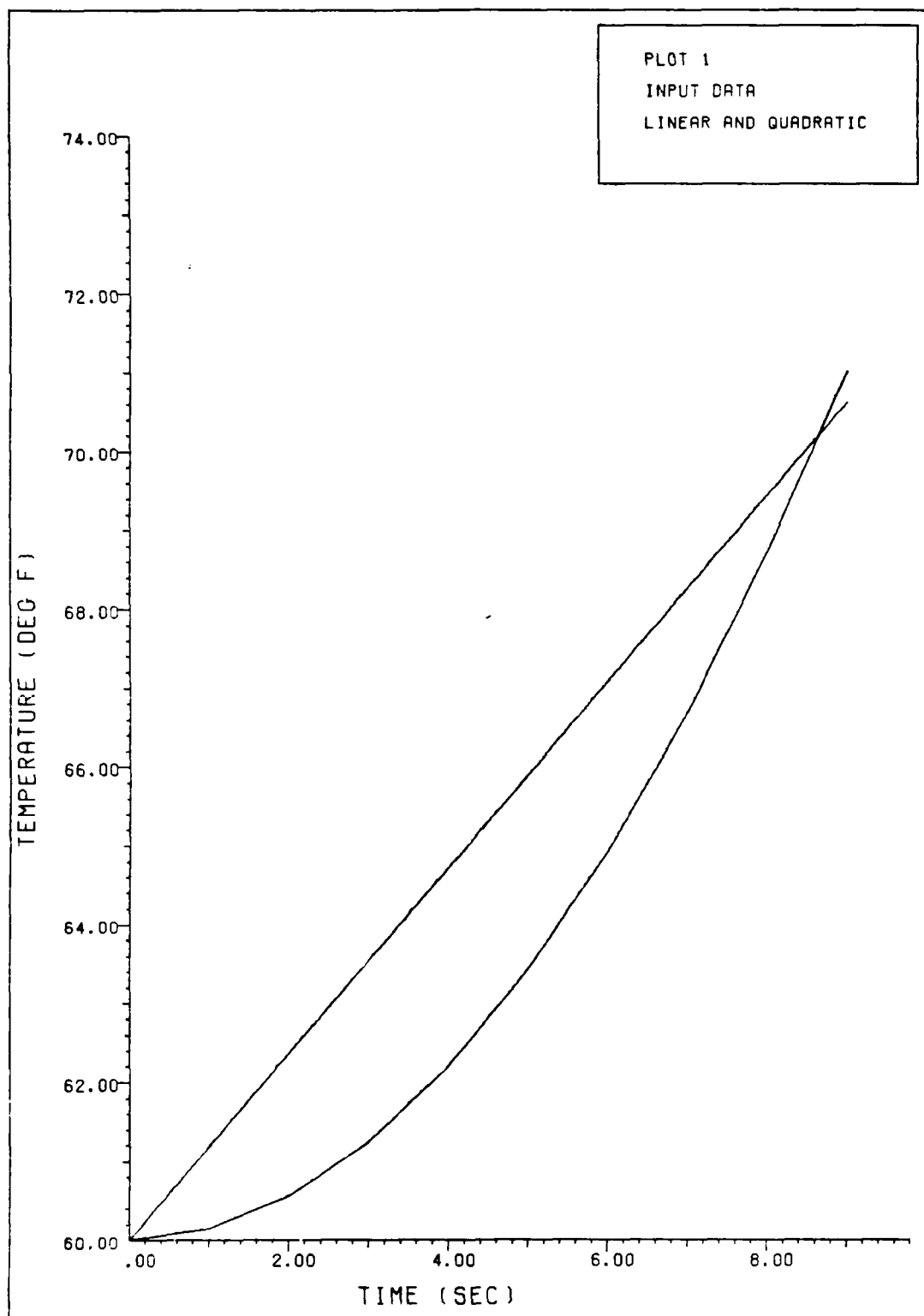


FIGURE 4.1

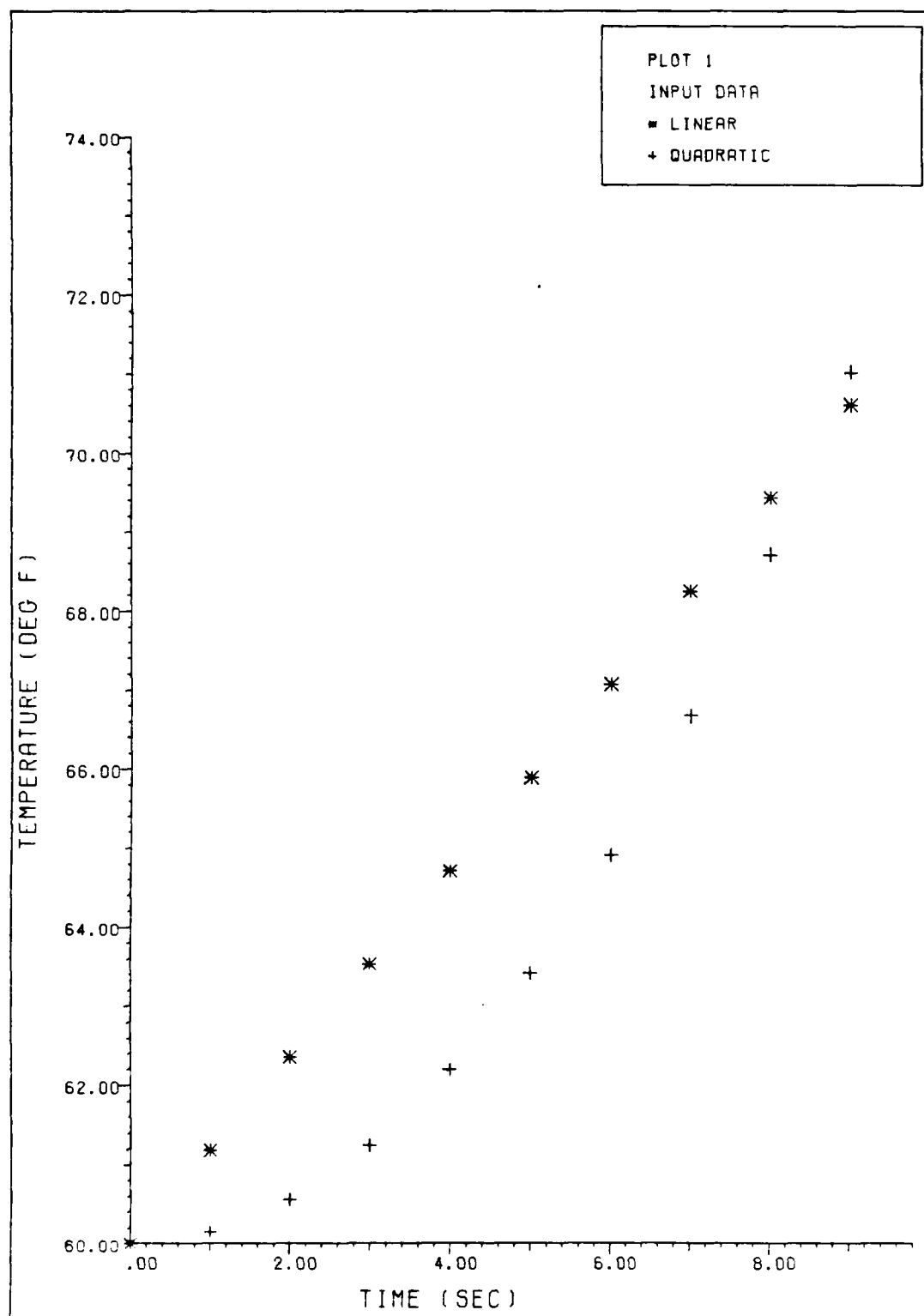


FIGURE 4.2

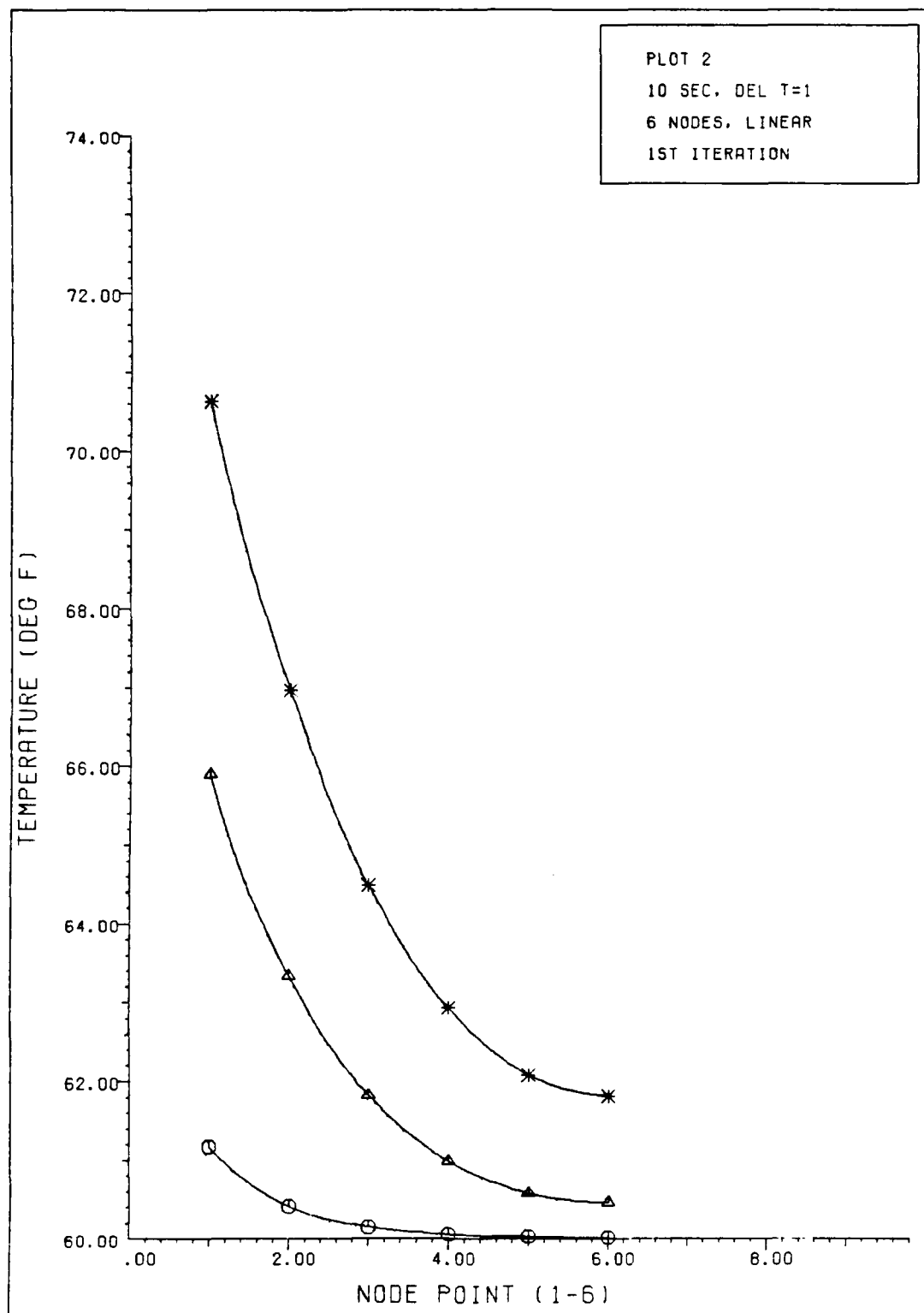


FIGURE 4.3 TEMP VS NODE POINT HISTORY (2,6,10 SEC)

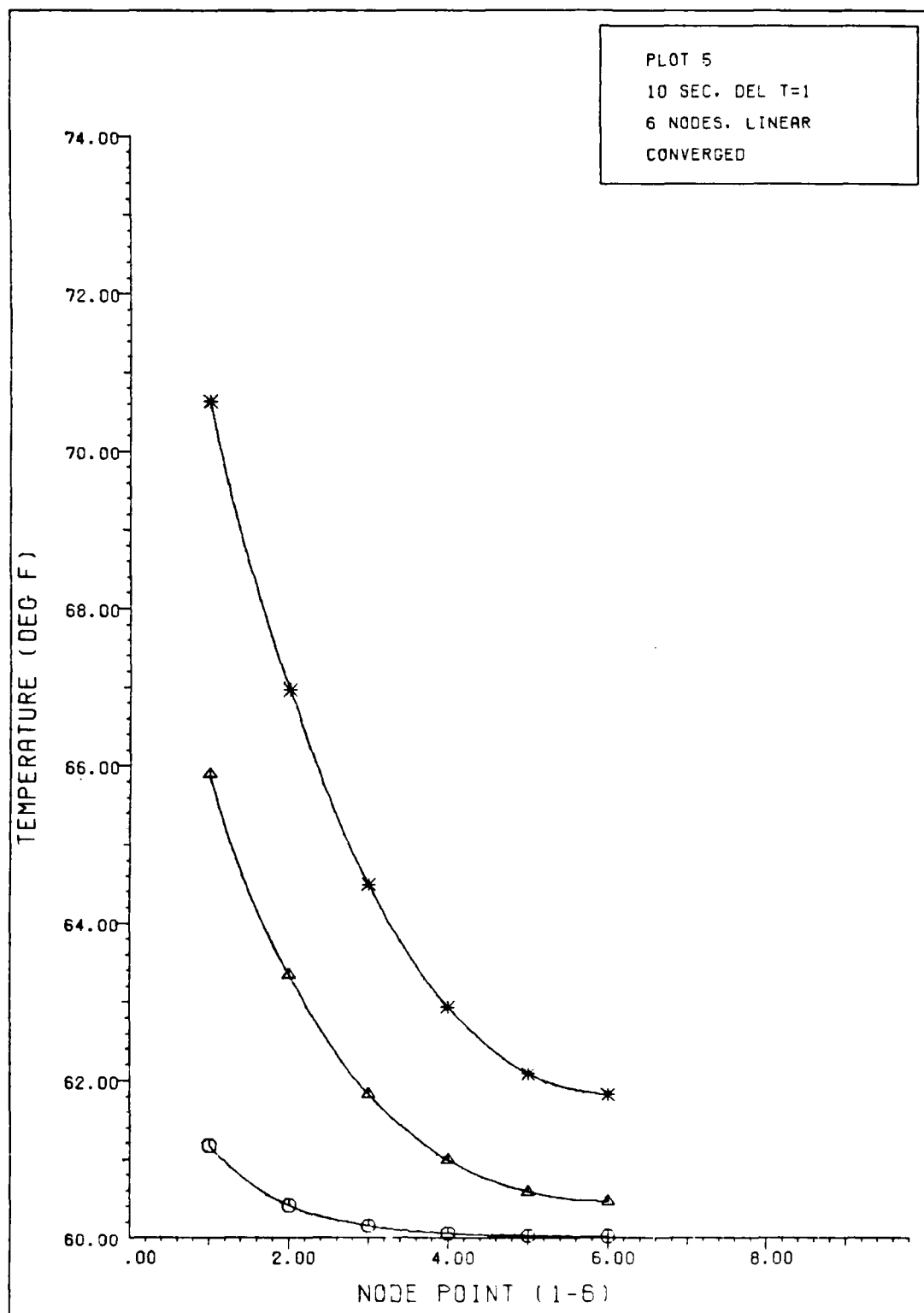


FIGURE 4.4 TEMP VS NODE POINT HISTORY (2,6,10 SEC)

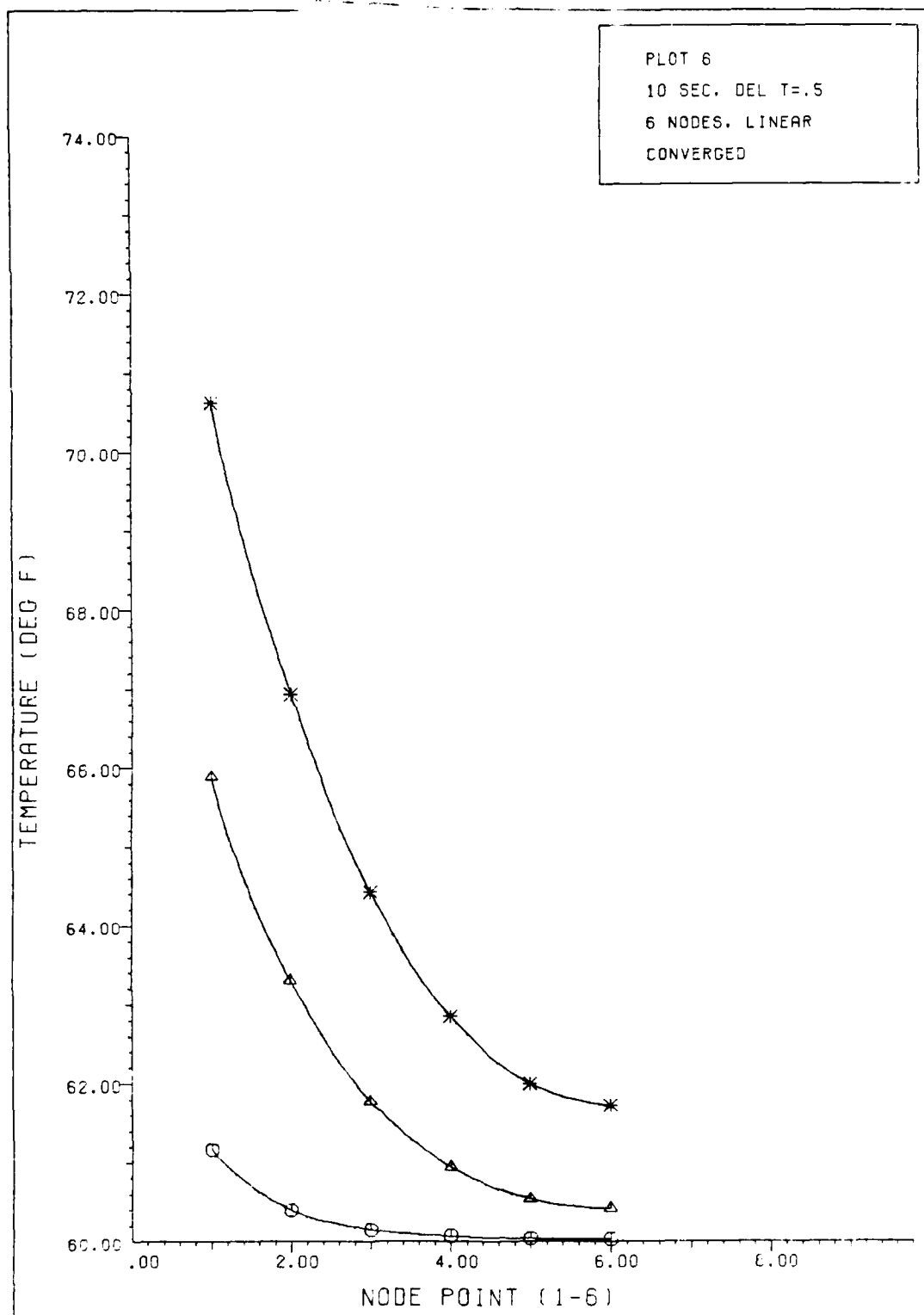


FIGURE 4.5 TEMP VS NODE POINT HISTORY (2.6, 10 SEC)

the utility of the program to be used in wind tunnel runs of much longer duration.

5.2 Recommendations

Actual wind tunnel test data from coaxial gages needs to be analyzed by the program to instill more confidence in the results. This would require that the data tape format be modified to be compatible with the program inputs.

Also, a prescribed model for h_{bar} as a function of angle of attack needs to be input to determine the ability of the program to estimate the piecewise linear derivatives, h_{a1} and h_{a2} . The additional input data would then be the wind tunnel model angle of attack at each thermocouple sample time.

Another potential modification to the program would be to use a second order time derivative approximation as opposed to the current first order approximation. It is suspected that the increased accuracy would improve the state estimates particularly for large time gaps in thermocouple data.

The temperature state estimates might be improved by incorporating a variable grid. An exponential grid generation scheme would concentrate node points near the surface where the largest temperature gradients exist.

A final recommendation, would be to make the program more 'user friendly' and to publish documentation much like a users manual.

V. CONCLUSIONS AND RECOMMENDATIONS

5.1 Conclusions

The program works for this simplified test case of known parameters and number of node points. The ability of the program to determine the temperature state appeared very good but definite conclusions cannot be drawn without comparing this data to actual thermocouple data. The ability of the program to estimate the parameters, h_0 and ϕ_k was demonstrated extremely well. Re-examination of the ϕ_c equations must be made before further conclusions may be made about the ability of the program to estimate this parameter. It is theoretically possible, however.

The objective, to validate the use of the modified HEATEST program for use of coaxial thermocouple gages on wind tunnel test articles, has been met for the special case of this analytical model. The program is capable of determining temperature states and estimating parameters with a high degree of accuracy. A note concerning the semi-infinite slab assumption needs to be emphasized. This assumption was made only for the data generation program to yield data for which the analytical solution was known. As mentioned in Chapter 1, some coaxial gages are available with backface temperature monitoring which would also provide another temperature measurement to enhance estimation of ϕ_k and ϕ_c . This feature would greatly enhance

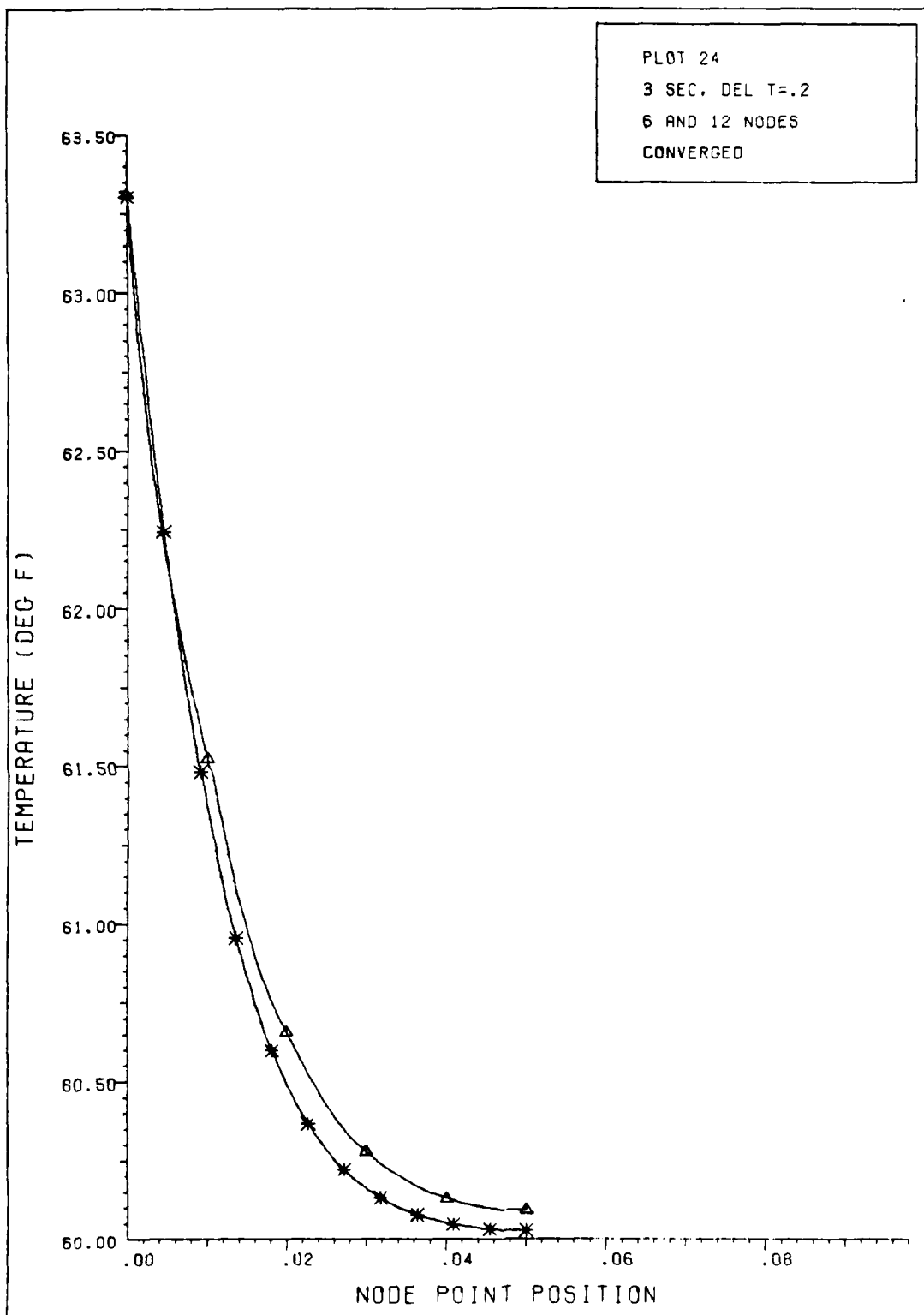


FIGURE 4.17 TEMP VS NODE POS. (6 NODES VS 12 NODES)

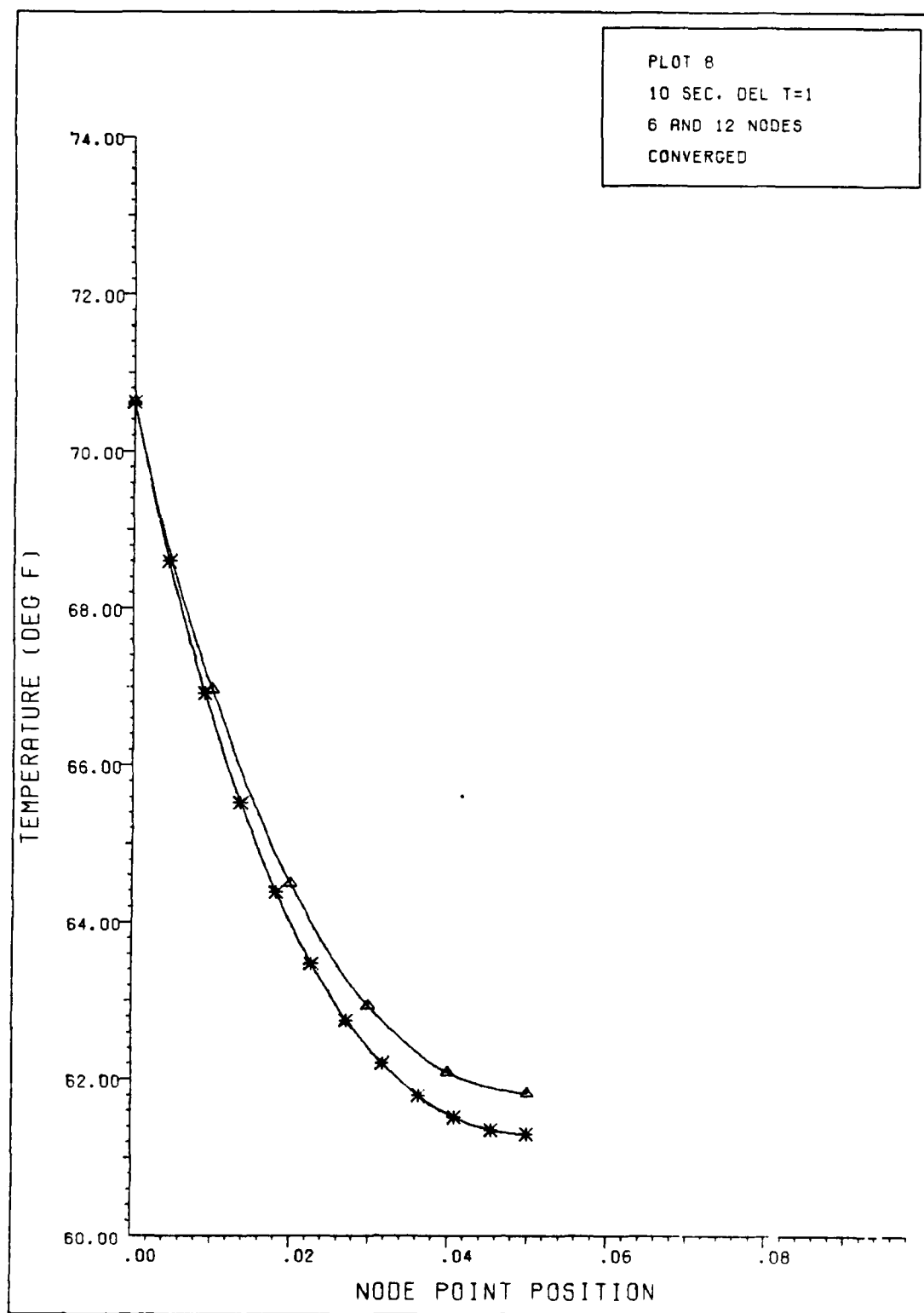


FIGURE 4.16 TEMP VS NODE POS.(6 NODES VS 12 NODES)

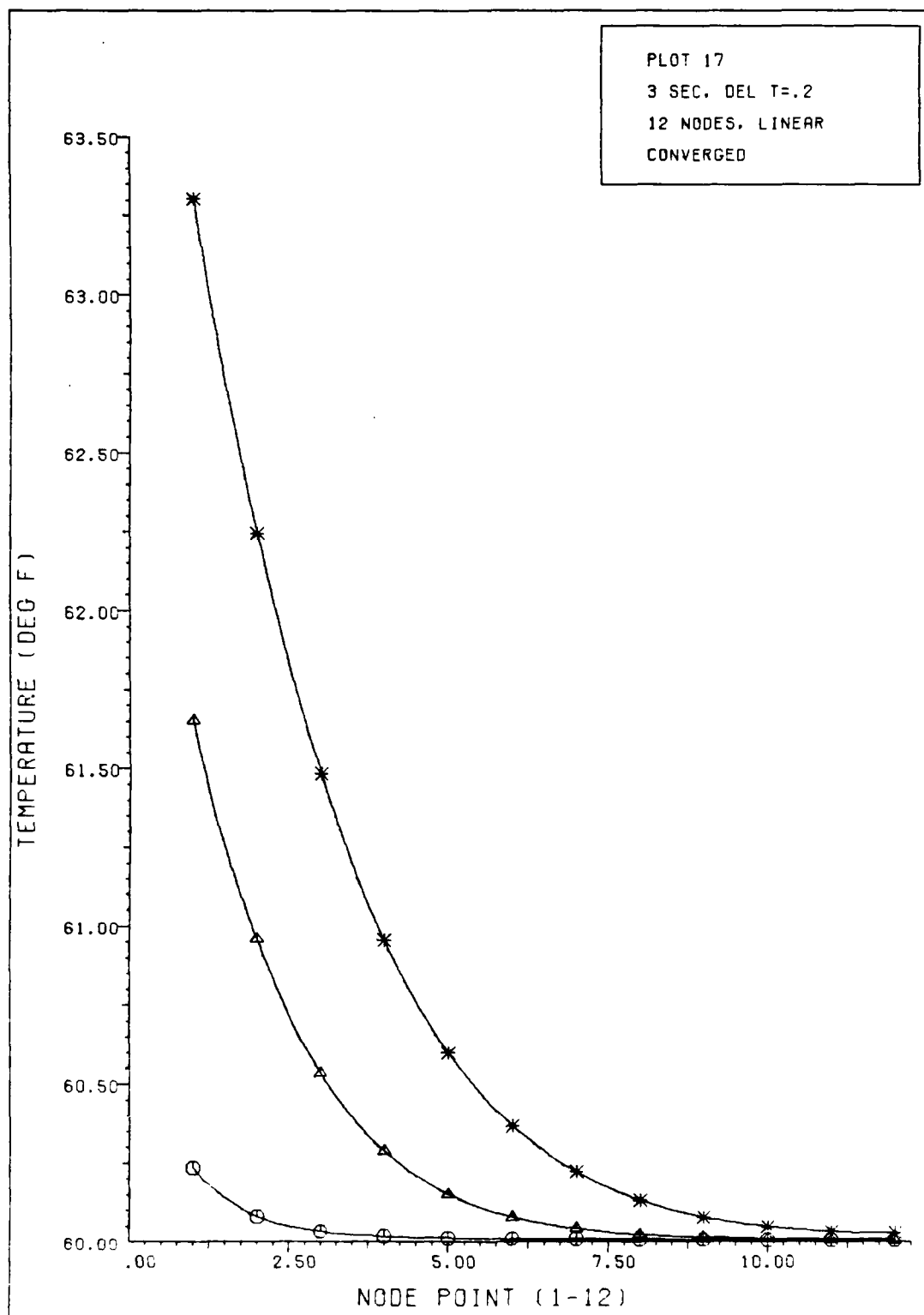


FIGURE 4.15 TEMP VS NODE POINT (.4, 1.6, 3.0 SEC.)

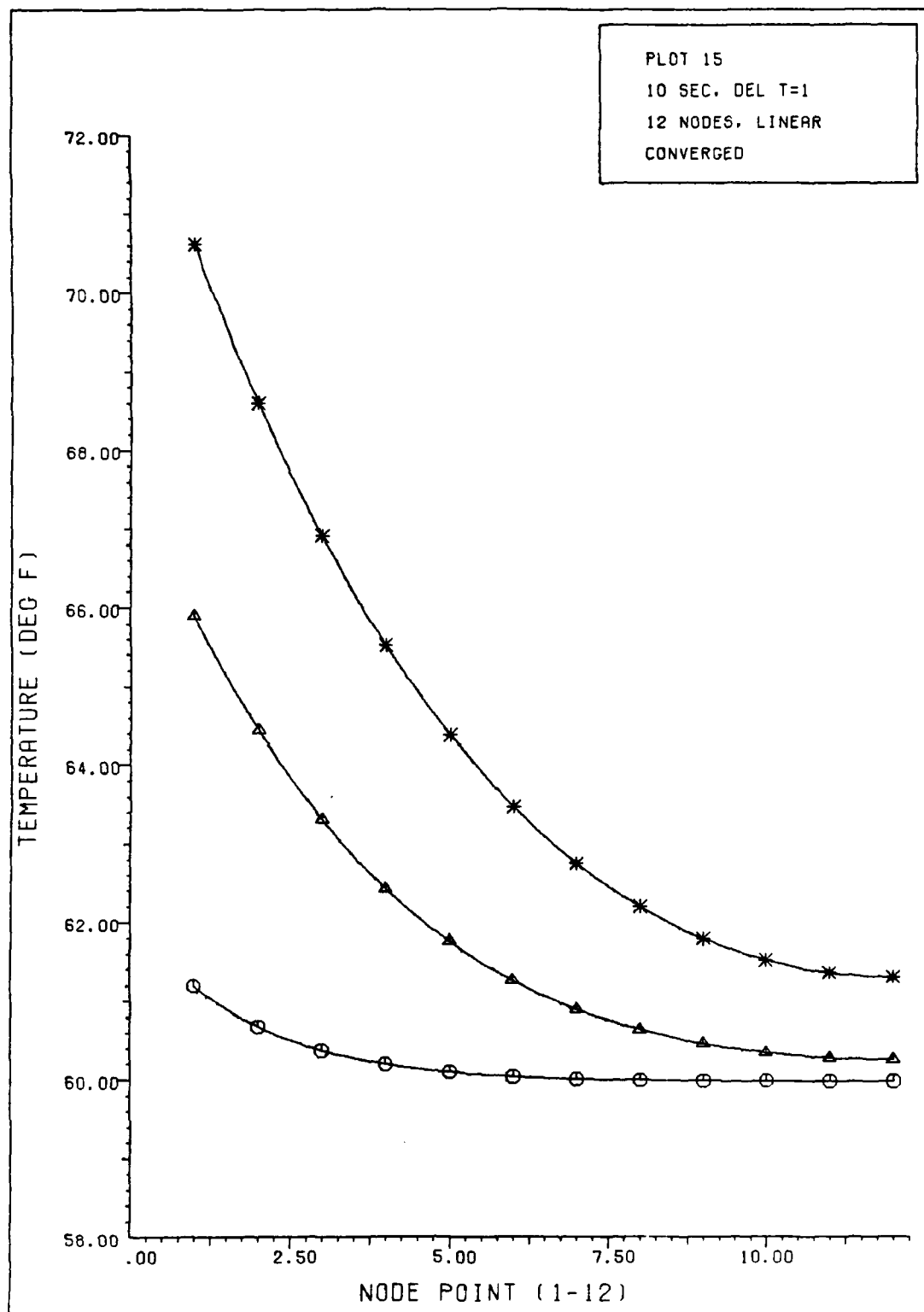


FIGURE 4.14 TEMP VS NODE POINT HISTORY(2.6.10 SEC)

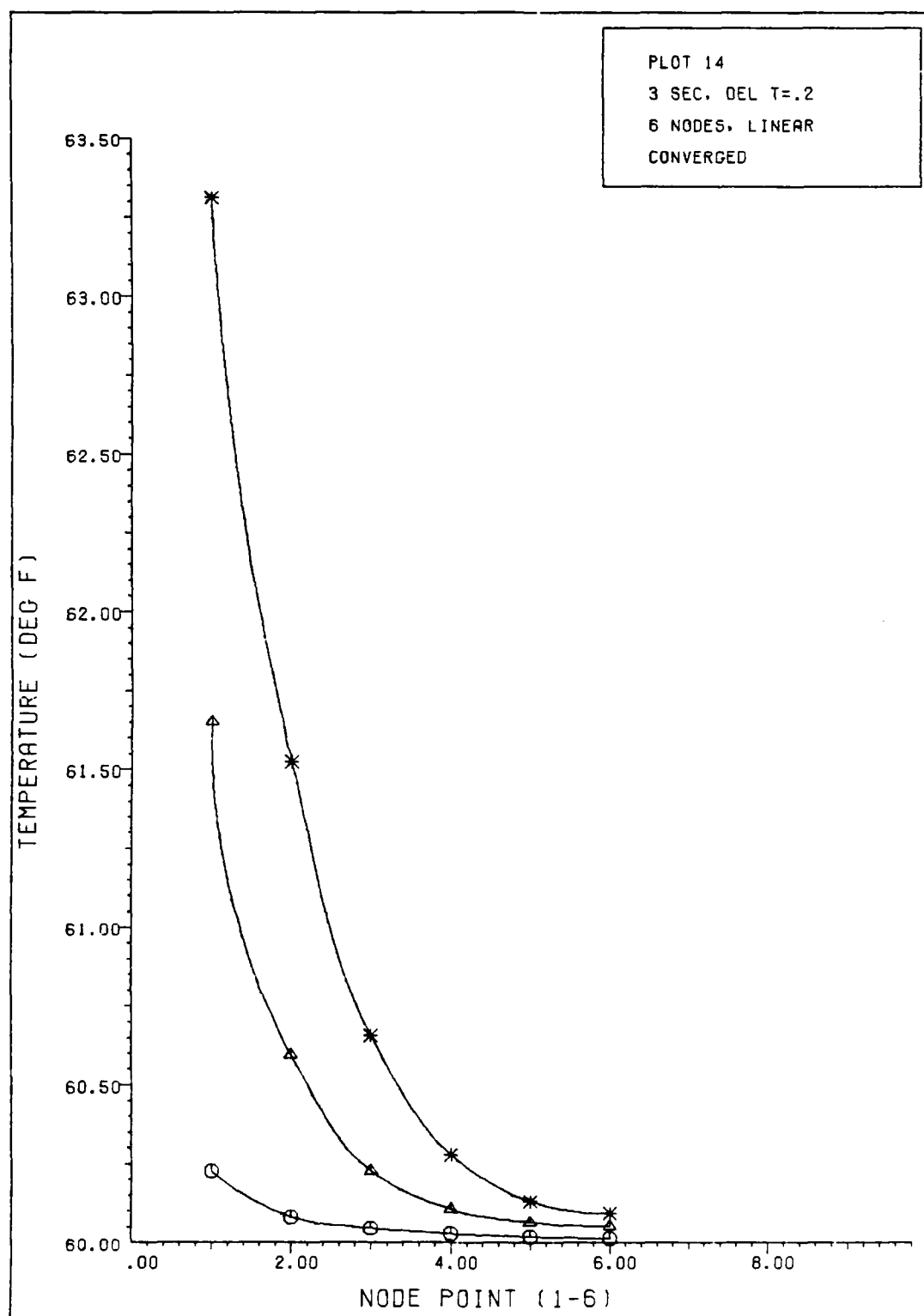


FIGURE 4.13 TEMP VS NODE POINT (.4, 1.6, 3.0 SEC)

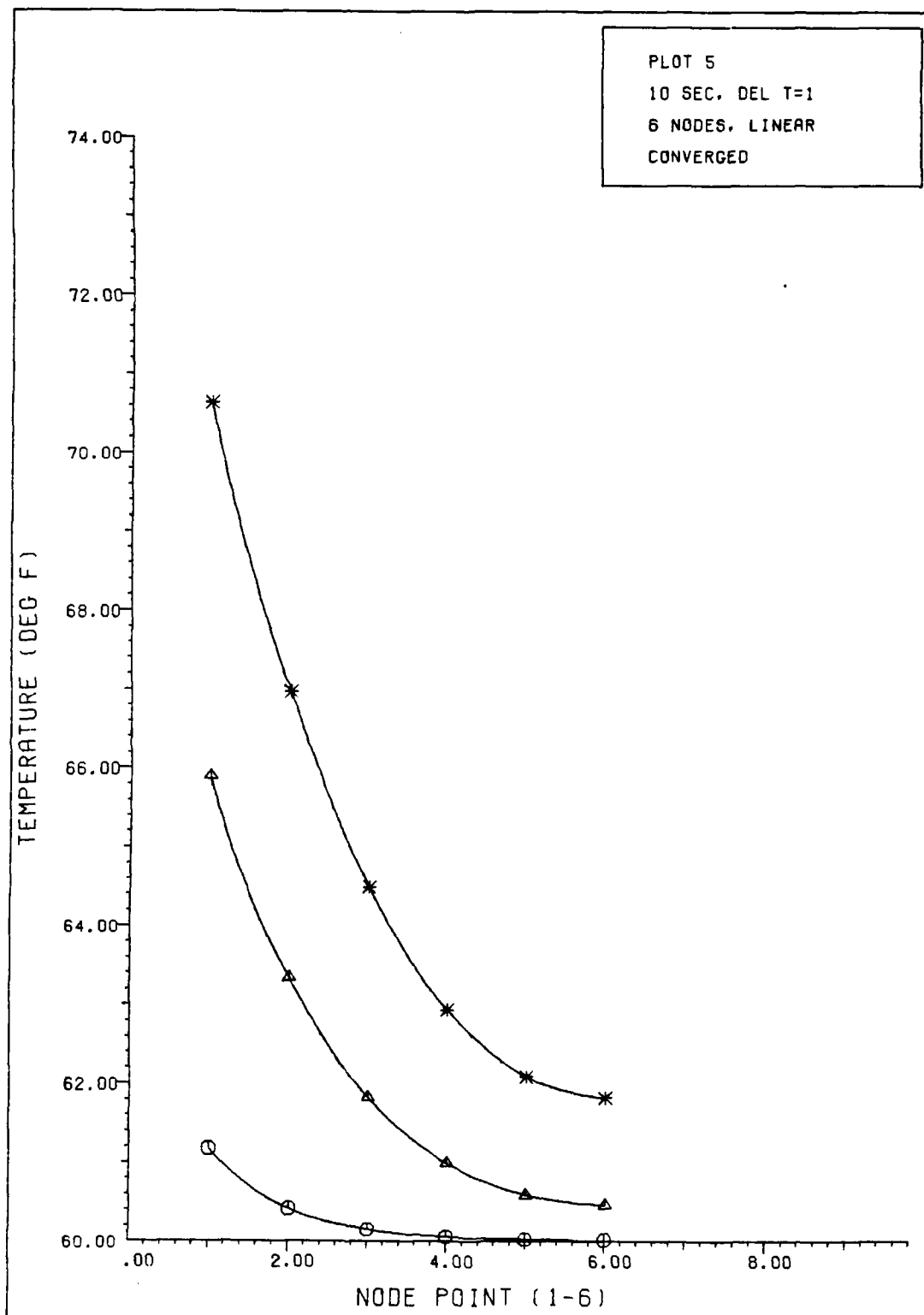


FIGURE 4.12 TEMP VS NODE POINT HISTORY(2.6,10 SEC)

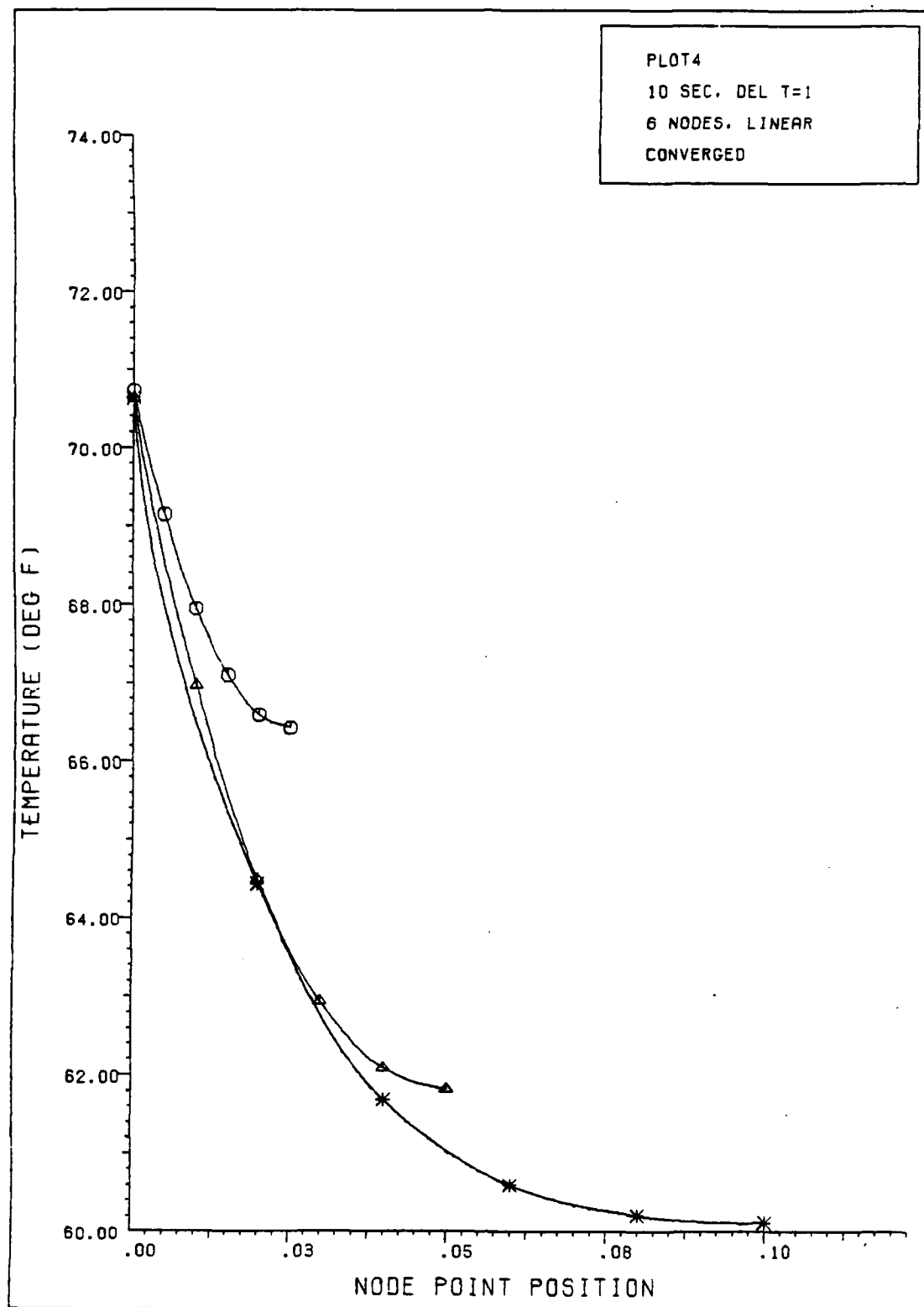


FIGURE 4.11 TEMP VS POS. (T/C LENGTH=.1,.05,.025FT)

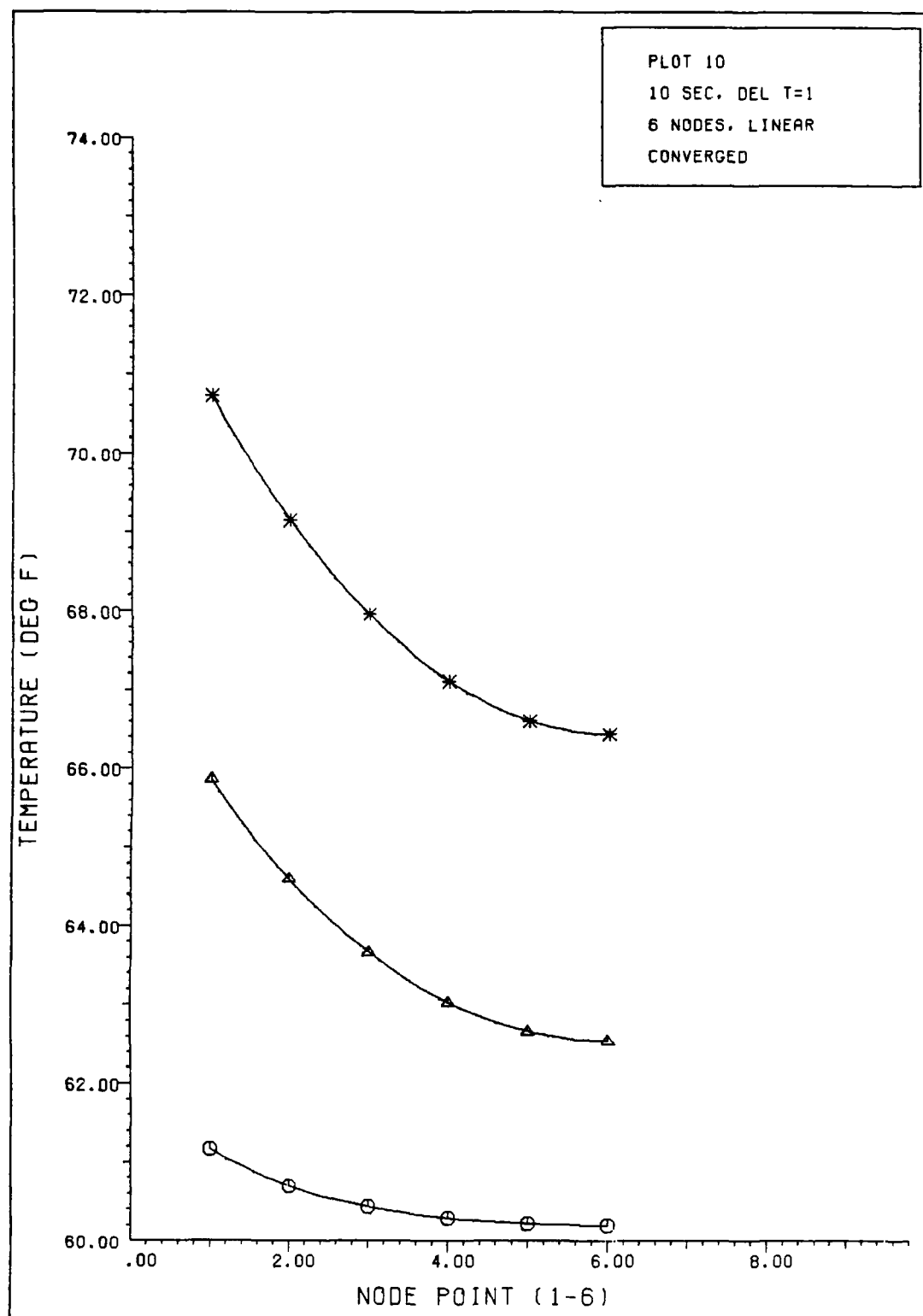


FIGURE 4.10 TEMP VS NODE POINT (T/C LENGTH=.025FT)

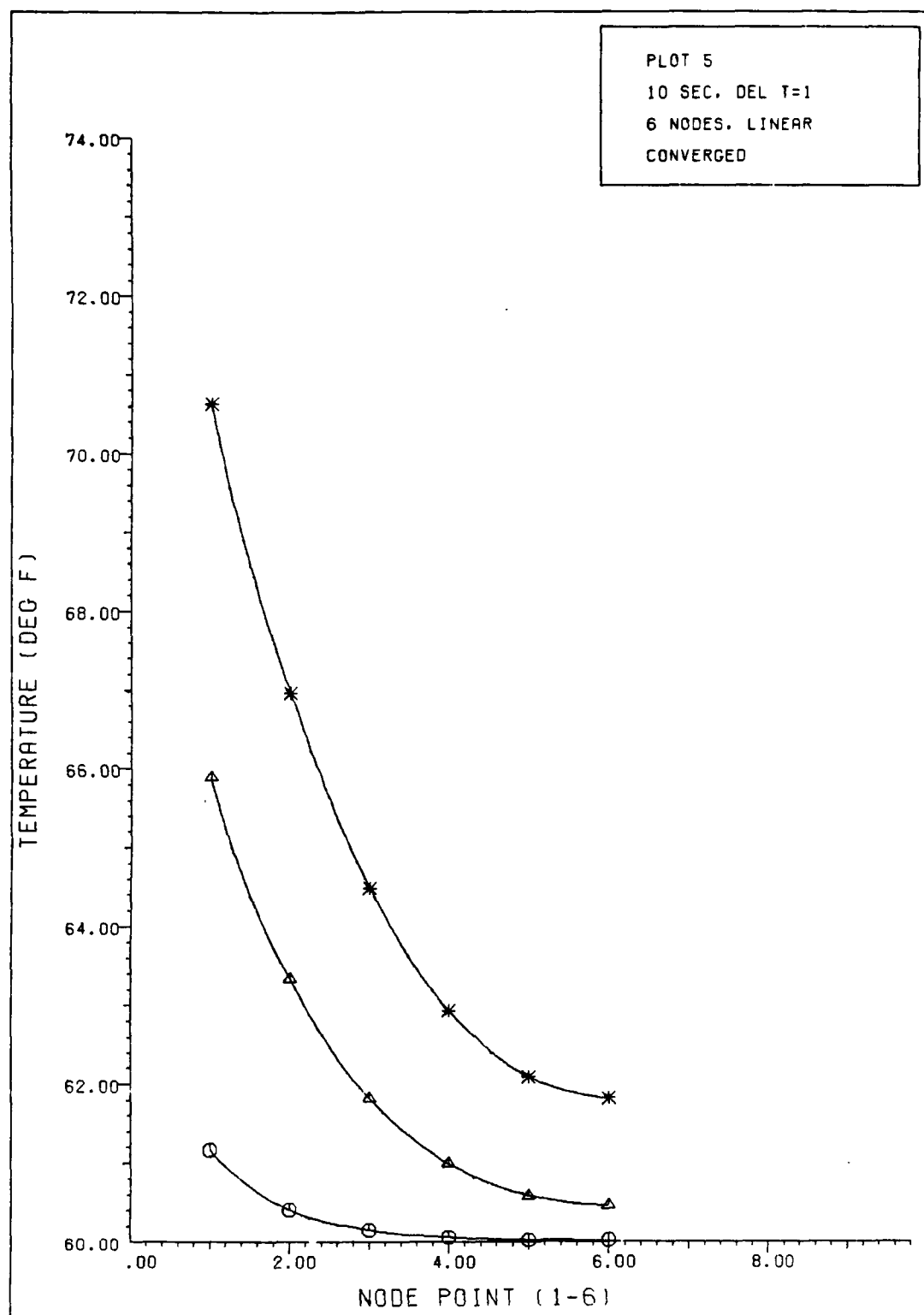


FIGURE 4.9 TEMP VS NODE POINT (T/C LENGTH= .05 FT)

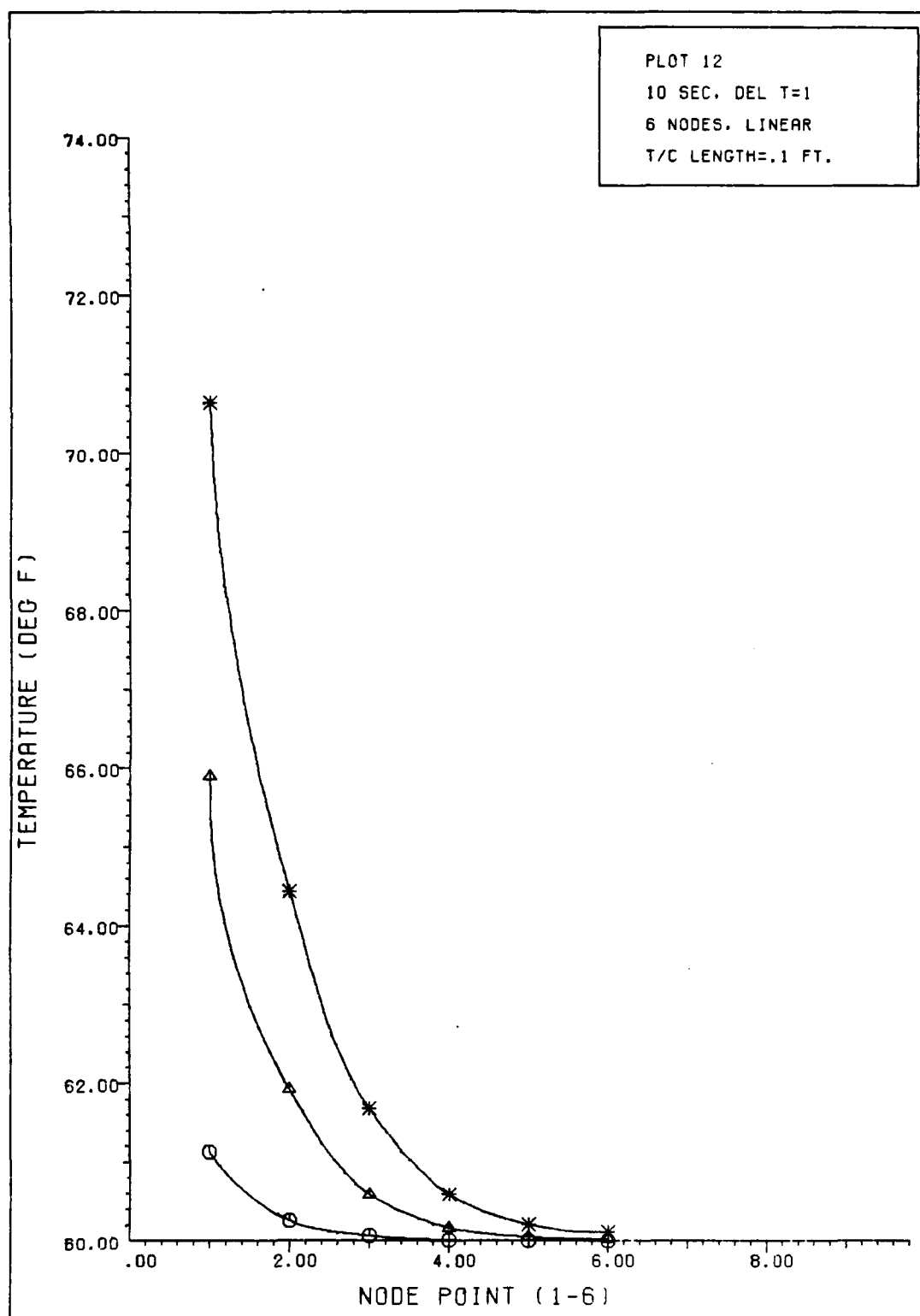


FIGURE 4.8 TEMP VS NODE POINT (T/C LENGTH= .1 FT.)

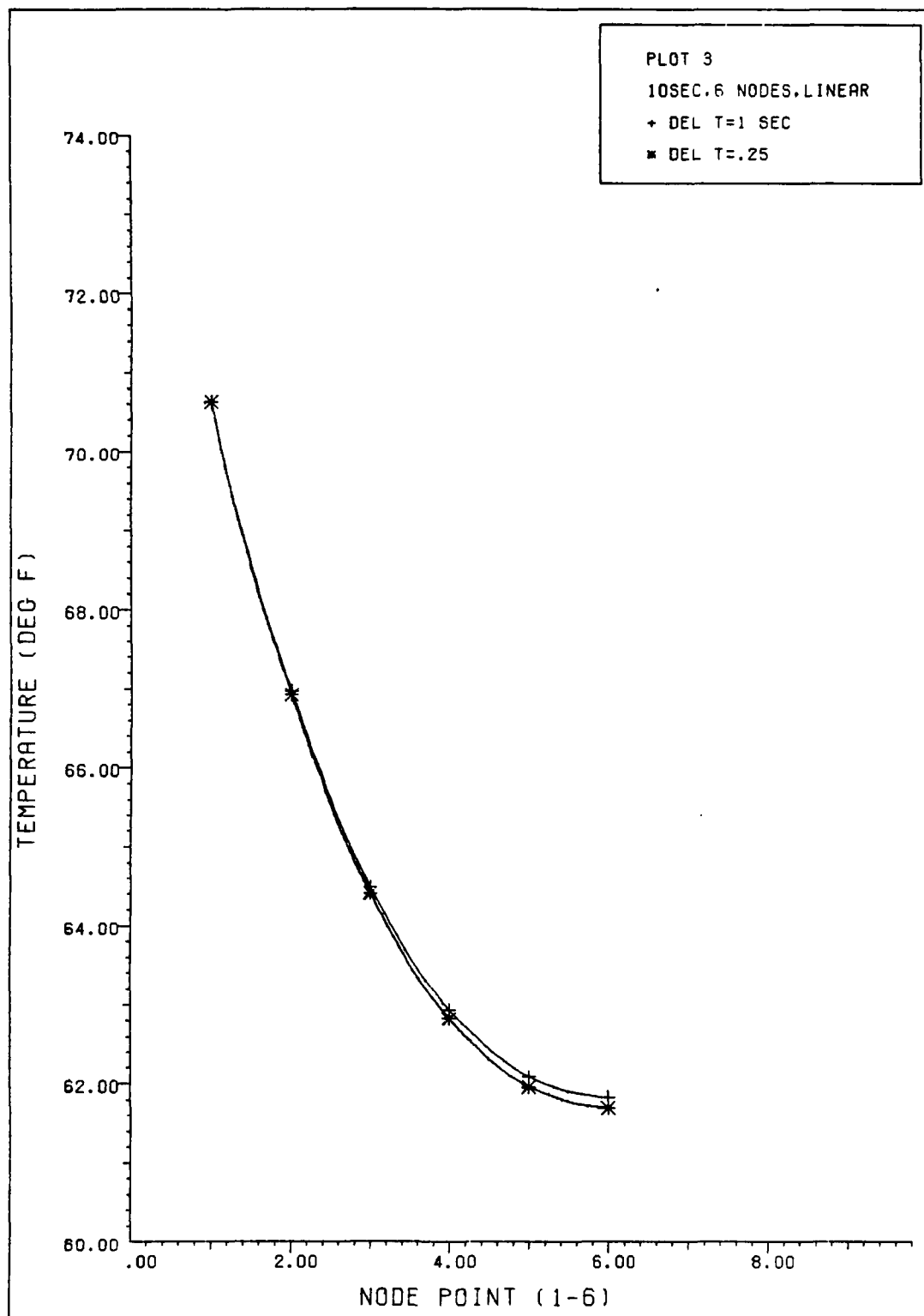


FIGURE 4.7 TEMP VS NODE POINT(DEL T=1 AND .25 SEC)

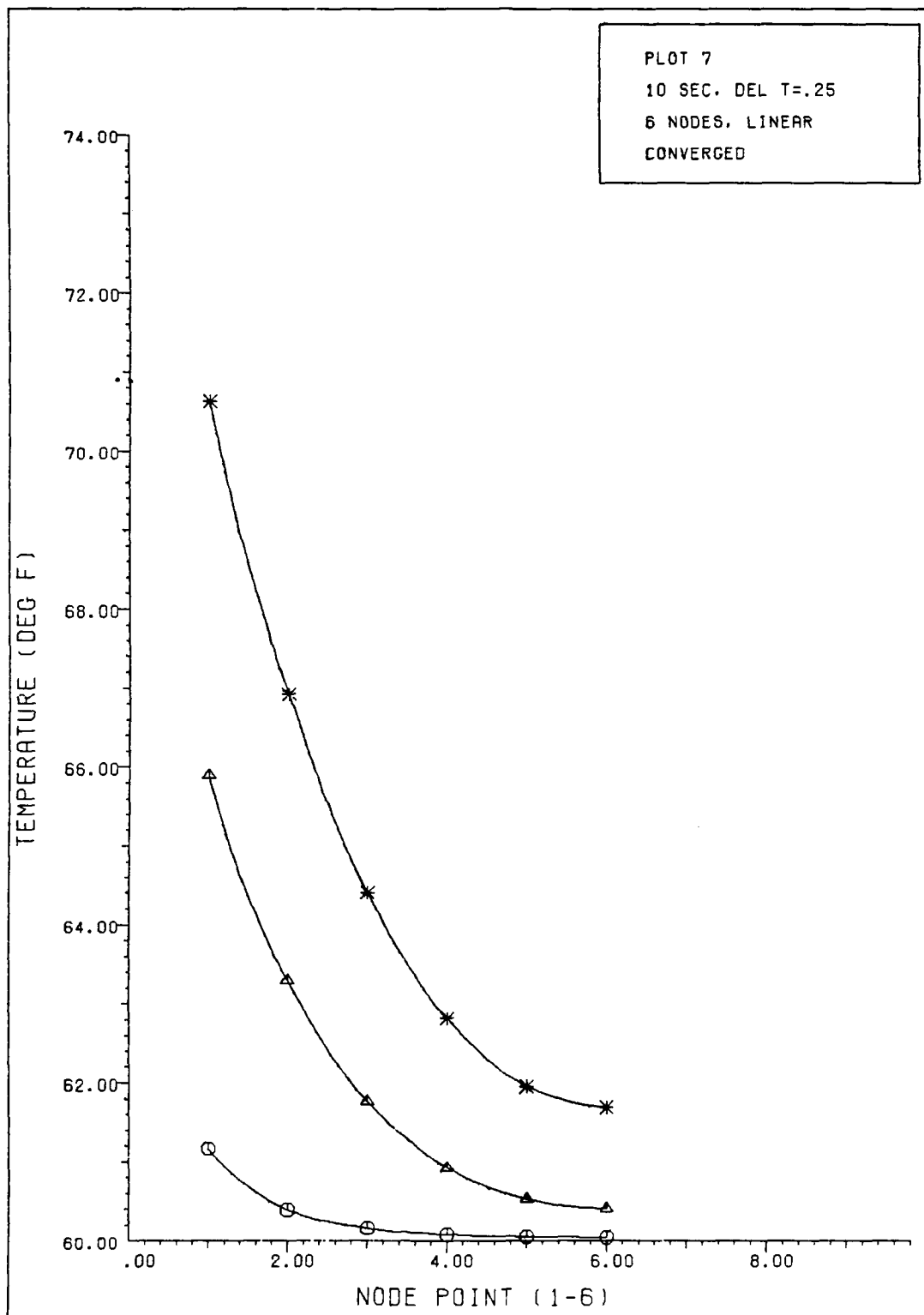
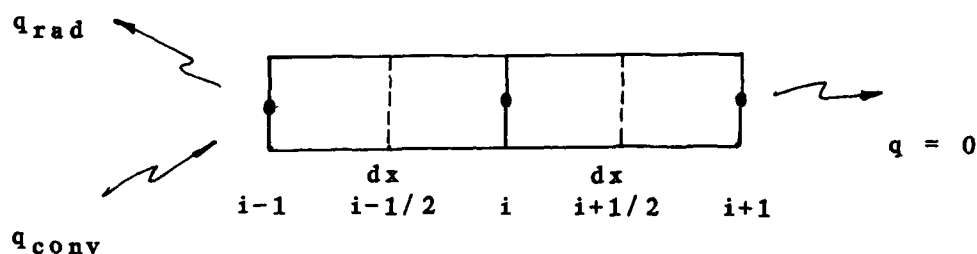


FIGURE 4.6 TEMP VS NODE POINT HISTORY (2.6,10 SEC)

APPENDIX A

Derivation of equations using a one-dimensional energy balance formulation are given as follows,



$$\text{Energy in the left face} = -k \partial T / \partial x = q_1$$

$$\text{Energy generated within the element} = q \, dx = 0$$

$$\text{Change in internal energy} = \rho c (\partial T / \partial \tau) dx$$

$$\begin{aligned} \text{Energy out of right face} &= -k (\partial T / \partial x)_{x+dx} \\ &= -\left[\frac{k \partial T}{\partial x} + \frac{\partial}{\partial x} \left(\frac{k \partial T}{\partial x} \right) dx \right] \end{aligned}$$

Then, combining the above and using Fourier's Law of Heat Conduction, ie,

energy in left face + energy within the element = change in internal energy + energy out right face
yields,

$$-\frac{k \partial T}{\partial x} + q dx = \rho c \frac{\partial T}{\partial \tau} dx - \left[\frac{k \partial T}{\partial x} + \frac{\partial}{\partial x} \left(\frac{k \partial T}{\partial x} \right) dx \right] \quad (\text{A-1})$$

or,

$$\rho c \frac{\partial T}{\partial \tau} = \frac{\partial}{\partial x} \left(\frac{k \partial T}{\partial x} \right) \quad (\text{A-2})$$

or, replacing T by U,

$$\rho c U_t = (k U_x)_x \quad (A-3)$$

or,

$$\rho c U_t = \left[\frac{k_{i-1/2}}{\Delta x_{i-1/2}} (U_{i-1}^n - U_i^n) - \frac{k_{i+1/2}}{\Delta x_{i+1/2}} (U_i^n - U_{i+1}^n) \right] \frac{1}{\Delta x} \quad (A-4)$$

where Δx may be written as,

$$\Delta x = \frac{\Delta x_{i+1/2} + \Delta x_{i-1/2}}{2}$$

then, writing the time gradient in first order backward difference form and expanding yields,

$$\begin{aligned} \rho c \frac{U_i^n - U_{i-1}^n}{\Delta t} &= \frac{2k_{i-1/2}}{\Delta x_{i-1/2}(\Delta x_{i+1/2} + \Delta x_{i-1/2})} U_{i-1}^n \\ &- \frac{2k_{i-1/2}}{\Delta x_{i-1/2}(\Delta x_{i+1/2} + \Delta x_{i-1/2})} U_i^n \\ &- \frac{2k_{i+1/2}}{\Delta x_{i+1/2}(\Delta x_{i+1/2} + \Delta x_{i-1/2})} U_i^n \\ &+ \frac{2k_{i+1/2}}{\Delta x_{i+1/2}(\Delta x_{i+1/2} + \Delta x_{i-1/2})} U_{i+1}^n \end{aligned} \quad (A-5)$$

then, specifying equal spacing for each node point,

$$\Delta x_{i+1/2} = \Delta x_{i-1/2}$$

and the equation becomes,

$$\begin{aligned} \rho c \frac{U_i^n - U_i^{n-1}}{\Delta t} &= \frac{k_{i-1/2}}{\Delta x^2} U_{i-1}^n \\ &- \left(\frac{k_{i-1/2}}{\Delta x^2} + \frac{k_{i+1/2}}{\Delta x^2} \right) U_i^n + \frac{k_{i+1/2}}{\Delta x^2} U_{i+1}^n \end{aligned} \quad (A-6)$$

Now, instead of estimating c and k directly, define two scaling parameters ϕ_c and ϕ_k such that c and k will remain constant. These two parameters are estimated by the HEATEST program.

$$\begin{aligned} \rho \phi_c c \Delta x \frac{U_i^n - U_i^{n-1}}{\Delta t} &= \phi_k \frac{k_{i-1/2}}{\Delta x} U_{i-1}^n \\ &- \phi_k \left(\frac{k_{i-1/2}}{\Delta x} + \frac{k_{i+1/2}}{\Delta x} \right) U_i^n + \phi_k \frac{k_{i+1/2}}{\Delta x} U_{i+1}^n \end{aligned} \quad (A-7)$$

This equation is applicable at all interior ($i \neq 1, i \neq i_{\max}$) points.

For the back face, assuming a semi-infinite solid, $i = i_{\max}$ and the equation becomes,

$$\frac{\rho \phi_c c \Delta x}{2} \frac{U_L^n - U_L^{n-1}}{\Delta t} = \frac{\phi_k k_{L-1/2}}{\Delta x} U_{L-1}^n - \frac{\phi_k k_{L-1/2}}{\Delta x} U_L^n \quad (A-8)$$

For the front face, ($i = 1$), the effects of radiation away from and convection toward the solid surface must be accounted for.

The radiation is modeled using the Stefan-Boltzmann Law,

$$q = \epsilon \sigma (U_1^4 - U_{\infty}^4) \quad (A-9)$$

where	ϵ	radiative emissivity
	σ	Stefan-Boltzmann constant
	U	Temperature ($^{\circ}R$)

The convective transfer of energy is modeled using Newton's Law of Cooling,

$$q = h(T_{aw} - T_w) \quad (A-10)$$

Non-dimensionalizing by a reference heat transfer coefficient, h_{ref} yields,

$$q = h_{bar} h_{ref} (T_{aw} - U_1) \quad (A-11)$$

where, $h_{bar} =$ convective heat transfer coefficient ratio
 $T_{aw} =$ adiabatic wall temp of test article

The dependance of the heat transfer coefficient on parameters other than those included in the reference heat transfer coefficient are summarized by the static transfer relation or heat transfer coefficient ratio, h/h_{ref} . Here, the ratio is assumed to be piecewise linear with respect to angle of attack as derived from Lagrange Interpolation Theory (Ref 6),

$$h_{\text{bar}} = h/h_{\text{ref}} = [h_0 + h_{a1}(\alpha - \alpha_1) + h_{a2}(\alpha - \alpha_2)] \quad (\text{A-12})$$

where h_0 is the magnitude of the heat transfer coefficient, h , at the reference condition specified at α_1 . Combining Equations A-7, A-9, A-11, and A-12 and evaluating at node one yields,

$$\begin{aligned} \frac{\rho \phi_c \Delta x}{2} \frac{U_1^n - U_1^{n-1}}{\Delta t} = & - \frac{\phi_k k_{k+1/2}}{\Delta x} U_1^n \\ & + \frac{\phi_k k_{1+1/2}}{\Delta x} U_2^n - \epsilon \sigma [(U_1^n)^4 - (U_{\text{amb}}^n)^4] \\ & + [h_0 + h_{a1}(\alpha - \alpha_1) + h_{a2}(\alpha - \alpha_2)] h_{\text{ref}} (T_{\text{aw}} - U_1^n) \end{aligned} \quad (\text{A-13})$$

Using the quasi-linearization as developed in Equation 2-6, the resultant form for determining the temperature time history at each node point is given in Equations 2-7 and 2-8.

The matrix form for the equations may be found after defining the following,

$$\begin{aligned} \text{RCX}_i &= \rho \phi_c \Delta x & \text{RCX}_1 &= \frac{\text{RCX}_1}{2} & \text{RCX}_L &= \frac{\text{RCX}_L}{2} \\ \text{RM}_i &= \frac{\phi_k k_{i-1/2}}{\Delta x} & \text{RM}_1 &= 0 \\ \text{RP}_i &= \frac{\phi_k k_{i+1/2}}{\Delta x} & \text{RP}_L &= 0 \end{aligned}$$

$$BBB_1 = \frac{RCX_1}{\Delta t} + RM_1 + RP_1 + 4\epsilon\sigma(U_1^{n,s})^3 + h_{bar}h_{ref}$$

$$BBB_i = \frac{RCX_i}{\Delta t} + RM_i + RP_i$$

(A-14)

Then, using Equations A-14 in Equations 2-7 and 2-8 yields the matrix form of Equation 2-9,

$$[A]\{U_i^n\} + \{b\} = 0 \quad (2-9)$$

where,

$$[A] = \begin{bmatrix} -1 & RP_i/BBB_i & 0 \\ RM_i/BBB_i & & \\ & RP_i/BBB_i & \\ 0 & RM_i/BBB_i & -1 \end{bmatrix}$$

(A-15)

and,

$$\{b\} = \left\{ \begin{array}{l} \frac{\epsilon\sigma[(3U_1^{n,s})^4 + U_{\omega}^4] + h_{bar}h_{ref}T_{aw} + RCX_1U_1^{n-1}}{BBB_1} \\ \frac{RCX_iU_i^{n-1}}{\Delta t} / BBB_i \\ \frac{RCX_LU_L^{n-1}}{\Delta t} / BBB_L \end{array} \right\}$$

(A-16)

APPENDIX B

The derivation of the sensitivity equations. The derivative of Equation 2-7 with respect to each parameter yields equations from which the HEATEST program propagates the sensitivity. A vector of parameters is formed and the sensitivity notation is as shown,

$$\Theta = [h_0, h_{a1}, h_{a2}, \rho_c, \phi_k]^T \quad S_{i,k} = \frac{\partial U}{\partial \Theta_k}$$

$\underbrace{\hspace{1cm}}$ parameter no
 $\underbrace{\hspace{1cm}}$ node point

Defining, $h_{bar} = [h_0 + h_{a1}(a-a_1) + (h_{a2}(a-a_2))]$, the sensitivity equations at node one are,

$$\begin{aligned} \Theta_1: \quad & \frac{\rho \phi_c c \Delta x}{2} \frac{S_{1,1}^n - S_{1,1}^{n-1}}{\Delta t} = \frac{-\phi_k(k_{i+1/2})}{\Delta x} S_{1,1}^n \\ & + \frac{\phi_k(k_{i+1/2})}{\Delta x} S_{2,1}^n - 4\epsilon\sigma(U_1^n)^3 S_{1,1}^n + h_{ref}(T_{aw} - U_1^n) \\ & - S_{1,1}^n h_{ref} h_{bar} \end{aligned} \quad (B-1)$$

$$\begin{aligned} \Theta_2: \quad & \frac{\rho \phi_c c \Delta x}{2} \frac{S_{1,2}^n - S_{1,2}^{n-1}}{\Delta t} = \frac{-\phi_k(k_{i+1/2})}{\Delta x} S_{1,2}^n \\ & + \frac{\phi_k(k_{i+1/2})}{\Delta x} S_{2,2}^n - 4\epsilon\sigma(U_1^n)^3 S_{1,2}^n \\ & + (a-a_1) h_{ref}(T_{aw} - U_1^n) - S_{1,2}^n h_{ref} h_{bar} \end{aligned} \quad (B-2)$$

$$\begin{aligned}
\theta_3: \quad & \frac{\rho \phi_c c \Delta x}{2} \frac{S_{1,3}^n - S_{1,3}^{n-1}}{\Delta t} = \frac{-\phi_k(k_{i+1/2})}{\Delta x} S_{1,3}^n \\
& + \frac{\phi_k(k_{i+1/2})}{\Delta x} S_{2,3}^n - 4\varepsilon\sigma(U_1^n)^3 S_{1,3}^n \\
& + (\alpha - \alpha_2) h_{ref}(T_{aw} - U_1^n) - S_{1,3}^n h_{ref} h_{bar}
\end{aligned}$$

(B-3)

$$\begin{aligned}
\theta_4: \quad & \frac{S_{1,4}^n - S_{1,4}^{n-1}}{\Delta t} = \frac{-2\phi_k k_{i+1/2}}{\rho \phi_c c \Delta x^2} S_{1,4}^n + \frac{2\phi_k k_{i+1/2}}{\rho \phi_c^2 c \Delta x^2} U_1^n \\
& + \frac{2\phi_k k_{i+1/2}}{\rho \phi_c c \Delta x^2} S_{1,4}^n - \frac{2\phi_k k_{i+1/2}}{\rho \phi_c^2 c \Delta x^2} U_2^n \\
& - \frac{8\varepsilon\sigma(U_1^n)^3}{\rho \phi_c c \Delta x} S_{1,4}^n + \frac{2\varepsilon\sigma(U_1^4 - U_2^4)}{\rho \phi_c^2 c \Delta x} \\
& - \frac{2h_{bar} h_{ref}}{\rho \phi_c c \Delta x} S_{1,4}^n - \frac{2h_{bar} h_{ref}(T_{aw} - U_1)}{\rho \phi_c^2 c \Delta x}
\end{aligned}$$

(B-4)

$$\begin{aligned}
\theta_5: \quad & \frac{\rho \phi_c c \Delta x}{2} \frac{S_{1,5}^n - S_{1,5}^{n-1}}{\Delta t} = \frac{-\phi_k k_{i+1/2}}{\Delta x} S_{1,5}^n \\
& - \frac{k_{i+1/2}}{\Delta x} U_1^n + \frac{\phi_k k_{i+1/2}}{\Delta x} S_{2,5}^n + \frac{k_{i+1/2}}{\Delta x} U_2^n \\
& - 4\varepsilon\sigma U_1^3 S_{1,5} - S_{1,5} h_{ref} h_{bar}
\end{aligned}$$

(B-5)

The sensitivity equations at the interior node points are as

follows,

$$\theta_i, i = 1, 2, 3$$

$$\begin{aligned} \rho \phi_c c \Delta x \frac{S_{i,k}^n - S_{i,k}^{n-1}}{\Delta t} &= \frac{\phi_k k_{i-1/2}}{\Delta x} S_{i-1,k} \\ &- \frac{\phi_k (k_{i-1/2} + k_{i+1/2})}{\Delta x} S_{i,k} \\ &+ \frac{\phi_k k_{i+1/2}}{\Delta x} S_{i+1,k} \end{aligned}$$

(B-6)

$$\theta_4:$$

$$\begin{aligned} \frac{S_{i,4}^n - S_{i,4}^{n-1}}{\Delta t} &= \frac{\phi_k k_{i-1/2}}{\rho \phi_c c \Delta x^2} S_{i-1,4}^n - \frac{\phi_k k_{i-1/2}}{\rho \phi_c^2 c \Delta x^2} U_{i-1}^n \\ &- \frac{\phi_k}{\rho \phi_c c \Delta x} \frac{(k_{i-1/2} + k_{i+1/2})}{\Delta x} S_{i,4}^n \\ &+ \frac{\phi_k}{\rho \phi_c^2 c \Delta x} \frac{(k_{i-1/2} + k_{i+1/2})}{\Delta x} U_i^n + \frac{\phi_k k_{i+1/2}}{\rho \phi_c c \Delta x^2} S_{i+1,4}^n \\ &- \frac{\phi_k k_{i+1/2}}{\rho \phi_c^2 c \Delta x^2} U_{i+1}^n \end{aligned}$$

(B-7)

$$\theta_5:$$

$$\begin{aligned} \rho \phi_c c \Delta x \frac{S_{i,5}^n - S_{i,5}^{n-1}}{\Delta t} &= \frac{\phi_k k_{i-1/2}}{\Delta x} S_{i-1,5} \\ &- \frac{\phi_k (k_{i-1/2} + k_{i+1/2})}{\Delta x} S_{i,5} \end{aligned}$$

$$\begin{aligned}
& + \frac{\phi_k k_{i+1/2}}{\Delta x} S_{i+1,5} + \frac{k_{i-1/2}}{\Delta x} U_{i-1} \\
& - \frac{(k_{i-1/2} + k_{i+1/2})}{\Delta x} U_i + \frac{k_{i+1/2}}{\Delta x} U_{i+1}
\end{aligned}$$

(B-8)

The backface equations are of the same form as the node 1 equations without the convection and radiation terms.

If Equations A-14 are used to reduce the equations to the form of Equation 2-9, the sensitivity equations become,

$$[A']\{S_{i,k}\} + \{d_k\} = 0 \quad (B-9)$$

where the $[A]$ matrix for the sensitivity equations is the same as the $[A]$ matrix for the temperature equations, A-15. The $\{d\}$ vectors for each parameter are listed as follows,

$$\begin{aligned}
\{d\}_1 &= \left\{ \begin{aligned} & \left(h_{ref}(T_{aw} - U_1) + \frac{RCX_1}{\Delta t} S_{1,1}^{n-1} \right) / BBB_1 \\ & \left(\frac{RCX_i}{\Delta t} S_{i,1}^{n-1} \right) / BBB_i \end{aligned} \right\} \\
\{d\}_2 &= \left\{ \begin{aligned} & \left((\alpha - \alpha_1) h_{ref}(T_{aw} - U_1) + \frac{RCX_1}{\Delta t} S_{1,2}^{n-1} \right) / BBB_1 \\ & \left(\frac{RCX_i}{\Delta t} S_{i,2}^{n-1} \right) / BBB_i \end{aligned} \right\}
\end{aligned}$$

$$\{d\}3 = \left\{ \begin{array}{l} \left((a-a_2) h_{ref} (T_{aw} - U_1) + \frac{RCX_1}{\Delta t} S_{1,3}^{n-1} \right) / BBB_1 \\ \left(\frac{RCX_i}{\Delta t} S_{i,3}^{n-1} \right) / BBB_i \end{array} \right\}$$

$$\{d\}4 = \left\{ \begin{array}{l} \left(\frac{-RP_1 (U_2 - U_1) + \varepsilon \sigma (U_1^4 - U_2^4) - h_{bar} h_{ref} (T_{aw} - U_1)}{\phi_c} + \frac{RCX_1}{\Delta t} S_{1,4}^{n-1} \right) / BBB_1 \\ \left(\frac{-RM_i (U_{i-1} - U_i) + RP_i (U_i - U_{i+1}) + RCX_i S_{i,4}}{\phi_c} \right) / BBB_i \end{array} \right\}$$

$$\{d\}5 = \left\{ \begin{array}{l} \left(\frac{RP_1 (U_2 - U_1) + RCX_1 S_{1,5}^{n-1}}{\phi_k} \right) / BBB_1 \\ \left(\frac{RM_i U_{i-1} - (RM_i + RP_i) U_i + RP_i U_{i+1} + RCX_i S_{i,5}^{n-1}}{\phi_k} \right) / BBB_i \end{array} \right\}$$

(B-10)

APPENDIX C

The HEATEST program follows.


```

56 DATA LABELX/SHLABEL/
57 DATA DTP/O/
58 DATA ERALOW/.5/
59 DATA KA,KAF,IPE/1,1,1/
60 DATA THDEP/.05/
61 DATA NPTSS/6/
62 DATA TRAD/O./
63
64 C READ INPUT DECK DATA
65 C
66 CALL INPUT
67 C
68 C NODE POINT STRUCTURE
69 C
70 DX(1)=THDEP/NPTSS
71 NODES(1)=1
72 NODES(2)=NPTSS
73 C
74 C ENTER OUTER/PARAMETER ESTIMATION ITERATION LOOP
75 C
76 C
77 READ(5,3990)ITRJSK,ITCSK,NPTPC,IIC
78 FORMAT(8(8X,I2))
79 READ(5,4001)INTERV,IFXFLG
80 FORMAT(6X,I4,9X,I1,10X,7L1)
81 DELS=(TSTOP-TSTART)/INTERV
82 TSTOP=TSTOP
83
84 4 TSTOP=TSTART+DELS
85 IF(IFXFLG)THEN
86 READ(5,4011,END=4034)NRPIER,IFX,FAUTO,TSTOP,KFOPT,NPTPC
87 FORMAT(4X,I1,5X,I8,I1,10X,7L1/8X,F10.5,5X,I1,8X,I2)
88 READ(5,4012,END=4034)FREAD,READ
89 FORMAT(3X,I3L1,13F8.4)
90 DO 4016 II=1,5
91 IF(FREAD(II))QP(II)=READ(II)
92 DO 4017 II=1,2
93 IF(FREAD(II))ALPH(II)=READ(II)
94 4013 FORMAT(3X,7L1,7F10.5)
95 GO TO 4033
96 IFXFLG=.FALSE.
97 TSTOP=TSTOPF
98 4033 CONTINUE
99 END IF
100 IF(TSTOP.GT.TSTOPF)TSTOP=TSTOPF
101 IFXSUM=0
102 DO 30 I=1,NPAR
103 IFXSUM=IFXSUM+IFX(I)
104 NRPI=NRPI+1
105 DO 198 ITPRAM=1,NRPI
106 KA=1
107 KAF=1
108 REWIND 3
109 REWIND 4
110 REWIND 10
111 REWIND 13
112 CALL ZERO(S,NPAR,1)

```

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FHEATEST2 49
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FHEATEST2 52
OCT30 1
UPSEP19 2
UPSEP19 3
FHEATEST2 54
FHEATEST2 55
FHEATEST2 58
FHEATEST2 57
FHEATEST2 58
FHEATEST2 59
FHEATEST2 60
UPSEP19 4
OCT24 1
OCT24 2
FHEATEST2 120
FHEATEST2 121
FHEATEST2 122
FHEATEST2 123
FHEATEST2 124
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FHEATEST2 126
FHEATEST2 127
FHEATEST2 128
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FHEATEST2 131
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FHEATEST2 134
FHEATEST2 135
UPSEP24 1
UPSEP24 2
FHEATEST2 138
UPSEP19 5
UPSEP19 6
FHEATEST2 141
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FHEATEST2 160

```

DO=-LONG/-OT,ARG=-COMMON/-FIXED,CS= USER/-FIXED,DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST,PL=5000
FTN5,I,ANSI=0,L=OUTS,LO=S/-A.

```

1  SUBROUTINE TPS3(DTT)
2  COMMON /CTCMT/NTCT
3  COMMON/COMTUN/T,TAW1,ALPHA,H,V,RHO,P,TEMP,C,TRAD,RHOG,
4  8TO,TSINK,XFT,DEL,PDEL
5  COMMON /CHEAT/Q,IS,OREF,TW,M1,RENS,HBAR,HREF
6  COMMON/COSP/NPTSS,USI(6,6),PHI(6,6),NPT,PC(6,6),RR,
7  AQD(6,6),QDT(6,6),QUE(6,6),A(6,6),RCX(6,6),RP(6,6),RM(6,6)
8  COMMON/CSENS/SUSI(6,5),UM1(6)
9  COMMON /ICTPS2/TINIT(1),ERALLOW,E
10 COMMON /CDX/DX(1)
11 REAL M1
12 DIMENSION AAAA(6),CCCC(6),DDDD(6),G(6),W(6)
13 EQUIVALENCE (QD(1,1),AAAA(1)),(QD(1,2),CCCC(1)),(QD(1,3),DDDD(1)),
14 8(QD(1,5),G(1)),(QDT(1,1),W(1))
15 DATA SIG/4.781E-13/
16 DATA MIT/2/
17
18 C SHIFT STORAGE
19 DO 460 I=1,NPTSS
20 460 UM1(I)=USI(I)
21
22 C FORM TRIANGULAR MATRIX
23 C
24 DO 511 I=2,NPTSS
25 511 CONTINUE
26 C TRIANGULAR SOLUTION
27 DO 540 M=1,MIT
28 C I=1
29 540 BB8=RCX(1)/DTT+RP(1)+RM(1)+4.*E*SIG*(USI(1)+480.)*3.+
30 8HBAR*HREF
31 AAAA(1)=RM(1)/BB8
32 CCCC(1)=RP(1)/BB8
33 DDDD(1)=(RCX(1)+UM1(1)/DTT+E*SIG*(3*USI(1)+4*TRAD**4)+HBAR*HREF*
34 8TAW1)/BB8
35 G(1)=DDDD(1)
36 W(1)=-CCCC(1)
37 DO 520 I=2,NPTSS
38 520 W(I)=-CCCC(I)/(1.+AAAA(I)*W(I-1))
39 G(I)=(DDDD(I)+AAAA(I)*G(I-1))/(1.+AAAA(I)*W(I-1))
40 UNEW =G(NPTSS)
41 UERM=ABS(UNEW-USI(NPTSS))
42 USI(NPTSS)=UNEW
43 DO 530 L=2,NPTSS
44 I=NPTSS-L+1
45 UNEW=G(I)-W(I)*USI(I+1)
46 UERM=ABS(UNEW-USI(I))
47 UERMX=AMAX1(UERM,UERM)
48 USI(I)=UNEW
49 CONTINUE
50 IF(UERMX.LT.ERALLOW.AND.M.GE.5)GO TO 550
51 CONTINUE
52
53 530 CONTINUE
54 540 CONTINUE
55

```

UPAUG1 11
UPAUG1 12
OCT10 3
COMTUN 3
UPAUG1 14
UPAUG16 6
UPOCT09 5
UPAUG16 7
UPAUG1 15
UPAUG15 3
FTPS3 16
UPOCT09 8
FTPS3 19
FTPS3 20
UPAUG1 18
UPAUG1 17
FTPS3 38
FTPS3 39
FTPS3 40
FTPS3 48
FTPS3 47
FTPS3 49
FTPS3 57
FTPS3 58
FTPS3 59
FTPS3 60
UPSEP12 1
FTPS3 72
FTPS3 73
FTPS3 74
FTPS3 79
UPAUG1 18
UPAUG1 19
UPAUG1 20
UPAUG1 21
UPAUG1 22
UPAUG1 23
UPAUG1 24
UPAUG1 25
FTPS3 114
FTPS3 115
FTPS3 116
FTPS3 118
FTPS3 119
FTPS3 120
FTPS3 121
FTPS3 122
FTPS3 123
FTPS3 124
FTPS3 125
FTPS3 126
FTPS3 127
FTPS3 128
FTPS3 129

```

1 56 RCX(I)=RHOG*PHIC*ZP*DX(I)*.5
1 57 RP(I)=PHIK*Z/DX(I)
1 58 RM(I)=O.
1 59 A(1,1)=-RP(I)/RCX(I)
1 60 A(1,2)=RP(I)/RCX(I)
1 61 ALIN1=(-(4.*E*SIG*(USI(1)+480.))*.3.+HBAR*HREF)/RCX(I)
1 62 END IF
1 63
1 64 511 CONTINUE
1 65 A(1,1)=A(1,1)+ALIN1
1 66
1 67 C
1 68 C SYSTEM MATRIX COMPLETED
1 69 C
1 70 RETURN
1 71 END
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1  SUBROUTINE QUEMAT
2  COMMON /CONST/ XP1(13)
3  LOGICAL IFICENT, IFPLOT, IFPRINT
4  COMMON /CFLAG/ IFICENT, IFPLOT, IFPRINT
5  COMMON /COMTUN/ I, TA1, ALPHA, H, V, RHO, P, TEMP, C, TRAD, RHOG,
6  & TO, TSINK, XFT, DEL, PDEL
7  COMMON /CTCWMT/ NTCT
8  COMMON /CDX/ DX(1)
9  COMMON /COSP/ NPTSS, USI(6), PHI(6,6), NPT, PC(6,6), RR,
10 & QD(6,6), QDT(6,6), QUE(6,6), A(6,6), RCX(6), RP(6), RM(6)
11 COMMON /ICTPS2/ TINIT(1), ERALOW, E
12 COMMON /CSENS/ SUSI(6,5), UM1(6)
13 COMMON /CKF/ K(6), S1(6), J1(6,6), TC(2), NODES(2), KFOPT
14 REAL K, J1
15 COMMON /CICSTAT/ TR, SDIC, SDME, SDMEA, SDBN
16 COMMON /CHEAT/ Q, TS, QREF, TW, M1, RENS, HBAR, HREF
17 REAL M1
18 DIMENSION R(6,6)
19 EQUIVALENCE (QD(1,1), R(1,1))
20
21 RR = SDMEA
22
23 C MATRIX OF SPACIAL CORRELATIONS, R
24 C SPACIAL CORRELATIONS IN RSI MUST ALLOW FOR VARIABLE NODE STRUCTURE
25 DO 200 I=1,NPTSS
26 SUMR=0.
27 DO 200 J=I,NPTSS
28 XP=SUMR/TR
29 XP=ABS(XP)
30 IF (I.EQ.1) XP1(J)=XP*TR
31 IF (XP.GT.100.) XP=100.
32 R(I,J)=EXP(-XP)
33 IF (ABS(RP(J)).LT.1.E-8) GO TO 200
34 SUMR=SUMR+RCX(J)/RP(J)
35 R(J,I)=R(I,J)
36 PRINT*, 'XP1= ', (XP1(I), I=1,NPTSS)
37
38 C MODEL ERROR MATRIX, QUE
39 C
40 RC=SQRT((ABS((RCX(1)+RCX(2))*0.5)))
41 QER=(HBAR+HREF*(TAW1-USI(1))/(RM(1)+RP(1))/2)**2
42 DO 221 I=1,NPTSS
43
44 C WHILE I.EQ.1 ADD BOUNDARY NOISE DUE TO HEAT TERMS
45 IF (I.EQ.1) THEN
46 QUE(I,I)=((SDBN+QER)**2+(SDME+QER)**2)/2
47 ELSE
48 QUE(I,I) = (SDME+QER)**2.
49 END IF
50 END WHILE
51
52 IF (I.EQ.NPTSS) GO TO 222
53 IP1 = I+1
54 DO 221 J = IP1,NPTSS
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56 C
57 C READ CALCOMP PLOT SPECIFICATIONS
58 C
59 READ(5,1000)IPLOT
60 READ(5,2000)TSCALE,TMIN,TAXL
61 READ(5,2000)YSCALE,YMIN,YAXL
62 READ(5,2000)ASCALE,AMIN,AAXL
63 1000 FORMAT(10X,12.7(3X,12))
64 C
65 C READ TIMES
66 C
67 C START-STOP TIMES / PRINT TIME STEP
68 READ(5,2000)TSTART,TSTOP,DTPTNT
69 C DATA FOR I.C. SMOOTHER
70 READ(5,100)SMIC,ISMTH
71 100 FORMAT(8X,11.8X,F10.5)
72 C # OF ITERATIONS/FIX=0 FIXES PARAMETER/AUTO FLAG FOR REFERENCE
73 READ(5,2010)NRPITER,IFX,FAUTO,KFOPT
74 C NEWTON-RAPHSON ACCELERATION PARAMETERS(O)ACC(2)
75 READ(5,2015)ACC
76 C HEATING MODEL INITIAL PARAMETERS
77 READ(5,2020)(QP(II),II=1,NPAR)
78 C INITIAL REFERENCE VALUES FOR HEATING MODEL
79 READ(5,2030)ALPH
80 2010 FORMAT(5X,12.3X,51.1,5X,7L1.8X,11)
81 2015 FORMAT(5X,8F5.2/5X,8F5.2)
82 2020 FORMAT(10X,8F8.4/28X,5F8.4)
83 2030 FORMAT(10X,7F8.4)
84 RETURN
85 END
FINPUT 129
FINPUT 130
FINPUT 131
FINPUT 132
FINPUT 133
FINPUT 134
FINPUT 135
FINPUT 140
FINPUT 142
FINPUT 143
FINPUT 144
FINPUT 145
FINPUT 146
CHUCK 34
UPSEP24 3
CHUCK 36
FINPUT 147
FINPUT 148
FINPUT 149
FINPUT 150
FINPUT 151
UPSEP24 4
FINPUT 153
UPSEP24 5
OCT10 8
FINPUT 156
FINPUT 157
FINPUT 158
FINPUT 159
FINPUT 160

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1  SUBROUTINE IC
2  LOGICAL IFICENT,IFPLOT,IFPRINT
3  COMMON /CFLAG/IFICENT,IFPLOT,IFPRINT
4  COMMON /CTCMT/NTCT
5  COMMON /CDX/DX(1)
6  COMMON /COSP/NPTSS,USI(6),PHI(6,6),NPT,PC(6,6),RR,
7  &QD(6,6),QDT(6,6),QUE(6,6),A(6,6),RCX(6),RP(6),RM(6)
8  COMMON /ICTPS2/TINIT(1),ERALOW,E
9  COMMON /CSENS/SUSI(6,5),UM1(6)
10 COMMON /CKF/K(6),S1(6),J1(6,6),TC(2),NODES(2),KFOPT
11 REAL K,J1
12 COMMON /CICSTAT/TR,SDIC,SDME,SDMEA,SDBN
13 COMMON /CHEAT/Q,TS,QREF,TW,M1,RENS,HBAR,HREF
14 REAL M1
15 DIMENSION R(6,6)
16 EQUIVALENCE (QD(1,1),R(1,1))
17
18 C INITIAL SENSITIVITIES
19 C
20 C RR = SDMEA
21 C
22 C MATRIX OF SPACIAL CORRELATIONS, R
23 C
24 C SPACIAL CORRELATIONS IN RSI MUST ALLOW FOR VARIABLE NODE STRUCTURE
25 DO 200 I=1,NPTSS
26 SUMR=0.
27 DO 200 J=I,NPTSS
28 XP=SUMR/TR
29 XP=ABS(XP)
30 IF(XP.GT.100.)XP=100.
31 R(I,J)=EXP(-XP)
32 IF(ABS(RP(J)).LT.1.E-8)GO TO 200
33 SUMR=SUMR+RCX(J)/RP(J)
34 200 R(J,I)=R(I,J)
35 C
36 C COVARIANCE MATRIX OF TEMP ICS, PC
37 C
38 DO 210 I = 1,NPTSS
39 PC(I,I) = (SDIC*USI(I))*2.
40 IF (I.EQ.NPTSS) GO TO 211
41 IP1 = I+1
42 DO 210 J = IP1,NPTSS
43 PC(I,J) = USI(I)*USI(J)*SDIC**2.*R(I,J)
44 210 PC(J,I) = PC(I,J)
45 211 CONTINUE
46 RETURN
47 END

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FIC 2
UPOCT09 21
UPOCT09 22
UPAUG16 31
UPAUG16 32
UPAUG16 6
UPOCT09 5
UPAUG16 33
UPAUG16 7
UPAUG16 8
FDKF 3
FIC 15
UPAUG16 34
FIC 17
UPOCT09 7
FIC 19
FIC 20
FIC 21
FIC 22
FIC 23
FIC 24
FIC 25
FIC 26
FIC 27
FIC 28
FIC 29
FIC 30
DCT30 8
FIC 32
FIC 33
FIC 34
FIC 35
FIC 36
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FIC 40
FIC 41
FIC 42
FIC 43
FIC 44
FIC 45
FIC 46
FIC 47
FIC 48
FIC 49
FIC 50

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56 C
57 C READ CALCOMP PLOT SPECIFICATIONS
58 C
59 READ(5,1000)IPL0T
60 READ(5,2000)TSCALE,TMIN,TAXL
61 READ(5,2000)YSCALE,YMIN,YAXL
62 READ(5,2000)ASCAL,AMIN,AAXL
63 1000 FORMAT(10X,12,7(3X,12))
64 C
65 C READ TIMES
66 C
67 C START-STOP TIMES / PRINT TIME STEP
68 READ(5,2000)TSTART,TSTOP,DTPEM
69 C DATA FOR I.C. SMOOTHER
70 READ(5,100)SMIC,TSMT
71 100 FORMAT(6X,11,8X,F10.5)
72 C # OF ITERATIONS/FIX=0 FIXES PARAMETER/AUTO FLAG FOR REFERENCE
73 READ(5,2010)NRPTER,IFX,FAUTO,KFOPT
74 C NEWTON-RAPHSON ACCELERATION PARAMETERS(O)ACC(2)
75 READ(5,2015)ACC
76 C HEATING MODEL INITIAL PARAMETERS
77 READ(5,2020)(QP(II),II=1,NPAR)
78 C INITIAL REFERENCE VALUES FOR HEATING MODEL
79 READ(5,2030)ALPH
80 2010 FORMAT(5X,12,3X,5I11,5X,7L11,8X,11)
81 2015 FORMAT(5X,8F5.2/5X,8F5.2)
82 2020 FORMAT(10X,8F8.4/26X,5F8.4)
83 2030 FORMAT(10X,7F8.4)
84 RETURN
85 END

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FINPUT 129
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FINPUT 146
CHUCK 34
UPSEP24 3
CHUCK 36
FINPUT 147
FINPUT 148
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FINPUT 150
FINPUT 151
UPSEP24 4
FINPUT 153
UPSEP24 5
OCT10 8
FINPUT 158
FINPUT 157
FINPUT 158
FINPUT 159
FINPUT 160

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DO=-LONG/-OT,ARG=-COMMON/-FIXED,CS= USER/-FIXED,DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST,PL=5000
FTN5,1,ANSI=0,L=OUTS,LO=S/-A.

```
1 SUBROUTINE INPUT
2 LOGICAL IFICENT,IFPLOT,IFPRINT
3 CHARACTER *30 VEH,FLTDT,TMANV,CTPT
4 COMMON /CFLAG/IFICENT,IFPLOT,IFPRINT
5 COMMON/CTCMNT/NTCT
6 COMMON/CTIME/TSTART,TSTOP,DTPEM,NRPITER,ITPRAM
7 COMMON/ICTPS2/TINIT(1),ERALLOW,E
8 COMMON/CDX/DX(1)
9 LOGICAL FAUTO
10 DIMENSION FAUTO(7)
11 DIMENSION QP(5)
12 COMMON/CPARAM/HO,HALF(2),PHIC,PHIK,ZP,Z,ALPH(2),KA,S(5),
13 &CIF(5,5),KAF,IFX(5),ACC(5),IFXSUM,NPAR,DALPH(2)
14 EQUIVALENCE (HO,QP(1))
15 COMMON/CKF/K(8),S1(8),J1(8,8),TC(2),NODES(2),KFOPT
16 REAL K,J1
17 COMMON /CSMTH/ UICSM(8),PICSM(8,8),UAP(8),PAP(8,6),
18 &SMIC,TSMTH,W(8,8)
19 LOGICAL SMIC
20 COMMON /CCOM/VEH,FLTDT,TMANV,CTPT
21 COMMON /CICSTAT/TR,SDIC,SDME,SDMEA,SDBN
22 COMMON /CFPLOT/IPLOT,TSCALE,TMIN,TAXL,YSCALE,YMIN,YAXL,ASCALE,AMIN
23 * ,AAXL
24 C READ VEHICLE/MANEUVER UNIQUES
25 C
26 C
27 READ(5,3000)VEH
28 3000 FORMAT(A30)
29 READ(5,3000)FLTDT
30 READ(5,3000)TMANV
31 READ(5,3000)CTPT
32 C
33 C
34 READ(5,1000)NTCT
35 C
36 IFICENT = T ICS FROM DISK / F CONSTANT ICS FROM CARDS
37 C IFPLOT = T CREATE CALCOMP / F NO PLOT FILE
38 C IFPRINT = T PRINT TIME SERIES / F NO TEMP TIME SERIES OUTPUT
39 C
40 READ(5,4000)IFICENT
41 4000 FORMAT(8(9X,L1))
42 READ(5,4000)IFPLOT
43 READ(5,4000)IFPRINT
44 2000 FORMAT(10X,F10.5,3(5X,F10.5)/10X,F10.5,3(5X,F10.5))
45 C
46 C READ IN INITIAL TEMPERATURES (USED IF IFICENT = F)
47 C
48 READ(5,2000)TINIT
49 C
50 C
51 C INITIAL COVARIANCE AND ERROR STATISTICS
52 C
53 READ(5,2001)TR,SDIC,SDME,SDMEA
54 READ(5,2001)SDBN
55 2001 FORMAT(10X,F10.5,3(5X,F10.5))
56 C
57 C
58 C
59 C
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PRINT *, 'T, USI=', TLEST, (USI(II), II=1, NPTSS)
REWIND 3
TSTART=TSTOP
GO TO 4
END IF
REWIND 3
WRITE(3) USI, PC, SUSI
PRINT *, 'T, USI=', TLEST, (USI(II), II=1, NPTSS)
998 IF (IFPLOT) CALL FPLOT
C
C PLOT CALCOMP FILE FROM TAPE12
C
STOP
999 STOP 'END-OF-FILE ENCOUNTERED ON INPUT TAPE IN HEATEST'
END

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FHEATEST2 450
FHEATEST2 451
FHEATEST2 452
FHEATEST2 453
FHEATEST2 454
FHEATEST2 455
FHEATEST2 456
FHEATEST2 457
CHUCK 32
FHEATEST2 459
FHEATEST2 460
FHEATEST2 461
FHEATEST2 462
FHEATEST2 463
FHEATEST2 464


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1 284 IF(NLAB.EQ.O)GO TO 120
1 285 24 CONTINUE
1 286 GO TO 100
1 287 END IF
1 288
1 289 C C PROPAGATION TO TSTOP
1 290 C
1 291 120 DELT=TSTOP-TLEST
1 292 CALL TPS3(DELT)
1 293 TLEST=TSTOP
1 294 C WRITE SMOOTHED I.CS TO TAPE3
1 295 IF(SMIC)THEN
1 296 REWIND (3)
1 297 WRITE(3) UICSM,PICSM,SUSI
1 298 REWIND (3)
1 299 IIC=1
1 300 ENDIF
1 301
1 302 C C EXIT TEMPERATURE/STATE ESTIMATION LOOP
1 303 C
1 304 C IF(ITPRAM.LE.NRPITER.OR.NRPITER.EQ.O) THEN
1 305 C
1 306 C UPDATE PARAMETER ESTIMATES - LIST RESULTS
1 307 C
1 308 CALL PAREST
1 309 IT=1
1 310 DO 197 II=1,NPAR
1 311 IF(IFX(II).EQ.O)THEN
1 312 CI(II)=0.
1 313 GO TO 197
1 314 END IF
1 315 CI(II)=SQRT(ABS(CIF(IT,II)))
1 316 IT=II+1
1 317
1 318 197 CONTINUE
1 319 IF(ITPRAM.EQ.1)WRITE(6,3079)
1 320 &T54,'PHIK',T66,'ZP',T76,'Z',T88,'ALPHA1',T100,
1 321 &'ALPHA2',T112,'ALPHA3',T68,'QBETA',T18,'QLOGRE',T30,'QDELE',T42
1 322 &,'QDELEBF',T54,'QMACH',T66,'PHIKB',T1X,'(CRAMER-RAO BOUND)')
1 323 WRITE(6,3080)ITPRAM,(CI(I),I=1,10),(CI(I),I=1,10),
1 324 &(QP(I),I=1,16),(CI(I),I=11,16),
1 325 3080 FORMAT(1X,I2,2X,10(F8.5,4X)/5X,10(1X,(' ',E8.2,')'),1X)/
1 326 &5X,6(F8.5,4X)/5X,6(1X,(' ',E8.2,')'),1X)
1 327 WRITE(6,3081)AVERAGE
1 328 3081 FORMAT(1X,'AVERAGE ERROR=',E12.5/)
1 329 C
1 330 END IF
1 331 198 CONTINUE
1 332 IIC=1
1 333
1 334 C C
1 335 C IF(SMIC) GO TO 998
1 336 IF(TSTOP.LT.ISTOPF-1.E-6)THEN
1 337 C RESET INITIAL CONDITIONS WITH FOLLOWING DATA TO DISK
1 338 C
1 339 REWIND 3
1 340 WRITE(3)USI,PC,SUSI

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FHEATEST2 342
FHEATEST2 343
FHEATEST2 344
FHEATEST2 345
FHEATEST2 346
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FHEATEST2 350
FHEATEST2 351
CHUCK 23
CHUCK 24
CHUCK 27
CHUCK 28
CHUCK 29
HAROLD 26
CHUCK 30
FHEATEST2 352
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FHEATEST2 371
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OCT30 7
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FHEATEST2 397
FHEATEST2 444
CHUCK 1
FHEATEST2 445
FHEATEST2 446
FHEATEST2 447
FHEATEST2 448
FHEATEST2 449

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2 227 C SAVE A PRIORI VALUES
1 228 DO 27 I=1,NPTSS
1 229 UAP(I)=USI(I)
1 230 DO 27 J=1,NPTSS
1 231 PAP(I,J)=PC(I,J)
1 232 C
1 233 DO 25 I=1,NTCT
1 234 PU(I)=USI(NODES(I))
1 235 PE(I)=SQRT(ABS(PC(NODES(I),NODES(I))))
1 236 CALL KF
1 237 E1 = SQRT(ABS(PC(1,1)))
1 238 DO 26 I=1,NTCT
1 239 PUF(I)=USI(NODES(I))
1 240 PEF(I)=SQRT(ABS(PC(NODES(I),NODES(I))))
1 241 WRITE(9,99)I,PU(1),PE(1),PUF(1),PEF(1),TC(1)
1 242 FORMAT(1X,F5.2,3X,2(3X,F9.5,3X,'(','E8.2,')'),5X,F9.5)
1 243 WRITE(9,98)(USI(I),I=1,NPTSS)
1 244 FORMAT(11,8(F8.5,3X))
1 245 C INITIALIZE SMOOTHER
1 246 IF(ICOUNT.EQ.0)THEN
2 247 DO 1000 I=1,NPTSS
2 248 UICSM(I)=USI(I)
2 249 DO 1000 J=1,NPTSS
2 250 PICS(I,J)=PC(I,J)
2 251 W(I,J)=PC(I,J)
2 252 ENDIF
1 253 TSMTH=TLEST
1 254 C SMOOTH I.C.'S
1 255 IF(SMIC)THEN
2 256 IF(ICOUNT.NE.0)THEN
3 257 CALL FPSM(TSTART)
3 258 DO 510 IM=1,NPTSS
3 259 VAR(IM)=SQRT(ABS(PICSM(IM,IM)))
3 260 ENDIF
2 261 ENDIF
2 262 C WRITE APOSTERIORI STATE ESTIMATE TO TAPE10/TAPE12
2 263 C
1 264 IF (ITPRAM.GT.NRPITER) THEN
2 265 IF (IFPRINT.WRITE(10)1,TLEST,USI(1),E1,EQUI,PU,PE,PUF,TC,
2 266 & Q,QREF,ALPHA,BETA,RENS,DELE,DELBF,M1
2 267 & M1
2 268 IF(1FPLT)WRITE(12,3095)TLEST,USI(1),EQUI,PU(1),TC(1),
2 269 & ALPHA,BETA,RENS,DELE,DELBF,M1,QN
2 270 TRITE=TLEST
2 271 END IF
2 272 DTP = 0.
1 273 C
1 274 ICOUNT= 1
1 275 C KF UPDATE COMPLETE - SET UP FOR NEXT PROPAGATION INTERVAL
1 276 C
1 277 C UPDATE THERMAL PROPERTIES/A MATRIX BASED ON UPDATED STATES
1 278 C NOTE! PROPERTIES MAY NEED UPDATED MORE OFTEN IF TC SAMPLE RATE IS LOW
1 279 CALL MAKEA
1 280 READ NEXT THERMOCOUPLE SAMPLE
1 281 DO 24 II=1,ITCSK
1 282 READ(13,END=999)NLAB,TUPDT,(TC(I),I=1,NTCT)
1 283

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CHUCK 11
CHUCK 12
CHUCK 13
CHUCK 14
CHUCK 15
CHUCK 16
FHEATEST2 312
FHEATEST2 313
FHEATEST2 314
FHEATEST2 315
FHEATEST2 316
FHEATEST2 317
FHEATEST2 318
FHEATEST2 319
OCT30 2
OCT30 3
OCT30 4
OCT30 5
HAROLD 10
HAROLD 11
HAROLD 12
HAROLD 13
HAROLD 14
HAROLD 15
HAROLD 16
HAROLD 17
HAROLD 18
CHUCK 17
CHUCK 18
HAROLD 19
HAROLD 20
HAROLD 21
HAROLD 22
CHUCK 21
CHUCK 22
FHEATEST2 320
FHEATEST2 321
FHEATEST2 322
FHEATEST2 323
FHEATEST2 324
UPCOTO9 16
FHEATEST2 325
FHEATEST2 326
FHEATEST2 327
FHEATEST2 328
FHEATEST2 329
FHEATEST2 330
FHEATEST2 331
FHEATEST2 332
FHEATEST2 333
HAROLD 25
FHEATEST2 334
FHEATEST2 335
FHEATEST2 336
FHEATEST2 337
FHEATEST2 338
FHEATEST2 339
FHEATEST2 340
UPAUG24 1

```

170 DO 500 IM=1,NPTSS
171   UICSM(IM)=USI(IM)
172   VAR(IM)=SORT(ABS(PC(IM,IM)))
173   PRINT*, 'UIC= ',(UICSM(IM),IM=1,NPTSS)
174   PRINT*, 'VAR= ',(VAR(IM),IM=1,NPTSS)
175   ICOUNT=0
176
177 C PROPAGATION TO TRAJECTORY SAMPLE TIME/TIMES
178
179 C
180   100 IF(T.LT.TUPDT.AND.T.LT.TSTOP)THEN
181     DELT=T-TLEST
182     CALL TPS3(DELT)
183     CALL SENS(DELT)
184     CALL TPSOSP2(DELT)
185     TLEST=T
186     DTP=DTP+DELT
187
188 C
189 C WHEN DTP.GE.DTPENT THEN WRITE TEMP/STATE ESTIMATES TO TAPE10/TAPE12
190 C
191 C
192   IF ((DTP.GE.DTPENT).AND.(ITPRAM.GT.NRPITER)) THEN
193     DTP = 0.
194     IF (IFPRINT)WRITE(6)0,TLEST,USI(1),TC(1),HBAR,HREF,ALPHA,TO
195     IF (IFPRINT)WRITE(10)0,TLEST,USI(1),EQUI,USI(NODES(1)),Q,
196     &QREF,ALPHA,BETA,RENS,DELE,DELBF,M1
197     IF (IFPLOT)WRITE(12,3055)TLEST,USI(1),EQUI,USI(NODES(1)),TC(1),
198     &ALPHA,BETA,RENS,DELE,DELBF,M1,QN
199     3055 FORMAT(8E13.7)
200     TRITE=TLEST
201   END IF
202
203 C
204 C READ NEXT TRAJECTORY SAMPLE
205 DO 22 II=1,ITRJSK
206   READ(4,END=999)NLAB,T,(FLT1(I),I=1,NLAB)
207   IF (NLAB.EQ.0)T=TSTOP
208 22 CONTINUE
209 CALL HEATTUN
210 GO TO 100
211 END IF
212
213 C PROPAGATION TO THERMOCOUPLE SAMPLE TIME/TIMES
214 C
215 IF(TUPDT.LE.TSTOP)THEN
216   DELT=TUPDT-TLEST
217   CALL TPS3(DELT)
218   CALL SENS(DELT)
219   CALL TPSOSP2(DELT)
220   TLEST=TUPDT
221   DTP=DTP+DELT
222   IF(TUPDT.GE.T)THEN
223     DO 23 II=1,ITRJSK
224       READ(4,END=999)NLAB,T,(FLT1(I),I=1,NLAB)
225       IF (NLAB.EQ.0)T=TSTOP
226 23 CONTINUE
227 CALL HEATTUN
228 END IF
229
230 C KALMAN UPDATES
231 C
232 C
233 C
234 C
235 C
236 C

```

HAROLD 4
 HAROLD 5
 HAROLD 6
 HAROLD 7
 HAROLD 8
 HAROLD 9
 FHEATEST2 262
 FHEATEST2 263
 FHEATEST2 264
 FHEATEST2 265
 FHEATEST2 266
 FHEATEST2 267
 FHEATEST2 268
 FHEATEST2 269
 FHEATEST2 270
 FHEATEST2 271
 FHEATEST2 272
 FHEATEST2 273
 FHEATEST2 274
 FHEATEST2 275
 UPOCT09 13
 UPOCT09 14
 FHEATEST2 277
 FHEATEST2 278
 FHEATEST2 280
 FHEATEST2 281
 FHEATEST2 282
 FHEATEST2 283
 FHEATEST2 284
 FHEATEST2 285
 FHEATEST2 286
 FHEATEST2 287
 FHEATEST2 288
 FHEATEST2 289
 UPUG16 24
 FHEATEST2 291
 FHEATEST2 292
 FHEATEST2 293
 FHEATEST2 294
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 FHEATEST2 298
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 FHEATEST2 300
 FHEATEST2 301
 FHEATEST2 302
 FHEATEST2 303
 FHEATEST2 304
 FHEATEST2 305
 FHEATEST2 306
 UPUG16 25
 FHEATEST2 308
 FHEATEST2 309
 FHEATEST2 310
 FHEATEST2 311
 CHUCK 10

```

113 CALL ZERO(CIF,NPAR,NPAR)
114 IF(IIC.EQ.O)CALL ZERO(SUSI,NPTSS,NPAR)
115
116 C READ C & B FILE LABEL ON TRAJ TAPE
117 C
118 READ(4)C,LIC,ITAIL,TITLE,ITEST,FLT,DFLT,DREQ,DCOM
119 33 IF(EOF(4).NE.O.)CALL EXIT
120 IF(C.NE.CLAB)CALL EXIT
121 READ(4) LABEL,NRSECT,NREM,(REMARK(I),I=1,NREM),NLAB,
122 *(PLAB1(I),PLAB2(I),NUM(I),I=1,NLAB)
123 IF(LABEL.NE.LABELX)CALL EXIT
124
125 C READ C & B FILE LABEL ON THE T/C MEAS TAPE
126 C
127 7 READ(13)C,LIC,ITAIL,TITLE,ITEST,FLT,DFLT,DREQ,DCOM
128 44 IF(EOF(13).NE.O.)CALL EXIT
129 IF(C.NE.CLAB)CALL EXIT
130 READ(13) LABEL,NRSECT,NREM,(REMARK(I),I=1,NREM),NLAB,
131 *(PLAB1(I),PLAB2(I),NUM(I),I=1,NLAB)
132 IF(LABEL.NE.LABELX)CALL EXIT
133
134 C SET TEMPERATURE INITIAL CONDITIONS
135 C
136 IF(IIC.EQ.O)THEN
137 C INITIAL TEMPERATURES
138 C
139 DO 403 I=1,NPTSS
140 403 USI(I)=TINIT(1)
141 IF(IFICENT)THEN
142 READ(3)USI
143 END IF
144 END IF
145 IF(IIC.NE.O)THEN
146 REWIND 3
147 READ(3)USI,PC,SUSI
148 REWIND 3
149 END IF
150 EQUI=USI(1)
151 C INITIALIZE SMOOTHER
152 C INITIALIZE PARAMETERS AT TSTART
153 DTP=0.
154 TTEST=TSTART
155 C READ FIRST TRAJECTORY SAMPLE
156 10 READ(4,END=999)NLAB,T,(FLT1(I),I=1,NLAB)
157 IF(NLAB.EQ.O)GO TO 999
158 IF(T.LT.TSTART)GO TO 10
159 C CALCULATE REFERENCE HEATING
160 CALL HEATTUN
161 C READ FIRST THERMOCOUPLE SAMPLES AND LOCAL PRESSURE
162 20 READ(13,END=999)NLAB,TUPDT,(TC(I),I=1,NTCT)
163 IF(NLAB.EQ.O)GO TO 999
164 IF(TUPDT.LT.TSTART)GO TO 20
165 C INITIALIZE THERMAL PROPERTIES/A MATRIX
166 CALL MAKEA
167 IF(IIC.EQ.O)CALL IC
168 CALL QUEMAT
169 C

```

FHEATEST2 161
FHEATEST2 162
FHEATEST2 163
FHEATEST2 164
FHEATEST2 165
FHEATEST2 166
FHEATEST2 167
FHEATEST2 168
FHEATEST2 169
FHEATEST2 170
FHEATEST2 171
FHEATEST2 172
FHEATEST2 173
FHEATEST2 174
FHEATEST2 175
FHEATEST2 176
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FHEATEST2 181
FHEATEST2 182
FHEATEST2 183
FHEATEST2 184
FHEATEST2 185
FHEATEST2 186
FHEATEST2 187
UPAUG18 21
FHEATEST2 188
FHEATEST2 190
FHEATEST2 191
FHEATEST2 192
FHEATEST2 193
FHEATEST2 194
FHEATEST2 195
FHEATEST2 196
FHEATEST2 197
FHEATEST2 198
CHUCK 2
FHEATEST2 199
FHEATEST2 200
FHEATEST2 201
FHEATEST2 202
FHEATEST2 203
FHEATEST2 204
FHEATEST2 205
FHEATEST2 206
UPAUG18 22
FHEATEST2 208
UPAUG18 23
FHEATEST2 210
FHEATEST2 211
FHEATEST2 212
FHEATEST2 213
HAROLD 3
FHEATEST2 215
FHEATEST2 261

56	IF(UERMX.GT.ERALDW)WRITE(6,1000)ERALDW,N.T,UERMX,USI(1)		
57	1000 FORMAT(1X,15HMAX ERROR TEMP~,E12.6,2X,12,2X,4(E12.6,1X))		
58	550 CONTINUE		
59	570 RETURN		
60	END		
		FTPS3	130
		FTPS3	131
		FTPS3	132
		FTPS3	133
		FTPS3	134

```

1  SUBROUTINE SENS(DTT)
2  COMMON/COMTUN/T,TAW1,ALPHA,H,V,RHO,P,TEMP,C,TRAD,RHOG,
3  &TO,TSINK,XFT,DEL,PDEL
4  COMMON /CHEAT/Q,TS,QREF,TW,M1,RENS,HBAR,HREF
5  COMMON/COSP/NPTS,PHI(6,6),NPT,PC(6,6),RR,
6  &QD(6,6),QDT(6,6),QUE(6,6),A(6,6),RCX(6),RP(6),RM(6)
7  COMMON /ICTPS2/TINIT(1),ERALLOW,E
8  COMMON /CDX/DX(1)
9  COMMON/CSNS/SUSI(8,5),UM1(8)
10 LOGICAL FAUTO
11 DIMENSION FAUTO(7)
12 DIMENSION QP(5)
13 COMMON/CPARAM/HO,HALF(2),PHIC,PHIK,ZP,Z,ALPH(2),KA,S(5),
14 &CIF(5,5),KAF,IFX(5),ACC(5),IFXSUM,NPAR,DALPH(2)
15 EQUIVALENCE (HO,QP(1))
16 REAL M1
17 DIMENSION AA(8),BB(8),CC(8),DD(8),AAA(8),CCC(8),DDD(8),W(6),G(6)
18 EQUIVALENCE (QD(1,1),AA(1)),(QD(1,2),BB(1)),(QD(1,3),CC(1)),
19 &(QD(1,4),DD(1)),(QD(1,5),AAA(1))
20 EQUIVALENCE (QDT(1,1),CCC(1)),(QDT(1,2),DDD(1)),(QDT(1,3),W(1)),
21 &(QDT(1,4),G(1))
22 DATA SIG/4.781E-13/
23 DATA E/.3/
24 DATA NPTS/40/
25
26 C IF (DTT.EQ.0.0) GO TO 999
27
28 C BACKWARD-DIFFERENCE FORMULATION OF DIFF. EQS.
29
30 C I=1
31 RCX(1)=RHOG*PHIC*ZP*DX(1)*.5
32 RP(1)=PHIK*Z/DX(1)
33 RM(1)=0.
34 C SET UP TRIANGULAR MATRIX (COMMON TERMS ONLY)
35 DO 520 I=1,NPTS
36 BB(I)=RCX(I)/DTT+RM(I)+RP(I)
37 AA(I)=RM(I)
38 CC(I)=RP(I)
39 DD(I)=RCX(I)/DTT
40 CONTINUE
41
42 520 BB(1)=BB(1)+4.*E*SIG*(USI(1)+460.)*+3* HBAR *HREF
43
44 C I=1 SENSITIVITY FOR EACH PARAMETER
45 DO 530 IP=1,NPAR
46 IF (IFX(IP).EQ.0) GO TO 530
47 DO 531 I=1,NPTS
48 DDD(I)=DD(I)*SUSI(I,IP)
49
50 C SENSITIVITY FOR UNIT SURFACE CONDUCTANCE NAUT (THETA ONE)
51 IF (IP.EQ.1) DDD(1)=DDD(1)+HREF*(TAW1-USI(1))
52 C SENSITIVITIES FOR EACH HEATING MODEL PARAMETER
53 IF (IP.EQ.2) DDD(1)=DDD(1) +HREF*(TAW1-USI(1))*(ALPHA-ALPH(1))
54 IF (IP.EQ.3) DDD(1)=DDD(1) +HREF*(TAW1-USI(1))*(ALPHA-ALPH(2))
55 C SENSITIVITY FOR SPECIFIC HEAT FACTOR PHIC

```

UPAUG1

26

OCT10

3

COMTUN

3

UPAUG16

45

UPAUG16

6

UPOCT09

5

UPAUG16

46

UPAUG15

4

UPAUG16

7

CPARAM

2

OCT10

2

UPAUG16

1

UPAUG16

2

UPAUG16

3

UPAUG16

4

FSENS3

18

UPOCT09

9

FSENS3

18

FSENS3

19

FSENS3

20

FSENS3

21

UPAUG16

47

UPAUG16

48

FSENS3

23

FSENS3

25

UPAUG1

28

FSENS3

28

FSENS3

29

FSENS3

30

FSENS3

35

FSENS3

36

UPAUG1

29

UPAUG1

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FSENS3

43

FSENS3

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FSENS3

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FSENS3

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FSENS3

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FSENS3

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FSENS3

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FSENS3

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UPAUG1

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FSENS3

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FSENS3

67

FSENS3

68

FSENS3

69

FSENS3

70

FSENS3

71

FSENS3

72

UPAUG1

32

UPAUG1

33

FSENS3

82

UPAUG1

34

UPAUG1

35

UPAUG1

36

```

58      IF(IP.EQ.4) THEN
59      DDD(1)=DDD(1)-RP(1)*(USI(2)-USI(1))-(HBAR*HREF*(TAW1-
60      A*USI(1))+E*SIG*((USI(1)+460.)*4-TRAD**4))/PHIC
61      DO 533 I=2, NPTSS-1
62      DDD(I)=DDD(I)-(RM(I)*(USI(I-1)-USI(I))+RP(I)*(USI(I)-USI(
63      &I+1)))/PHIC
64      DDD(NPTSS)=DDD(NPTSS)+RM(NPTSS)*(USI(NPTSS-1)-USI(NPTSS))/PHIC
65      END IF
66      C SENSITIVITY FOR CONDUCTIVITY FACTOR PHIK
67      IF(IP.EQ.5) THEN
68      DDD(1)=DDD(1)+RP(1)*(USI(2)-USI(1))/PHIK
69      DO 534 I=2, NPTSS-1
70      DDD(I)=DDD(I)+(RM(I)*USI(I-1)-(RM(I)+RP(I))*USI(I)+RP(I)
71      &+USI(I+1))/PHIK
72      DDD(NPTSS)=DDD(NPTSS)+RM(NPTSS)*(USI(NPTSS-1)-USI(NPTSS))/PHIK
73      END IF
74      DO 535 I=1, NPTSS
75      AAA(I)=AA(I)/BB(I)
76      CCC(I)=CC(I)/BB(I)
77      DDD(I)=DDD(I)/BB(I)
78      C
79      C TRIAGONAL SOLUTION
80      G(1)=DDD(1)
81      W(1)=-CCC(1)
82      DO 536 I=2, NPTSS
83      W(I)=-CCC(I)/(1.+AAA(I)*W(I-1))
84      G(I)=(DDD(I)+AAA(I)*G(I-1))/(1.+AAA(I)*W(I-1))
85      C BACKWARD SUBSTITUTION
86      SUSI(NPTSS,IP)=G(NPTSS)
87      DO 537 L=2, NPTSS
88      I=NPTSS-L+1
89      SUSI(I,IP)=G(I)-W(I)*SUSI(I+1,IP)
90      CONTINUE
91      RETURN
92      END

```

UPAUG1 37
OCT30 15
OCT30 16
UPAUG1 40
OCT30 42
UPAUG1 43
UPAUG1 44
UPAUG1 45
UPAUG1 46
UPAUG1 47
UPAUG1 48
UPAUG1 49
UPAUG1 50
UPAUG1 51
UPAUG1 52
FSENS3 105
FSENS3 106
FSENS3 107
FSENS3 108
FSENS3 109
FSENS3 110
FSENS3 111
FSENS3 112
FSENS3 113
FSENS3 114
FSENS3 115
FSENS3 116
FSENS3 117
FSENS3 118
FSENS3 119
FSENS3 120
FSENS3 121
FSENS3 122
FSENS3 123
FSENS3 124
FSENS3 125
FSENS3 126

DO=-LONG/-OT,ARG=-COMMON/-FIXED,CS= USER/-FIXED,DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST,PL=5000
FTNS,I,ANSI=0,L=OUTS,LO=S/-A.

```

1  SUBROUTINE PAREST
2  LOGICAL FAUTO
3  DIMENSION FAUTO(7)
4  DIMENSION QP(5)
5  COMMON/CPARAM/HO,HALF(2),PHIC,PHIK,ZP,Z,ALPH(2),KA,S(5),
6  &CIF(5,5),KAF,IFX(5),ACC(5),IFXSUM,NPAR,DALPH(2),
7  EQUIVALENCE (HO,QP(1))
8  COMMON/ICTPS2/TINIT(1),ERALOW,E
9  COMMON/CDX/DX(1)
10 COMMON/CTIME/TSTART,TSTOP,DTPTENT,NRPITER,ITPRAM
11 COMMON/MAIN2/IMA2
12 DIMENSION CIFI(6,6)
13 DATA KIN,KOUT/5,6/
14 IMA2=16
15 NR=IFXSUM
16 IF(NR.EQ.1)THEN
17   CIFI(1,1)=1./CIFI(1,1)
18   GO TO 20
19   END IF
20
21 C INVERT CONDITIONAL INFORMATION MATRIX,CIF
22 CALL GMINV(NR,NR,CIF,CIFI,MR,1)
23 DO 15 IR=1,NR
24   DO 15 IC=1,NR
25     CIFI(IR,IC)=CIFI(IC,IR)
26   IT=1
27   DO 29 IP=1,NPAR
28     IF(IFX(IP),EQ.0)GO TO 29
29     JT=1
30     DO 28 JP=1,NPAR
31       IF(IFX(JP),EQ.0)GO TO 28
32       QP(IP)=QP(IP)+CIFI(IT,JT)*S(JT)
33       JT=JT+1
34     CONTINUE
35     IT=IT+1
36   CONTINUE
37   RETURN
38   END

```

FPAREST 2
CPARAM 2
OCT10 2
UPAUG16 1
UPAUG16 2
UPAUG16 3
UPAUG16 4
UPSEP24 7
UPSEP24 8
FPAREST 8
FPAREST 9
UPSEP24 9
FPAREST 12
FPAREST 13
FPAREST 14
FPAREST 15
FPAREST 16
FPAREST 17
FPAREST 18
FPAREST 19
FPAREST 20
FPAREST 21
FPAREST 22
FPAREST 23
FPAREST 24
UPSEP24 10
FPAREST 28
FPAREST 29
FPAREST 30
FPAREST 31
FPAREST 32
OCT24 3
FPAREST 35
FPAREST 36
FPAREST 37
FPAREST 38
FPAREST 41
FPAREST 42


```

SUBROUTINE KF
DO=-LONG/-OT, ARG=-COMMON/-FIXED, CS= USER/-FIXED, DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST, PL=5000
FTNS, I, ANSI=0, L=OUTS, LO=S/-A.

SUBROUTINE KF
LOGICAL IFICENT, IFPLOT, IFPRINT
COMMON /CSMTH/ UICSM(6,6), PICS(6,6), UAP(6,6), PAR(6,6),
&SMIC, TSMTH, W(6,6)
COMMON /CFLAG/IFICENT, IFPLOT, IFPRINT
COMMON /CTCMNT/NTCT
COMMON /CPC/NPTPC
LOGICAL FAUTO
DIMENSION FAUTO(7)
COMMON /CPARAM/HO, HALF(2), P4IC, PHIK, ZP, Z, ALPH(2), KA, S(5),
&CIF(5,5), KAF, IFX(5), ACC(5), IFXSUM, NPAR, DALPH(2)
EQUIVALENCE (HO, QP(1))
COMMON /ICTPS2/TINIT(1), ERALOW, E
COMMON /CDX/DX(1)
COMMON /CSENS/SUSI(6,5), UM1(6)
COMMON /COSP/NPTSS, USI(6), PHI(6,6), NPT, PC(6,6), RR,
&QD(6,6), QDT(6,6), QUE(6,6), A(6,6), RCX(6,6), RP(6,6), RM(6)
COMMON /CTIME/TSTART, TSTOP, DTPENT, NRPITER, ITPRAM
COMMON /CPARAM2/AVERROR, EQUI, UMEAS
COMMON /CKF/K(6), S1(6), J1(6,6), TC(2), NODES(2), KFOPT
REAL K, J1
COMMON /MAIN1/INP, IPVT(6), WORK(6)
INP=6

SDMEA=RR

C
C
C THE FOLLOWING CONTROL CONSTRUCT SORTS KF UPDATE ITERATIONS
C REQUIRED BY VECTOR UPDATES AS SPECIFIED IN THE INPUT DECK
C
DO 98 I=1, NPTSS
DO 5 IIT=1, NTCT
IF(I.EQ.NODES(IIT)) THEN
NODE=NODES(IIT)
UMEAS=TC(IIT)
GO TO 10
END IF
5 CONTINUE
GO TO 98
10 ERROR=UMEAS-UAP(NODE)
C
IF(KAF.EQ.1) AERROR=ERROR
AERROR=((KAF-1)*AERROR+ERROR)/KAF
KAF=KAF+1
C
C SCORE RUNNING SUMS FOR JACOBIAN OF LIKELIHOOD FN. S.
C AND CONDITIONAL INFORMATION MATRIX, CIF
C
R=(SDMEA*UMEAS)**2.
DO 26 KO=1, NPAR
S1(KO)=SUSI(NODE, KO)*IFX(KO)*ERROR/(PC(NODE, NODE)+R)
DO 25 L=1, NPAR
25 J1(KO, L)=SUSI(NODE, KO)*IFX(KO)*IFX(L)*SUSI(NODE, L)/(PC(NODE, NODE)+
*R)

```

1 FKFSUM 2
 2 FKFSUM 3
 3 OCT10 11
 4 OCT10 12
 5 UPOCT09 25
 6 UPSEP24 12
 7 FKFSUM 8
 8 CPARAM 2
 9 OCT10 2
 10 UPAG16 1
 11 UPAG16 2
 12 UPAG16 3
 13 UPAG16 4
 14 UPAG24 3
 15 UPAG24 4
 16 UPAG16 7
 17 UPAG16 6
 18 UPOCT09 5
 19 FKFSUM 16
 20 UPOCT09 26
 21 UPAG16 8
 22 FDKF 3
 23 OCT10 4
 24 OCT10 5
 25 FKFSUM 21
 26 FKFSUM 22
 27 FKFSUM 23
 28 FKFSUM 24
 29 FKFSUM 25
 30 FKFSUM 26
 31 FKFSUM 29
 32 FKFSUM 30
 33 FKFSUM 31
 34 FKFSUM 32
 35 FKFSUM 33
 36 FKFSUM 34
 37 FKFSUM 35
 38 FKFSUM 36
 39 FKFSUM 37
 40 HAROLD 38
 41 FKFSUM 39
 42 FKFSUM 40
 43 FKFSUM 41
 44 FKFSUM 42
 45 FKFSUM 43
 46 FKFSUM 44
 47 FKFSUM 45
 48 FKFSUM 46
 49 FKFSUM 47
 50 FKFSUM 48
 51 FKFSUM 49
 52 FKFSUM 50
 53 FKFSUM 51
 54 FKFSUM 52
 55 FKFSUM 53

```

56      26      CONTINUE
57      IT=1
58      DO 29 IP=1,NPAR
59      IF(IPX(IP).EQ.O)GO TO 29
60      S(IT)=S1(IP)+S(IT)
61      JT=1
62      DO 28 JP=1,NPAR
63      IF(IPX(JP).EQ.O)GO TO 28
64      CIF(IT,JT)=J1(IP,JP)+CIF(IT,JT)
65      JT=JT+1
66      CONTINUE
67      IT=IT+1
68      29      CONTINUE
69      C
70      C      COMPUTE KALMAN GAIN, K
71      C
72      DO 30 IK=1,NPTSS
73      K(IK)=PAP(IK,MODE)/(PAP(MODE,MODE)+R)
74      C IF KOPT=1 UPDATE
75      C IF KOPT=2 UPDATE EXCEPT ON LAST ITERATION(ITPRAM-NRPTITER)
76      C IF KOPT=3 DO NOT UPDATE
77      C IF KOPT=4 UPDATE COVARIANCE AND SENSITIVITY ONLY
78      C IF KFOPT=5 ONLY UPDATE TEMP ON LAST ITERATION
79      C IF KFOPT=8 ONLY UPDATE ON LAST ITERATION
80      GO TO(101,102,103,104,101,101)KFOPT
81      102      IF(ITPRAM.GT.NRPTITER)GO TO 103
82      C
83      C      STATE UPDATE
84      C
85      101      IF(KFOPT.GE.6.AND.ITPRAM.LE.NRPTITER)GO TO 103
86      IF(KFOPT.EQ.5.AND.ITPRAM.LE.NRPTITER)GO TO 104
87      DO 40 IO=1,NPTSS
88      USI(IO)=USI(IO)+K(IO)*(UMEAS-UAP(MODE))
89      C
90      C      SENSITIVITY UPDATE
91      C
92      104      CONTINUE
93      DO 35 IP=1,NPAR
94      DO 35 L=1,NPTSS
95      35      SUSI(L,IP)=SUSI(L,IP)-K(L)*SUSI(MODE,IP)
96      C
97      C      COVARIANCE UPDATE, PC - JOSEPH FORM
98      NPTSS=NPTPC
99      C
100      CALL ZERO(QD(1,1),NPTSS,NPTSS)
101      DO 50 IC=1,NPTSS
102      QD(IC,IC) = 1.0
103      50      QD(IC,MODE) = QD(IC,MODE)-K(IC)
104      CALL MAT4(NPTSS,NPTSS,PC(1,1),QD(1,1),QDT(1,1))
105      CALL MAT4(NPTSS,1,R,K(1),QD(1,1))
106      DO 55 IPC=1,NPTSS
107      DO 55 JPC=1,NPTSS
108      PC(IPC,JPC) = QDT(IPC,JPC)+QD(IPC,JPC)
109      NPTSS=INP
110      98      CONTINUE
111      103      CONTINUE
112      RETURN

```

FKFSUM 54
FKFSUM 55
FKFSUM 56
FKFSUM 57
FKFSUM 58
FKFSUM 59
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FKFSUM 62
FKFSUM 63
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FKFSUM 66
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FKFSUM 68
FKFSUM 69
FKFSUM 70
HAROLD 37
FKFSUM 72
FKFSUM 73
FKFSUM 74
FKFSUM 75
FKFSUM 76
FKFSUM 77
FKFSUM 78
FKFSUM 79
FKFSUM 80
FKFSUM 81
FKFSUM 82
FKFSUM 83
FKFSUM 84
FKFSUM 85
HAROLD 38
FKFSUM 87
FKFSUM 88
FKFSUM 89
FKFSUM 90
FKFSUM 91
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FKFSUM 93
FKFSUM 94
FKFSUM 95
FKFSUM 96
FKFSUM 97
FKFSUM 98
FKFSUM 99
FKFSUM 100
FKFSUM 101
FKFSUM 102
FKFSUM 103
FKFSUM 104
FKFSUM 105
FKFSUM 106
FKFSUM 107
FKFSUM 108
FKFSUM 109
FKFSUM 110

111

FKFSUM

END

113

DO=-LONG/-OT,ARG=-COMMON/-FIXED,CS=USER/-FIXED,DB=-TB/-SB/-SL/ER/-ID/-PMD/-ST,PL=5000
FTN5,I,ANSI=0,L=OUTS,LO=S/-A.

```

1  SUBROUTINE FPSM(TSTART)
2  COMMON /CONST/ XP1(13)
3  COMMON /CTCHMT/NTCT
4  COMMON /CSMTH/ UICSM(6),PICSM(6,6),UAP(6),PAP(6,6),
5  &SMIC,TSMTH,W(6,6)
6  LOGICAL SMIC
7  COMMON/COSP/NPTSS,USI(6),PHI(6,6),NPT,PC(6,6),RR,
8  &QD(6,6),QDT(6,6),QUE(6,6),A(6,6),RCX(6),RP(6),RM(6)
9  COMMON/CKF/K(6),S1(6),J1(6,6),TC(2),NODES(2),KFQPT
10 REAL K,J1
11 DIMENSION RINV(8,8),HTR(6,8),SFP(6,6),GAIN(6),
12 &WRK1(6,6),WRK2(6,6)
13 COMMON /MAIN1/INP,IPVT(6),WORK(6)
14 INP=6
15
16 C THIS ROUTINE IS A FIXED POINT SMOOTHER ALGORITHM
17 C
18 CALL ZERO(SFP,NPTSS,NPTSS)
19 DO 10 I=1,NTCT
20   SFP(NODES(I),NODES(I))=1./((RR+TC(I))*2.
21 C
22 C FORM I-SP AND FIND PHIT
23 CALL MMUL(SFP,PC,NPTSS,NPTSS,NPTSS,WRK1)
24 DO 30 I=1,NPTSS
25   DO 30 J=1,NPTSS
26     WRK1(I,J)=-WRK1(I,J)
27     IF(I.EQ. J)WRK1(I,J)=1.0+WRK1(I,J)
28   WRK2(I,J)=PHI(J,I)
29   DO 35 I=1,NPTSS
30     DO 35 J=1,NPTSS
31       PHI(I,J)=WRK2(I,J)
32 C
33 C FORM W=W*PHIT*(I-SP)
34 CALL MMUL(PHI,WRK1,NPTSS,NPTSS,NPTSS,WRK2)
35 CALL MMUL(W,WRK2,NPTSS,NPTSS,NPTSS,WRK1)
36 DO 40 I=1,NPTSS
37   DO 40 J=1,NPTSS
38     W(I,J)=WRK1(I,J)
39 C
40 C SOLVE FOR COVARIANCE -- P=P-W(S*PAP*S + S)WTRAN
41 CALL MMUL(PAP,SFP,NPTSS,NPTSS,NPTSS,WRK1)
42 CALL MMUL(SFP,WRK1,NPTSS,NPTSS,NPTSS,WRK2)
43 DO 50 I=1,NPTSS
44   DO 50 J=1,NPTSS
45     WRK1(I,J)=WRK2(I,J)+SFP(I,J)
46   CALL TRI(NPTSS,WRK1,W,PHI,WRK2,NPTSS)
47   DO 60 I=1,NPTSS
48     DO 60 J=1,NPTSS
49       PICSM(I,J)=PICSM(I,J) - WRK2(I,J)
50 C
51 C SOLVE FOR SMOOTHED STATE (SCALAR UPDATES)
52 DO 150 I=1,NPTSS
53   DO 110 IYT=1,NTCT
54     IF(I.EQ. NODES(IYT))THEN
55       NODE=NODES(IYT)

```

HAROLD 39
 HAROLD 40
 OCT10 13
 UPOCT09 1
 UPOCT09 2
 FCSMTH 4
 UPAUG18 6
 UPOCT09 5
 UPAUG18 8
 FDKF 3
 UPOCT09 3
 UPOCT09 4
 OCT10 4
 OCT10 5
 FPSMIC 8
 FPSMIC 9
 FPSMIC 10
 HAROLD 42
 HAROLD 43
 HAROLD 44
 FPSMIC 25
 FPSMIC 26
 FPSMIC 27
 FPSMIC 28
 FPSMIC 29
 FPSMIC 30
 FPSMIC 31
 HAROLD 45
 HAROLD 46
 HAROLD 47
 HAROLD 48
 FPSMIC 33
 FPSMIC 34
 CHUCK1 4
 FPSMIC 38
 FPSMIC 37
 FPSMIC 38
 FPSMIC 39
 FPSMIC 40
 FPSMIC 41
 FPSMIC 42
 FPSMIC 43
 FPSMIC 44
 FPSMIC 45
 FPSMIC 46
 HAROLD 48
 FPSMIC 49
 FPSMIC 50
 FPSMIC 51
 FPSMIC 52
 FPSMIC 53
 FPSMIC 54
 FPSMIC 55
 FPSMIC 56

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      UMEAS=TC(ITT)
      GO TO 120
      ENDIF
      110 CONTINUE
      GO TO 150
      120 R=(RR*UMEAS)**2.
      C
      C COMPUTE GAIN
      DO 121 IJ=1,NPTSS
      121 GAIN(IJ)=W(IJ,NODE)/R
      C
      C UPDATE
      DO 125 IO=1,NPTSS
      TCONST=TSTART+XP1(IO)
      IF(TSMTH.GT.TCONST) GO TO 125
      UICSM(IO)=UICSM(IO)+GAIN(IO)*(UMEAS-UAP(NODE))
      125 CONTINUE
      150 RETURN
      END

      FPSMIC 57
      FPSMIC 58
      FPSMIC 59
      FPSMIC 60
      FPSMIC 61
      FPSMIC 62
      FPSMIC 63
      FPSMIC 64
      FPSMIC 65
      FPSMIC 66
      FPSMIC 67
      FPSMIC 68
      FPSMIC 69
      HAROLD 50
      HAROLD 51
      HAROLD 52
      HAROLD 53
      FPSMIC 71
      FPSMIC 72
      FPSMIC 73
```


SUBROUTINE MEXP 74/855 OPT

```

1 SUBROUTINE MEXP(N,SUB1,TIME,SUB2,Q,QT,N2)
2 DIMENSION SUB1(N2,N2),SUB2(N2,N2)
3 DIMENSION Q(N2,N2),QT(N2,N2)
4 C MULTIPLY ELEMENTS OF SUB1 BY TIME
5 DO 102 I=1,N
6 DO 102 J=1,N
7 SUB1(I,J)=SUB1(I,J)*TIME
8 102 SUB2(I,J)=SUB1(I,J)
9 C GENERATE IDENTITY MATRIX FOR INPUT Q, FOR HQR
10 DO 30 I=1,N
11 DO 30 J=1,N
12 Q(I,J)=0.
13 IF(I.EQ. J) Q(I,J)=1.
14 CALL HQR(N,SUB2,Q,IERR,N2)
15 C MATRIX SUB2 HAS BEEN DESTROYED
16 C Q IS NOW AN ORTHOGONAL TRANSFORMATION MATRIX
17 DO 40 I=1,N
18 DO 40 J=1,N
19 QT(I,J)=Q(J,I)
20 C QT IS NOW THE TRANSPOSE AND THE INVERSE, OF Q
21 CALL MULT(SUB1,Q,N,SUB2,N2)
22 CALL MULT(QT,SUB2,N,SUB1,N2)
23 C SUB1 NOW CONTAINS THE TRIANGULAR MATRIX QT*A*Q
24 DO 50 I=1,N
25 DO 50 J=1,N
26 SUB2(I,J)=0.
27 CALL FUNCT(I,N,SUB1,SUB2,N2)
28 C SUB2 NOW HOLDS EXP(A*TIME) IN TRIANGULAR FORM
29 CALL MULT(SUB2,QT,N,SUB1,N2)
30 CALL MULT(Q,SUB1,N,SUB2,N2)
31 C SUB2 NOW HOLDS EXP(A*TIME) IN ORIGINAL BASIS FORM
32 RETURN
33 END
34

```

```

SUBROUTINE FUNCT(R,S,T,F,MM)
  DIMENSION T(MM,MM),F(MM,MM)
  INTEGER R,S
  REAL EXP
  DO 10 I=R,S
    C THE IF-BLOCK GIVES 14-DIGIT ACCURACY WITHOUT UNDERFLOW
    IF( T(I,I) .LT. -43.) THEN
      F(I,I)=0.
    ELSE
      F(I,I)=EXP( T(I,I) )
    END IF
  10 CONTINUE
  C PROCESS THE KTH SUPERDIAGONAL
  N=S-R+1
  NN=N-1
  C NN = NUMBER OF SUPERDIAGONALS IN THE BLOCK
  IF(NN .EQ. 0) RETURN
  DO 13 K=1,NN
    LL=S-K
    DO 12 I=R,LL
      DIFF=T(I,I)-T(I+K,I+K)
      IF(ABS(DIFF) .EQ. 0.0) GO TO 14
      G=T(I,I+K)*(F(I,I)-F(I+K,I+K))
      KK=K-1
      IF(KK .EQ. 0) GO TO 12
      DO 11 M=1,KK
        11 G=G+(F(I,I+M)*T(I+M,I+K)-T(I,I+K-M)*F(I+K-M,I+K))
        12 F(I,I+K)=G/DIFF
      13 CONTINUE
      RETURN
    14 MM=MM+1
    RETURN
  END
  
```

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 3 FUNCT
 4 FUNCT
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 33 FUNCT
 34 FUNCT


```

SUBROUTINE HOR(IGH,H,Z,IERR,N2)
INTEGER I,J,K,L,M,N,EN,II,JJ,LL,MM,NA,NM,NN,N2,
X IGH,ITS,LOW,MP2,ENM2,IERR,MINO
REAL H(N2,N2),Z(N2,N2)
REAL P,Q,R,S,T,W,X,Y,RA,SA,VI,VR,ZZ,NORM
REAL MACHEP, SORT, ABS, SIGN, REAL, AIMAG
LOGICAL NOTLAS
COMPLEX Z3,CMPLEX

MACHEP IS A PARAMETER THAT SPECIFIES PRECISION
MACHEP=0.000000000001
NM=IGH
N=IGH
LOW=1
IERR=0
NORM=0
K=1

C COMPUTE MATRIX NORM
DO 50 I=1,N
DO 40 J=K,N
40 NORM=NORM+ABS(H(I,J))
K=I
50 CONTINUE
EN=IGH
T=0.0
** SEARCH FOR NEXT EIGENVALUES**
60 IF(EN.LT. LOW) GO TO 1001
ITS=0
NA=EN-1
ENM2=NA-1
**LOOK FOR SINGLE SMALL SUB-DIAGONAL ELEMENT
FOR L=EN STEP -1 UNTIL LOW DO **
70 DO 80 LL=LOW, EN
L=EN+LOW-LL
IF(L.EQ. LOW) GO TO 100
S=ABS(H(L-1,L-1))+ABS(H(L,L))
IF(S.EQ. 0.0) S=NORM
IF(ABS(H(L,L-1)).LE. MACHEP * S) GO TO 100
80 CONTINUE
** FORM SHIFT **
100 X=H(EN,EN)
IF(L.EQ. EN) GO TO 270
Y=H(NA,NA)
W=H(EN,NA)*H(NA,EN)
IF(L.EQ. NA) GO TO 280
IF(ITS.EQ. 30) GO TO 1000
IF(ITS.NE. 10 .AND. ITS.NE. 20) GO TO 130
** FORM EXCEPTIONAL SHIFT **
T=T+X
DO 120 I=LOW,EN
120 H(I,I)=H(I,I)-X
S=ABS(H(EN,NA))+ABS(H(NA,ENM2))
X=0.75*S

```

2 HOR
 3 HOR
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 55 HOR
 56 HOR

84/11/19 13 14 29

FTN 5.1+587

74/855 OPT=0,ROUND=A/ S/ M/-D,-DS

* ARDITINF HQP

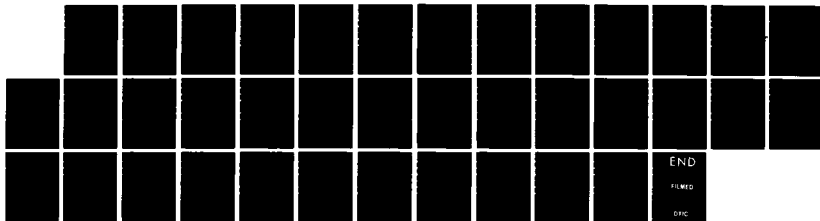
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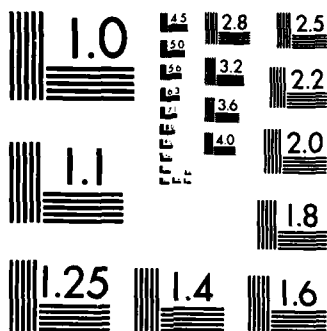
      Y = X
      W = 0.4375 * S * S
130 ITS = ITS + 1
      *** LOOK FOR TWO CONSECUTIVE SMALL SUB-DIAGONAL ELEMENTS***
      *** FOR M=EN-2 STEP -1 UNTIL L DO ***
      DO 140 MM = L, ENM2
        M = ENM2 + L - MM
        ZZ = H(M,M)
        R = X - ZZ
        S = Y - ZZ
        P = (R * S - W) / (H(M+1,M) + H(M,M+1))
        Q = H(M+1,M+1) - ZZ - R - S
        R = H(M+2,M+1)
        S = ABS(P) + ABS(Q) + ABS(R)
        P = P/S
        Q = Q/S
        R = R/S
        IF (M EQ L) GO TO 150
        IF (ABS(H(M,M-1)) * (ABS(Q) + ABS(R)) .LE. MACHEP * ABS(P))
          X = (ABS(H(M-1,M-1)) + ABS(ZZ) + ABS(H(M+1,M+1))) GO TO 150
140 CONTINUE
150 MP2 = M + 2
      DO 160 I = MP2, EN
        H(I,I-2) = 0.0
        IF (I EQ MP2) GO TO 160
        H(I,I-3) = 0.0
160 CONTINUE
      C * DOUBLE QR STEP INVOLVING ROWS L TO EN AND COLUMNS M TO EN *
      DO 280 K = M, NA
        NOTLAS = K .NE. NA
        IF (K EQ M) GO TO 170
        P = H(K,K-1)
        Q = H(K+1,K-1)
        R = 0.0
        IF (NOTLAS) R = H(K+2,K-1)
        X = ABS(P) + ABS(Q) + ABS(R)
        IF (X EQ 0.0) GO TO 260
        P = P/X
        Q = Q/X
        R = R/X
170 S = SIGN(SQRT(P+P+Q+Q+R+R),P)
        IF (K EQ M) GO TO 180
        H(K,K-1) = -S * X
        GO TO 190
180 IF (L NE M) H(K,K-1) = -H(K,K-1)
190 P = P + S
        X = P/S
        Y = Q/S
        ZZ = R/S
        Q = Q/P
        R = R/P
      *** ROW MODIFICATION ***
      DO 210 J = K, N
        P = H(K,J) + Q * H(K+1,J)
        IF (.NOT. NOTLAS) GO TO 200
        P = P + R * H(K+2,J)
        H(K+2,J) = H(K+2,J) - P * ZZ

```

HQR 57
 HQR 58
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 HQR 111
 HQR 112
 HQR 113

AD-A153 039 HEATING PARAMETER ESTIMATION USING COAXIAL THERMOCOUPLE 2/2
GAGES IN WIND TUN. (U) AIR FORCE INST OF TECH
WRIGHT-PATTERSON AFB OH SCHOOL OF ENGI. N T CAHOON
UNCLASSIFIED DEC 84 AFIT/GAE/AA/84D-3 F/G 9/2 NL





MICROCOPY RESOLUTION TEST CHART
NATIONAL BUREAU OF STANDARDS-1963-A


```

1  SUBROUTINE MULT2(N,X,Y,Z,N2)
2  C  COMPUTES Z=X*Y"
3  DIMENSION X(N2,N2),Y(N2,N2),Z(N2,N2)
4  DO 20 I=1,N
5  DO 20 J=1,N
6  Z(I,J)= 0.
7  DO 20 K=1,N
8  Z(I,J)= Z(I,J)+X(I,K)*Y(K,J)
9  RETURN
10 END

```

```

MULT2 2
MULT2 3
MULT2 4
MULT2 5
MULT2 6
MULT2 7
MULT2 8
MULT2 9
MULT2 10
MULT2 11

```



```

1  SUBROUTINE SGEFA(A,LDA,N,IPVT,INFO)
2  INTEGER LDA,N,IPVT(1),INFO
3  REAL A(LDA,1)
4
5  C SGEFA FACTORS A REAL MATRIX BY GAUSSIAN ELIMINATION.
6  C
7  C ON ENTRY:
8  C A: THE MATRIX TO BE FACTORED
9  C LDA: THE LEADING DIMENSION OF THE ARRAY A
10 C N: THE ORDER OF THE ARRAY A
11 C
12 C ON RETURN
13 C A: AN UPPER TRIANGULAR MATRIX AND THE MULTIPLIERS
14 C WHICH WERE USED TO OBTAIN IT.
15 C IPVT: AN INTEGER VECTOR OF PIVOT INDICES
16 C INFO: = 0 NORMAL VALUE.
17 C = K IF U(K,K).EQ. 0.0 THIS IS NOT AN ERROR
18 C CONDITION FOR SGEFA, BUT INDICATES THAT
19 C SGEFA WILL DIVIDE BY ZERO WHEN CALLED.
20 C THIS IS FROM LINPACK USER'S GUIDE, VERSION 08/14/78
21 C REAL T
22 C INTEGER ISAMAX,J,K,KP1,L,NM1
23 C
24 C GAUSSIAN ELIMINATION WITH PARTIAL PIVOTING
25 C INFO= 0
26 C NM1= N-1
27 C IF(NM1.LT. 1) GO TO 70
28 C DO 60 K=1,NM1
29 C KP1= K+1
30 C
31 C FIND L = PIVOT INDEX
32 C L= ISAMAX(N-K+1,A(K,K),1)+K-1
33 C IPVT(K)= L
34 C
35 C ZERO PIVOT IMPLIES THIS COLUMN IS TRIANGULARIZED
36 C IF(A(L,K).EQ. 0.0EO) GO TO 40
37 C
38 C INTERCHANGE IF NECESSARY
39 C IF(L.EQ. K) GO TO 10
40 C T=A(L,K)
41 C A(L,K)= A(K,K)
42 C A(K,K)= T
43 C A(K,K)= T
44 C 10 CONTINUE
45 C
46 C COMPUTE MULTIPLIERS
47 C T= -1.0EO/A(K,K)
48 C CALL SSCAL(N-K,T,A(K+1,K),1)
49 C
50 C ROW ELIMINATION WITH COLUMN INDEXING
51 C DO 30 J=KP1,N
52 C T= A(L,J)
53 C IF(L.EQ. K) GO TO 20
54 C A(L,J)= A(K,J)
55 C A(K,J)= T
56 C 20 CONTINUE

```

```
56  
57  
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68  
  
30 CONTINUE  
40 GO TO 50  
40 CONTINUE  
50 INFO= K  
60 CONTINUE  
70 CONTINUE  
      IPVT(N)= N  
      IF(A(N,N) .EQ. 0.0E0) INFO= N  
      RETURN  
      END  
  
SGEFA 57  
SGEFA 58  
SGEFA 59  
SGEFA 60  
SGEFA 61  
SGEFA 62  
SGEFA 63  
SGEFA 64  
SGEFA 65  
SGEFA 66  
SGEFA 67  
SGEFA 68
```

```

1  SUBROUTINE SGEDI(A,LDA,N,IPVT,WORK)
2  INTEGER LDA,N,IPVT(1)
3  REAL A(LDA,1),WORK(1)
4
5  SGEDI COMPUTES INVERSE OF MATRIX A USING
6  FACTORS COMPUTED BY SGEFA.
7
8  ON ENTRY:
9  A: THE OUTPUT FROM SGEFA, REAL(LDA,N)
10 LDA: THE LEADING DIMENSION OF ARRAY A
11 N: THE ORDER OF MATRIX A
12 IPVT: THE PIVOT VECTOR FROM SGEFA, INTEGER(N)
13 WORK: WORK VECTOR, CONTENTS DESTROYED, REAL(N)
14
15 ON RETURN:
16 A: INVERSE OF THE ORIGINAL MATRIX
17
18 ERROR CONDITION: A DIVISION BY ZERO WILL OCCUR IF THE
19 INPUT FACTOR CONTAINS A ZERO ON THE DIAGONAL.
20 IT WILL NOT OCCUR IF SGEFA HAS SET INFO=0
21
22 THIS IS FROM LINPACK USER'S GUIDE, VERSION 08/14/78
23 REAL T
24 INTEGER I,J,K,KB,KP1,L,NM1
25
26 C COMPUTE INVERSE
27 DO 100 K=1,N
28   A(K,K)= 1.OEO/A(K,K)
29   T= -A(K,K)
30   CALL SSCAL(K-1,T,A(1,K),1)
31   KP1= K+1
32   IF(N .LT. KP1) GO TO 80
33   DO 80 J=KP1,N
34     T= A(K,J)
35     CALL SAXPY(K,T,A(1,K),1,A(1,J),1)
36   80 CONTINUE
37   90 CONTINUE
38   100 CONTINUE
39
40 C FORM INVERSE(U)*INVERSE(L)
41 NM1= N-1
42 IF(NM1 .LT. 1) GO TO 140
43 DO 130 KB=1,NM1
44   K= N-KB
45   KP1= K+1
46   DO 110 I=KP1,N
47     WORK(I)= A(I,K)
48     A(I,K)= O.OEO
49   110 CONTINUE
50   DO 120 J=KP1,N
51     T= WORK(J)
52     CALL SAXPY(N,T,A(1,J),1,A(1,K),1)
53   120 CONTINUE
54   L= IPVT(K)
55   SGEDI

```

SUBROUTINE SGEDI

74/855 OPT=0,ROUND= A/ S/ M/-D.-DS

FTN 5.1+587

84/11/19. 13.14.28

PAGE

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59
60

IF(L.NE.K) CALL SSWAP(N,A(1,K),1,A(1,L),1)
130 CONTINUE
140 CONTINUE
RETURN
END

SGEDI 57
SGEDI 58
SGEDI 59
SGEDI 60
SGEDI 61

DO=-LONG/-OT,ARG=-COMMON/-FIXED,CS= USER/-FIXED,OB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST,PL=5000
 FTNS,I,ANSI=0,L=OUTS,LO=S/-A.

```

1  SUBROUTINE SAXPY(N,SA,SX,INCX,SY,INCY)
2  CONSTANT TIMES A VECTOR PLUS A VECTOR.
3  C
4  C USES UNROLLED LOOP FOR INCREMENTS= 1.
5  C FROM LINPACK USER'S GUIDE, VERSION 03/11/78
6  REAL SX(1),SY(1),SA
7  INTEGER I,INCX,INCY,IX,IY,M,MP1,N
8  IF(N.LE. 0) RETURN
9  IF(SA.EQ. 0.0) RETURN
10 IF(INCX.EQ. 1 .AND. INCY.EQ. 1) GO TO 20
11
12 C CODE FOR UNEQUAL INCREMENTS OR FOR
13 C EQUAL INCREMENTS NOT EQUAL TO 1
14 C
15 IX= 1
16 IY= 1
17 IF(INCX.LT. 0) IX= (-N+1)*INCX +1
18 IF(INCY.LT. 0) IY= (-N+1)*INCY +1
19 DO 10 I=1,N
20 SY(IY)= SY(IY)+ SA*SX(IX)
21 IX= IX+INCX
22 IY= IY+INCY
23 10 CONTINUE
24 RETURN
25
26 C CODE FOR BOTH INCREMENTS EQUAL TO 1
27 C CLEAN-UP LOOP
28 C
29 M=MOD(N,4)
30 IF(M.EQ. 0) GO TO 40
31 DO 30 I=1,M
32 SY(I)= SY(I)+ SA*SX(I)
33 30 CONTINUE
34 IF(N.LT. 4) RETURN
35 40 MP1= M+1
36 DO 50 I=MP1,N,4
37 SY(I)= SY(I)+ SA*SX(I)
38 SY(I+1)= SY(I+1)+ SA*SX(I+1)
39 SY(I+2)= SY(I+2)+ SA*SX(I+2)
40 SY(I+3)= SY(I+3)+ SA*SX(I+3)
41 50 CONTINUE
42 RETURN
43 END
44
45 SAXPY
46 SAXPY
47 SAXPY
48 SAXPY
49 SAXPY
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99 SAXPY
100 SAXPY

```

```

1  SUBROUTINE SSCAL(N,SA,SX,INCX)
2  SCALES A VECTOR BY A CONSTANT.
3  USES UNROLLED LOOPS FOR INCREMENT EQUAL TO 1.
4  C LINPACK USER'S GUIDE, VERSION 03/11/78
5
6  REAL SA,SX(1)
7  INTEGER I,INCX,M,MP1,N,NINCX
8  IF(N.LE.0) RETURN
9  IF(INCX.EQ.1) GO TO 20
10
11  C CODE FOR INCREMENT NOT EQUAL TO 1
12  C
13  NINCX= N*INCX
14  DO 10 I=1,NINCX,INCX
15  SX(I)= SA*SX(I)
16  10 CONTINUE
17  RETURN
18
19  C CODE FOR INCREMENT EQUAL TO 1.
20  C
21  CLEAN-UP LOOP
22  20 M= MOD(N,5)
23  IF(M.EQ.0) GO TO 40
24  DO 30 I=1,M
25  SX(I)= SA*SX(I)
26  30 CONTINUE
27  IF(N.LT.5) RETURN
28  40 MP1= M+1
29  DO 50 I=MP1,N,5
30  SX(I)= SA*SX(I)
31  SX(I+1)= SA*SX(I+1)
32  SX(I+2)= SA*SX(I+2)
33  SX(I+3)= SA*SX(I+3)
34  SX(I+4)= SA*SX(I+4)
35  50 CONTINUE
36  RETURN
37  END
38
39  SSCAL
40  SSCAL
41  SSCAL
42  SSCAL
43  SSCAL
44  SSCAL
45  SSCAL
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47  SSCAL
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49  SSCAL
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99  SSCAL
100 SSCAL

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SUBROUTINE SSWAP(N,SX, INCX, SY, INCY)
  INTERCHANGES TWO VECTORS.
  USES UNROLLED LOOPS FOR INCREMENTS EQUAL TO 1.
  LINPACK USER'S GUIDE, VERSION 03/11/78

  REAL SX(1),SY(1),STEMP
  INTEGER I, INCX, INCY, IX, IY, M, MP1, N
  IF(N .LE. 0) RETURN
  IF(INCX .EQ. 1 .AND. INCY .EQ. 1) GO TO 20

  C CODE FOR UNEQUAL INCREMENTS OR EQUAL INCREMENTS NOT EQUAL TO 1
  IX= 1
  IY= 1
  IF(INCX .LT. 0) IX= (-N+1)*INCX+1
  IF(INCY .LT. 0) IY= (-N+1)*INCY+1
  DO 10 I=1,N
    STEMP= SX(IX)
    SX(IX)= SY(IY)
    SY(IY)= STEMP
    IX= IX+INCX
    IY= IY+INCY
  10 CONTINUE
  RETURN

  C CODE FOR BOTH INCREMENTS EQUAL TO 1.
  CLEAN-UP LOOP
  20 M= MOD(N,3)
  IF(M .EQ. 0) GO TO 40
  DO 30 I=1,M
    STEMP= SX(I)
    SX(I)= SY(I)
    SY(I)= STEMP
  30 CONTINUE
  IF(N .LT. 3) RETURN
  40 MP1= M+1
  DO 50 I=MP1,N,3
    STEMP= SX(I)
    SX(I)= SY(I)
    SY(I)= STEMP
    STEMP= SX(I+1)
    SX(I+1)= SY(I+1)
    SY(I+1)= STEMP
    STEMP= SX(I+2)
    SX(I+2)= SY(I+2)
    SY(I+2)= STEMP
  50 CONTINUE
  RETURN
  END

```


DO=-LONG/-OT,ARG=-COMMON/-FIXED,CS=USER/-FIXED,DB=-TB/-SB/-SL/ER/-ID/-PMD/-ST,PL=5000
 FTNS,I,ANSI=0,L=OUTS,LO=S/-A.

```

1  SUBROUTINE FLYNE(M,TSCALE,TMIN,I,ASCALE,AMIN,AYL)
2  DIMENSION U(12)
3  EQUIVALENCE (T,U(1))
4  DATA IUNIT/12/
5  REWIND IUNIT
6  II=0
7  100 READ(IUNIT,1000,END=190)U
8  1000 FORMAT(8E13.7)
9  XO=(U(M)-TMIN)/TSCALE
10 YO=(U(I)-AMIN)/ASCALE+AYL
11 II=II+1
12 IF (II.EQ.1)CALL PLOT(XO,YO,3)
13 CALL PLOT(XO,YO,2)
14 GO TO 100
15 RETURN
16 END
17
2  FLINE
3  FLINE
4  FLINE
5  FLINE
6  FLINE
7  FLINE
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17 FLINE

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SUBROUTINE FLYNES
DO=-LONG/-OT,ARG=-COMMON/-FIXED,CS= USER/-FIXED,DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST,PL=5000
FTNS,I,ANSI=0,L=OUTS,LO=S/-A.

```

1  SUBROUTINE FLYNES(M,TSCALE,TMIN,I,ASCALE,AMIN,AVL,HT,ISKIP,NCHAR)
2  DIMENSION U(12)
3  EQUIVALENCE (T,U(1))
4  DATA IUNIT/12/
5  REVIND IUNIT
6  II=0
7  100 READ(IUNIT,1000,END=190)U
8  1000 FORMAT(6E13.7)
9  XO=(U(M)-TMIN)/TSCALE
10 YO=(U(I)-AMIN)/ASCALE+AYL
11 II=II+1
12 III=((II/ISKIP))*ISKIP
13 IF(II.NE.III)GO TO 100
14 CALL SYMBOL(XO,YO,HT,NCHAR,O.,-1)
15 GO TO 100
16 190 CONTINUE
17 RETURN
18 END
19 FLYNES
20 FLYNES
21 FLYNES
22 FLYNES
23 FLYNES
24 FLYNES
25 FLYNES
26 FLYNES
27 FLYNES
28 FLYNES
29 FLYNES

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```

56 C START PLOT SEQUENCE
57 HT=.07
58 CALL PLOT(4.,.5,-3)
59
60 C
61 YLAB="T2(DEG F )"
62 TLAB="TIME(SEC) "
63 CALL AXIS(0.,0.,TLAB,-10,TAXL,0.,TMIN,TSCALE)
64 CALL AXIS(0.,0.,YLAB,10,YAXL,90.,YMIN,YSCALE)
65 C 4 IN CALL POINTS TO 4TH VARIABLE IN READ,Y
66 CALL FLINE(1,TSCALE,TMIN,4,YSCALE,YMIN,0.)
67 C 5 POINTS TO Z
68 CALL FLINES(1,TSCALE,TMIN,5,YSCALE,YMIN,0.,HT,1,3)
69 C PLOT DEPENDENT VARIABLE
70 AYL=YAXL+1.
71 ALAB="ALPHA(DEG)"
72 CALL AXIS(0.,AYL,ALAB,10,AXL,90.,AMIN,ASCALE)
73 C 6 POINTS TO A
74 CALL FLINE(1,TSCALE,TMIN,6,ASCALE,AMIN,AYL)
75 C
76 C
77 C NEXT PLOT SEQUENCE
78 AXO=TAXL+2.
79 CALL PLOT(AXO,0.,-3)
80 YLAB=" Q/QREF "
81 YAXL=10.
82 YSCALE=.1
83 YMIN=0.
84 IFOX(1)=IFX(4)
85 DO 32 I=2,6
86 IFOX(I)=IFX(I+9)
87 CONTINUE
88 DO 200 I=1,6
89 IF(IFOX(I).EQ.0)GO TO 200
90 DATA XL/"ALPHA(DEG)","BETA(DEG)","LOG(RE) ","
91 &"DELE(DEG) ","DELBF(DEG) "," MACH "/"
92 DATA XXL/5.,6.,4.,4.,5.,5./
93 DATA XSC/5.,1.,1.,5.,5.,5./
94 DATA XM/20.,-3.,5.,-10.,0.,0./
95 CALL AXIS(0.,0.,YLAB,10,YAXL,90.,YMIN,YSCALE)
96 CALL AXIS(0.,0.,XL(I),-10,XXL(I),0.,XM(I),XSC(I))
97 C 1+5 POINTS TO ALPHA / 12 POINTS TO Q/QREF
98 IPT=I+5
99 CALL FLINE(IPT,XSC(I),XM(I),12,YSCALE,YMIN,0.)
100 CALL PLOT(AXO,0.,-3)
101 CONTINUE
102 CALL PLOTE(N)
103 RETURN
104 END

```

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94 HAROLD

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DO=-LONG/-OT,ARG=-COMMON/-FIXED,CS=USER/-FIXED,DB=-TB/-SB/-SL/ER/-ID/-PMD/-ST,PL=5000
FTNS,I,ANSI=0,L=OUTS,LO=S/-A.

```
1  SUBROUTINE FPLOT
2  COMMON/CFPLOT/FPLOT,TSCALE,TMIN,TAXL,YSCALE,YMIN,YAXL,
3  SASCALE,AMIN,AAXL
4  DIMENSION XL(6),XLL(6),XSC(6),XM(6)
5  DIMENSION IFOX(6)
6  DIMENSION DUM(1024)
7  LOGICAL FAUTO
8  DIMENSION FAUTO(7)
9  DIMENSION QP(5)
10 COMMON/CPARAM/HO,HALF(2),PHIC,PHIK,ZP,Z,ALPH(2),KA,S(5),
11 ACIF(5,5),KAF,IFX(5),ACC(5),IFXSUM,NPAR,DALPH(2),
12 EQUIVALENCE (HO,QP(1))
13 C INITIALIZE PLOTS AND WRITE PLOT FILE TO UNIT 2
14 CALL PLOTS(DUM,1024,2)
15 CALL FACTOR(.787402)
16 DATA IUNIT/12/
17 REWIND IUNIT
18 C FIND MAX AND MINS FOR SCALING
19 DATA TMN,TMX,YMN,YMx/1.E7,0.,.5000.,-.480./
20 DATA AMN,AMX/25.,.45./
21 C READ T,USI1,EQUI,USI2,UNEAS,ALPHA
22 100 READ(IUNIT,1000,END=190)T,U,V,Y,Z,A,B,R,DE,DB,DM,QN
23 1000 FORMAT(6E13.7)
24 TMN=AMIN1(TMN,T)
25 TMX=AMAX1(TMX,T)
26 YMN=AMIN1(YMN,U,V,Y,Z)
27 YMx=AMAX1(YMx,U,V,Y,Z)
28 AMN=AMIN1(AMN,A)
29 AMX=AMAX1(AMX,A)
30 GO TO 100
31 190 CONTINUE
32 IF(IPLT.GT.0)GO TO 195
33 C DEFAULT TIME AXIS LENGTH = 4 INCHES
34 TAXL=4.
35 TSCALE=IFIX(((TMX-TMN)/TAXL)+.999)
36 TMIN=TMN
37 C DEFAULT Y AXIS LENGTH = 4 INCHES
38 YAXL=4
39 DYMIN=25.
40 YSCALE=DYMIN*IFIX((YMX-YMN)/DYMIN/YAXL+1.899)
41 YMIN=YSCALE*IFIX(YMN/YSCALE)
42 C DEFAULT A AXIS LENGTH = 2 INCHES
43 AAXL=2.
44 DAMIN=5
45 ASCALE=DAMIN*IFIX((AMX-AMN)/DAMIN/AAXL+1.899)
46 AMIN=ASCALE*IFIX(AMN/ASCALE)
47 195 CONTINUE
48 C
49 C SCALE TIME USING INPUT TSCALE ONLY
50 C PUT IN NEGATIVE OR ZERO FOR TMIN AND TAXL
51 IF(TAXL.GT.0)GO TO 198
52 TMIN=TMN
53 TAXL=IFIX((TMX-TMN)/TSCALE+.999)
54 198 CONTINUE
55 C
```

FPLOT 2
FPLOT 3
FPLOT 4
FPLOT 5
HAROLD 55
FPLOT 6
CPARAM 2
OCT10 2
UPAUG16 1
UPAUG16 2
UPAUG16 3
UPAUG16 4
FPLOT 8
FPLOT 9
FPLOT 10
FPLOT 11
FPLOT 12
FPLOT 13
FPLOT 14
FPLOT 15
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FPLOT 49
FPLOT 50

DO=-LONG/-OT,ARG=-COMMON/-FIXED,CS= USER/-FIXED,DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST,PL=5000
FTNS,I,ANSI=0,L=OUTS,L0=S/-A.

1	SUBROUTINE MSCALE (N1,N2,A,X,B)	MSCALE	2
2	DIMENSION A(1), B(1)	MSCALE	3
3	COMMON /MAIN1/ NDIM	MSCALE	4
4	DIMENSION IRAY(8)	SYSTEMC	2
5	DATA IRAY/8* -0/	SYSTEMC	3
6	IRAY(4)=0	SYSTEMC	4
7	JENO=N2*NDIM	MSCALE	5
8	DO 1 I=1,N1	MSCALE	6
9	CALL SYSTEMC(144,IRAY)	DCT24	5
10	DO 1 IJ=1,JEND,NDIM	MSCALE	7
11	1 B(IJ)=X*A(IJ)	MSCALE	8
12	RETURN	MSCALE	9
13	END	MSCALE	10


```

1 SUBROUTINE MAT4 (N1,N2,X,Y,Z)
2 Z=XY" X=X" IS N2XN2, Y IS N1XN2, Z IS N1XN1
3 DIMENSION X(1), Y(1), Z(1)
4 COMMON /MAIN1/ NDIM
5 CALL MMUL (Y,X,N1,N2,N2,Z)
6 NN2=N2*NDIM
7 DO 3 I=1,N1
8 IM1=I-1
9 II=IM1+NDIM
10 JJ=I+II
11 DO 2 J=I,N1
12 TEMP=0.
13 KK=J
14 DO 1 K=I,NN2,NDIM
15 TEMP=TEMP+Y(K)*Z(KK)
16 1 KK=KK+NDIM
17 Z(JJ)=TEMP
18 2 JJ=JJ+NDIM
19 JJ=I
20 K=II+1
21 KK=II+IM1
22 DO 3 J=K, KK
23 Z(JJ)=Z(J)
24 JJ=JJ+NDIM
25 3 CONTINUE
26 RETURN
27 END
28

```

FUNCTION XNORM

DO=-LONG/-OT,ARG=-COMMON/-FIXED,CS=

USER/-FIXED,DB=-TB/-SB/-SL/

ER/-ID/-PMD/-ST

PL=5000

FTNS,I,ANSI=0,L=OUTS,LO=S/-A.

```

1      C
2      FUNCTION XNORM (N,A)
3      COMPUTES AN APPROXIMATION TO NORM OF A-- NOT A BOUND
4      DIMENSION A(1)
5      COMMON /MAIN1/ NDIM
6      NDIM=NDIM+1
7      NN=N*NDIM
8      C1=0.
9      TR=A(1)
10     IF (N.EQ.1) GO TO 4
11     I=2
12     DO 2 II=NDIM1,NN,NDIM
13     J=II
14     DO 1 JJ=1,II,NDIM
15     C1=C1+ABS(A(J)*A(JJ))
16     1 J=J+1
17     TR=TR+A(J)
18     2 I=I+1
19     TR=TR/FLOAT(N)
20     DO 3 II=1,NN,NDIM1
21     C1=C1+(A(II)-TR)**2
22     3 XNORM=ABS(TR)+SQRT(C1)
23     RETURN
24     END

```

2 XNORM
 3 XNORM
 4 XNORM
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 6 XNORM
 7 XNORM
 8 XNORM
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 10 XNORM
 11 XNORM
 12 XNORM
 13 XNORM
 14 XNORM
 15 XNORM
 16 XNORM
 17 XNORM
 18 XNORM
 19 XNORM
 20 XNORM
 21 XNORM
 22 XNORM
 23 XNORM
 24 XNORM

DO=-LONG/-OT,ARG=-COMMON/-FIXED,CS= USER/-FIXED,DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST,PL=5000
FTNS,I,ANSI=0,L=OUTS,LO=S/-A.

```

1  SUBROUTINE TRANS(A,N,X)
2  DIMENSION A(1),X(1)
3  COMMON/MAIN1/NDIM
4  DO 10 I=1,N
5  DO 10 J=1,N
6  II=I+NDIM*(J-1)
7  JJ=J+NDIM*(I-1)
8  A(II)=X(JJ)
9  RETURN
10 END

```

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TRANSP

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60 CONTINUE
70 CONTINUE
80 CONTINUE
90 CONTINUE
100 CONTINUE
RETURN
END

SGTSL
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SGTSL

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```

1  SUBROUTINE INTEG(N,A,T,QUE,PHI,QD,QDT,N2)
2  DIMENSION A(N2,N2),QUE(N2,N2),PHI(N2,N2),QD(N2,N2),QDT(N2,N2)
3  T2=-T*0.5
4  CALL MEXP(N,A(1,1),T2,PHI(1,1),QD(1,1),QDT(1,1),N2)
5  CALL TRI(N,QUE(1,1),PHI(1,1),QD(1,1),QDT(1,1),N2)
6  CALL MSCALE(N,N,QDT(1,1),4.0,QDT(1,1),N2)
7  CALL MULT(PHI(1,1),PHI(1,1),N,A(1,1),N2)
8  CALL TRI(N,QUE(1,1),A(1,1),QD(1,1),PHI(1,1),N2)
9  DO 10 I=1,N
10  DO 10 J=1,N
11  QDT(I,J)= QUE(I,J)+ QDT(I,J)+ PHI(I,J)
12  T6=T/6.0
13  CALL MSCALE(N,N,QDT(1,1),T6,QDT(1,1))
14  RETURN
15  END
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17 INTEG2
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1  SUBROUTINE TPOSOSP2(DT)
2  COMMON/COSP/NPTSS,USI(6),PHI(6,6),NPT,PC(6,6),RR,
3  &QD(6,6),QDT(6,6),QUE(6,6),A(6,6),RCX(6),RP(6),RM(6)
4  COMMON/COPT/N,IOA(8),IOB(8),IOC(8),IOFACE(8),IOD(8)
5  COMMON/CPC/NPTPC
6  COMMON /MAIN1/INP,IPVT(6),WORK(6)
7  INP=6
8  IF(DT .LE. 0.) GO TO 999
9  T=DT
10 CALL INTEG(NPTPC,A,I,QUE,PHI,QD,QDT,NPTSS)
11 CALL SGFEFA(A(1,1),NPTSS,NPTPC,IPVT(1))
12 CALL SGEDI(A(1,1),NPTSS,NPTPC,IPVT(1),WORK(1))
13 CALL MAT4(NPTPC,NPTPC,QDT(1,1),A(1,1),QD(1,1))
14 CALL MAT4(NPTPC,NPTPC,PC(1,1),A(1,1),QDT(1,1))
15 DO 20 I=1,NPTPC
16 DO 20 J=1,NPTPC
17   PHI(I,J)=A(I,J)
18   20 PC(I,J)= QDT(I,J)+ QD(I,J)
19 999 RETURN
20 END

```

FTPOSOSP3 2
UPAUG16 6
UPDOCT09 5
FTPOSOSP3 4
FTPOSOSP3 5
OCT10 4
OCT10 5
FTPOSOSP3 7
UPSEP24 13
FTPOSOSP3 9
FTPOSOSP3 10
FTPOSOSP3 11
FTPOSOSP3 12
FTPOSOSP3 13
FTPOSOSP3 14
FTPOSOSP3 15
HAROLD 54
FTPOSOSP3 16
FTPOSOSP3 17
FTPOSOSP3 18

DO=-LONG/-OT, ARG=-COMMON/-FIXED, CS= USER/-FIXED, DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST, PL=5000
FTNS, I, ANSI=0, L=OUTS, LO=S/-A.

1	SUBROUTINE EQUATE (NR, NC, A, B)	
2	DIMENSION A(1), B(1)	
3	COMMON /MAIN1/ NDIM	
4	NN=NC*NDIM	
5	NR1=NR-1	
6	DO 1 J=1, NN, NDIM	
7	II=J+NR1	
8	DO 1 IJ=J, II	
9	A(IJ)=B(IJ)	
10	1 CONTINUE	
11	RETURN	
12	END	

2	EQUATE
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13	EQUATE

DO=-LONG/-OT, ARG=-COMMON/-FIXED, CS= USER/-FIXED, DB=-TB/-SB/-SL/ ER/-ID/ -PMD/ -ST, PL=5000
FTNS.1, ANSI=0, L=OUTS, LO=S/-A.

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1  SUBROUTINE GMINV (NR, NC, A, U, MR, MT)
2  DIMENSION A(1), U(1), S(30)
3  COMMON /MAIN2/ NDIM
4  COMMON /INDJ/ KIN, KOUT
5  NDIM=NDIM+1
6  TOL=1.E-14
7  ADV=1.E-24
8  MR=NC
9  NRM1=NR-1
10 TOL1=0.
11 JJ=1
12 DO 1 J=1, NC
13 S(J)=DOT(NR, A(JJ), A(JJ))
14 IF (S(J).GT. TOL1) TOL1=S(J)
15 1 JJ=JJ+NDIM
16 TOL1=ADV*TOL1
17 ADV=TOL1
18 JJ=1
19 DO 14 J=1, NC
20 FAC=S(J)
21 JM1=J-1
22 JRM=JJ+NRM1
23 JCM=JJ+JM1
24 DO 2 I=JJ, JCM
25 U(I)=0.
26 U(JCM)=1.0
27 IF (J.EQ.1) GO TO 5
28 KK=1
29 DO 3 K=1, JM1
30 IF (S(K).EQ.1.0) GO TO 3
31 TEMP=-DOT(NR, A(JJ), A(KK))
32 CALL VADD (K, TEMP, U(JJ), U(KK))
33 3 KK=KK+NDIM
34 DO 4 L=1, 2
35 KK=1
36 DO 4 K=1, JM1
37 IF (S(K).EQ.0.) GO TO 4
38 TEMP=-DOT(NR, A(JJ), A(KK))
39 CALL VADD (NR, TEMP, A(JJ), A(KK))
40 CALL VADD (K, TEMP, U(JJ), U(KK))
41 4 KK=KK+NDIM
42 TOL1=TOL+FAC+ADV
43 FAC=DOT(NR, A(JJ), A(JJ))
44 5 IF (FAC.GT. TOL1) GO TO 9
45 DO 6 I=JJ, JRM
46 A(I)=0.
47 S(J)=0.
48 KK=1
49 IF (S(K).EQ.0.) KK=KK+NDIM
50 IF (S(K).EQ.0.) GO TO 8
51 DO 7 K=1, JM1
52 TEMP=-DOT(K, U(KK), U(JJ))
53 CALL VADD (NR, TEMP, A(JJ), A(KK))
54 7 KK=KK+NDIM
55 8 FAC=DOT(J, U(JJ), U(JJ))

```



```

56 MR=MR-1
57 GO TO 11
58 9 S(J)=1.0
59 KK=1
60 DO 10 K=1,JM1
61 IF (S(K).EQ.1.) GO TO 10
62 TEMP=-DOT(NR,A(JJ),A(KK))
63 CALL VADO (K,TEMP,U(JJ),U(KK))
64 10 KK=KK+NDIM
65 11 FAC=1./SQRT(FAC)
66 DO 12 I=JJ,JRH
67 12 A(I)=A(I)*FAC
68 DO 13 I=JJ,JCM
69 13 U(I)=U(I)*FAC
70 JJ=JJ+NDIM
71 IF (MR.EQ.NR.OR.MR.EQ.NC) GO TO 15
72 IF (MT.NE.O) WRITE (KOUT,19) NR,NC,MR
73 15 MEND=NC+NDIM
74 JJ=1
75 DO 18 J=1,NC
76 DO 18 I=1,NR
77 II=I-J
78 S(I)=0.
79 DO 18 KK=JJ,MEND,NDIM
80 18 S(I)=S(I)+A(II+KK)*U(KK)
81 II=J
82 DO 17 I=1,NR
83 U(II)=S(I)
84 17 II=II+NDIM
85 18 JJ=JJ+NDIM
86 RETURN
87 19 FORMAT (13,1HX,12.8H M: RANK,12)
88 END

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1 SUBROUTINE VADD (N,C1,A,B)
2 DIMENSION A(1), B(1)
3 DO 1 I=1,N
4 1 A(I)=A(I)+C1*B(I)
5 RETURN
6 END
```

DO=-LONG/-OT,ARG=-COMMON/-FIXED,CS= USER/-FIXED,DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST,PL=5000
FTNS,I,ANSI=0,L=OUTS,LO=S/-A.

FZERO
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FZERO
FZERO
FZERO
FZERO

SUBROUTINE ZERO (A,NR,NC)
DIMENSION A(NR,NC)
DO 1 IC=1,NC
DO 1 IR=1,NR
1 A(IR,IC)=0.0
RETURN
END

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A heat energy balance is applied to a coaxial thermocouple gage for parameter estimation in wind tunnel test articles. This method can significantly reduce wind tunnel test costs and time. Modifications to the data reduction program HEATEST (HEATIng ESTimation) are made. The program allows for transient test techniques to be used as well as assuming an isothermal wall. A non-linear convective heat transfer coefficient model may also be used. Data is generated to test the new program. Temperature profiles throughout the thermocouple gage were good and were compared with changes in time step, thermocouple length, and number of discrete node points. The estimation of the convective heat transfer coefficient and thermal conductivity were excellent.

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